ТНЕ

SSSSS		000000	FFFFI	FFFFFFFF	AA	AAAAA
SSSSSSSS	S 0000	000000000	FFFF	FFFFFFF	AAA	AAAAA
SSSSSSSSS	S 00000	000000000	FFFFF	FFFFFF	AAAA	AAAA
SSSS S	000000	00000	FFFF		AAAA	AAAA
SSSSS	00000	0000	FFFFF		AAAA	AAAA
SSSSSSSSS	0000	00000	FFFFFF	FFFFF	AAAA	AAAA
SSSSSSSSS	00000	0000	FFFFFFF	FFFF A	ААААААА	AAAAA
SSSSS	0000	0000	FFFF	A	ААААААА	AAAAA
S SSSS	00000	00000	FFFF	AA	ААААААА	AAAAA
SSSSSSSSSSS	00000000	00000	FFFF	AAA	A	AAAAA
SSSSSSSSS	0000000	000	FFFF	AAAA	i i i i i i i i i i i i i i i i i i i	AAAAA
SSSS	00000	1	FFFF	AAAA	i	ААААА

SOFTWARE

LIBRARIES

International Astronomical Union Division 1: Fundamental Astronomy Commission 19: Rotation of the Earth

Standards Of Fundamental Astronomy Board

Release 9

2012 March 1

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THE IAU-SOFA SOFTWARE LIBRARIES

SOFA stands for "Standards Of Fundamental Astronomy". The SOFA software libraries are a collection of subprograms, in sourcecode form, which implement official IAU algorithms for fundamentalastronomy computations. The subprograms at present comprise 131 "astronomy" routines supported by 55 "vector/matrix" routines, available in both Fortran77 and C implementations.

THE SOFA INITIATIVE

SOFA is an IAU Service which operates under Division 1 (Fundamental Astronomy) and reports through Commission 19 (Rotation of the Earth).

The IAU set up the SOFA initiative at the 1994 General Assembly, to promulgate an authoritative set of fundamental-astronomy constants and algorithms. At the subsequent General Assembly, in 1997, the appointment of a SOFA Review Board and the selection of a site for the SOFA Center (the outlet for SOFA products) were announced.

The SOFA initiative was originally proposed by the IAU Working Group on Astronomical Standards (WGAS), under the chairmanship of Toshio Fukushima. The proposal was for "...new arrangements to establish and maintain an accessible and authoritative set of constants, algorithms and procedures that implement standard models used in fundamental astronomy". The SOFA Software Libraries implement the "algorithms" part of the SOFA initiative. They were developed under the supervision of an international panel called the SOFA Review Board. The current membership of this panel is listed in an appendix.

A feature of the original SOFA software proposals was that the products would be self-contained and not depend on other software. This includes basic documentation, which, like the present file, will mostly be plain ASCII text. It should also be noted that there is no assumption that the software will be used on a particular computer and Operating System. Although OS-related facilities may be present (Unix make files for instance, use by the SOFA Center of automatic code management systems, HTML versions of some documentation), the routines themselves will be visible as individual text files and will run on a variety of platforms.

ALGORITHMS

The SOFA Review Board's initial goal has been to create a set of callable subprograms. Whether "subroutines" or "functions", they are all referred to simply as "routines". They are designed for use by software developers wishing to write complete applications; no runnable, free-standing applications are included in SOFA's present plans.

The algorithms are drawn from a variety of sources. Because most of the routines so far developed have either been standard "text-book" operations or implement well-documented standard algorithms, it has not been necessary to invite the whole community to submit algorithms, though consultation with authorities has occurred where necessary. It should also be noted that consistency with the conventions published by the International Earth Rotation Service was a stipulation in the original SOFA proposals, further constraining the software designs. This state of affairs will continue to exist for some time, as there is a large backlog of agreed extensions to work on. However, in the future the Board may decide to call for proposals, and is in the meantime willing to look into any suggestions that are received by the SOFA Center.

SCOPE

The routines currently available are listed in the next two chapters of this document.

The "astronomy" library comprises 131 routines (plus one obsolete Fortran routine that now appears under a revised name). The areas addressed include calendars, time scales, ephemerides, precessionnutation, star space-motion, star catalog transformations and geodetic/geocentric transformations.

The "vector-matrix" library, comprising 55 routines, contains a collection of simple tools for manipulating the vectors, matrices and angles used by the astronomy routines.

There is no explicit commitment by SOFA to support historical models, though as time goes on a legacy of superseded models will naturally accumulate. There is, for example, no support of B1950/FK4 star coordinates, or pre-1976 precession models, though these capabilities could be added were there significant demand.

Though the SOFA software libraries are rather limited in scope, and are likely to remain so for a considerable time, they do offer distinct advantages to prospective users. In particular, the routines are:

- * authoritative: they are IAU-backed and have been constructed with great care;
- * practical: they are straightforward to use in spite of being precise and rigorous (to some stated degree);
- * accessible and supported: they are downloadable from an easy-tofind place, they are in an integrated and consistent form, they come with adequate internal documentation, and help for users is available.

VERSIONS

Once it has been published, an issue is never revised or updated, and remains accessible indefinitely. Subsequent issues may, however, include corrected versions under the original routine name and filenames. However, where a different model is introduced, it will have a different name.

The issues will be referred to by the date when they were announced. The frequency of re-issue will be decided by the Board, taking into account the importance of the changes and the impact on the user community.

DOCUMENTATION

At present there is little free-standing documentation about individual routines. However, each routine has preamble comments which specify in detail what the routine does and how it is used.

The file sofa_pn.pdf describes the SOFA tools for precession-nutation and other aspects of Earth attitude and includes example code and (see the appendix) diagrams showing the interrelationships between the routines supporting the latest (IAU 2006/2000A) models.

PROGRAMMING LANGUAGES AND STANDARDS

The SOFA routines are available in two programming languages at present: Fortran77 and ANSI C. Related software in other languages is under consideration.

The Fortran code conforms to ANSI X3.9-1978 in all but two minor respects: each has an IMPLICIT NONE declaration, and its name has a prefix of "iau_" and may be longer than 6 characters. A global edit to erase both of these will produce ANSI-compliant code with no change in its function.

Coding style, and restrictions on the range of language features, have been much debated by the Board, and the results comply with the majority view. There is (at present) no document that defines the standards, but the code itself offers a wide range of examples of what is acceptable.

The Fortran routines contain explicit numerical constants (the INCLUDE statement is not part of ANSI Fortran77). These are drawn from the file consts.lis, which is listed in an appendix. Constants for the SOFA/C functions are defined in a header file sofam.h.

The naming convention is such that a SOFA routine referred to generically as "EXAMPL" exists as a Fortran subprogram iau_EXAMPL and a C function iauExampl. The calls for the two versions are very similar, with the same arguments in the same order. In a few cases, the C equivalent of a Fortran SUBROUTINE subprogram uses a return value rather than an argument.

Each language version includes a "testbed" main-program that can be used to verify that the SOFA routines have been correctly compiled on the end user's system. The Fortran and C versions are called t_sofa_f.for and t_sofa_c.c respectively. The testbeds execute every SOFA routine and check that the results are within expected accuracy margins. It is not possible to guarantee that all platforms will meet the rather stringent criteria that have been used, and an occasional warning message may be encountered on some systems.

COPYRIGHT ISSUES

Copyright for all of the SOFA software and documentation is owned by the IAU SOFA Review Board. The Software is made available free of charge for all classes of user, including commercial. However, there are strict rules designed to avoid unauthorized variants coming into circulation. It is permissible to distribute derived works and other modifications, but they must be clearly marked to avoid confusion with the SOFA originals.

Further details are included in the block of comments which concludes every routine. The text is also set out in an appendix to the present document.

ACCURACY

The SOFA policy is to organize the calculations so that the machine accuracy is fully exploited. The gap between the precision of the underlying model or theory and the computational resolution has to be kept as large as possible, hopefully leaving several orders of magnitude of headroom.

The SOFA routines in some cases involve design compromises between rigor and ease of use (and also speed, though nowadays this is seldom a major concern).

ACKNOWLEDGEMENTS

The Board is indebted to a number of contributors, who are acknowledged in the preamble comments of the routines concerned.

The Board's effort is provided by the members' individual institutes.

Resources for operating the SOFA Center are provided by Her Majesty's Nautical Almanac Office, operated by the United Kingdom Hydrographic Office.

SOFA Astronomy Library

PREFACE

The routines described here comprise the SOFA astronomy library. Their general appearance and coding style conforms to conventions agreed by the SOFA Review Board, and their functions, names and algorithms have been ratified by the Board. Procedures for soliciting and agreeing additions to the library are still evolving.

PROGRAMMING LANGUAGES

The SOFA routines are available in two programming languages at present: Fortran 77 and ANSI C.

Except for a single obsolete Fortran routine, which has no C equivalent, there is a one-to-one relationship between the two language versions. The naming convention is such that a SOFA routine referred to generically as "EXAMPL" exists as a Fortran subprogram iau_EXAMPL and a C function iauExampl. The calls for the two versions are very similar, with the same arguments in the same order. In a few cases, the C equivalent of a Fortran SUBROUTINE subprogram uses a return value rather than an argument.

GENERAL PRINCIPLES

The principal function of the SOFA Astronomy Library is to provide definitive algorithms. A secondary function is to provide software suitable for convenient direct use by writers of astronomical applications.

The astronomy routines call on the SOFA vector/matrix library routines, which are separately listed.

The routines are designed to exploit the full floating-point accuracy of the machines on which they run, and not to rely on compiler optimizations. Within these constraints, the intention is that the code corresponds to the published formulation (if any).

Dates are always Julian Dates (except in calendar conversion routines) and are expressed as two double precision numbers which sum to the required value.

A distinction is made between routines that implement IAU-approved models and those that use those models to create other results. The former are referred to as "canonical models" in the preamble comments; the latter are described as "support routines".

Using the library requires knowledge of positional astronomy and time-scales. These topics are covered in "Explanatory Supplement to the Astronomical Almanac", P. Kenneth Seidelmann (ed.), University Science Books, 1992. Recent developments are documented in the journals, and references to the relevant papers are given in the SOFA code as required. The IERS Conventions are also an essential reference. The routines concerned with Earth attitude (precession-nutation etc.) are described in the SOFA document sofa_pn.pdf.

ROUTINES

Calendars

CAL2JD	Gregorian calendar to Julian Day number
EPB	Julian Date to Besselian Epoch
EPB2JD	Besselian Epoch to Julian Date
EPJ	Julian Date to Julian Epoch

EPJ2JD	Julian	Epoch	tc	o Julian Da	ate			
JD2CAL	Julian	Date t	0	Gregorian	year	, month	ı, day,	fraction
JDCALF	Julian	Date t	0	Gregorian	date	for fo	ormatteo	d output

Time scales

D2DTF DAT DTDB	format 2-part JD for output Delta(AT) (=TAI-UTC) for a given UTC date TDB-TT
DTF2D	encode time and date fields into 2-part JD
TAITT	TAI to TT
TAIUT1	TAI to UT1
TAIUTC	TAI to UTC
TCBTDB	TCB to TDB
TCGTT	TCG to TT
TDBTCB	TDB to TCB
TDBTT	TDB to TT
TTTAI	TT to TAI
TTTCG	TT to TCG
TTTDB	TT to TDB
TTUT1	TT to UT1
UT1TAI	UT1 to TAI
UT1TT	UT1 to TT
UT1UTC	UT1 to UTC
UTCTAI	UTC to TAI
UTCUT1	UTC to UT1

Earth rotation angle and sidereal time

EE00 EE00A EE00B EE06A	equation of the equinoxes, IAU 2000 equation of the equinoxes, IAU 2000A equation of the equinoxes, IAU 2000B equation of the equinoxes, IAU 2006/2000A
EECT00	equation of the equinoxes complementary terms, IAU 2000
EQEQ94	equation of the equinoxes, IAU 1994
ERA00	Earth rotation angle, IAU 2000
GMST00	Greenwich mean sidereal time, IAU 2000
GMST06	Greenwich mean sidereal time, IAU 2006
GMST82	Greenwich mean sidereal time, IAU 1982
GST00A	Greenwich apparent sidereal time, IAU 2000A
GST00B	Greenwich apparent sidereal time, IAU 2000B
GST06	Greenwich apparent ST, IAU 2006, given NPB matrix
GST06A	Greenwich apparent sidereal time, IAU 2006/2000A
GST94	Greenwich apparent sidereal time, IAU 1994

Ephemerides (limited precision)

EPV00	Earth position and velocity
PLAN94	major-planet position and velocity

Precession, nutation, polar motion

NUM00A NUM00B NUM06A NUMAT NUT00A NUT00B	nutation matrix, IAU 2000A nutation matrix, IAU 2000B nutation matrix, IAU 2006/2000A form nutation matrix nutation, IAU 2000A nutation, IAU 2000B
NUTOGA	nutation, IAU 2006/2000A
NUT80	nutation, IAU 1980
NUTM80 OBL06	nutation matrix, IAU 1980 mean obliquity, IAU 2006
OBL00 OBL80	mean obliquity, IAU 1980
PB06	zeta,z,theta precession angles, IAU 2006, including bias
PFW06	bias-precession Fukushima-Williams angles, IAU 2006
PMAT00	precession matrix (including frame bias), IAU 2000
PMAT06	PB matrix, IAU 2006
PMAT76	precession matrix, IAU 1976
PN00	bias/precession/nutation results, IAU 2000
PNOOA	bias/precession/nutation, IAU 2000A
PN00B	bias/precession/nutation, IAU 2000B
PN06	bias/precession/nutation results, IAU 2006
PN06A PNM00A	bias/precession/nutation results, IAU 2006/2000A classical NPB matrix, IAU 2000A
PNM00A PNM00B	classical NPB matrix, IAU 2000B
PNM06A	classical NPB matrix, IAU 2006/2000A
PNM80	precession/nutation matrix, IAU 1976/1980
POGE	precession angles, IAU 2006, equinox based
POM00	polar motion matrix
PR00	IAU 2000 precession adjustments
PREC76	accumulated precession angles, IAU 1976
S00	the CIO locator s, given X,Y, IAU 2000A
SOOA	the CIO locator s, IAU 2000A
SOOB	the CIO locator s, IAU 2000B
S06 S06A	the CIO locator s, given X,Y, IAU 2006 the CIO locator s, IAU 2006/2000A
SP00	the TIO locator s', IERS 2003
XY06	CIP, IAU 2006/2000A, from series
XYSOOA	CIP and s, IAU 2000A
XYS00B	CIP and s, IAU 2000B
XYS06A	CIP and s, IAU 2006/2000A

Fundamental arguments for nutation etc.

FAD03	mean elongation of the Moon from the Sun
FAE03	mean longitude of Earth
FAF03	mean argument of the latitude of the Moon
FAJU03	mean longitude of Jupiter
FAL03	mean anomaly of the Moon
FALP03	mean anomaly of the Sun
FAMA03	mean longitude of Mars
FAME03	mean longitude of Mercury
FANE03	mean longitude of Neptune
FAOM03	mean longitude of the Moon's ascending node
FAPA03	general accumulated precession in longitude
FASA03	mean longitude of Saturn
FAUR03	mean longitude of Uranus
FAVE03	mean longitude of Venus

Star space i	notion
PVSTAR STARPV	space motion pv-vector to star catalog data star catalog data to space motion pv-vector

Star catalog conversions

FK52H	transform FK5 star data into the Hipparcos system
FK5HIP	FK5 to Hipparcos rotation and spin
FK5HZ	FK5 to Hipparcos assuming zero Hipparcos proper motion
H2FK5	transform Hipparcos star data into the FK5 system
HFK5Z	Hipparcos to FK5 assuming zero Hipparcos proper motion
STARPM	proper motion between two epochs

Geodetic/geocentric

EFORM	a,f for a nominated Earth reference ellipsoid
GC2GD	geocentric to geodetic for a nominated ellipsoid
GC2GDE	geocentric to geodetic given ellipsoid a,f
GD2GC	geodetic to geocentric for a nominated ellipsoid
GD2GCE	geodetic to geocentric given ellipsoid a,f

Obsolete

C2TCEO former name of C2TCIO

CALLS: FORTRAN VERSION

CALL iau BI00	(DPSIBI, DEPSBI, DRA)
	(DATE1, DATE2, RB, RP, RBP)
	(DATE1, DATE2, RB, RP, RBP)
CALL iau_BPN2XY	
	(DATE1, DATE2, RC21)
	(DATE1, DATE2, RC2I)
	(DATE1, DATE2, RC2I)
	(DATE1, DATE2, RBPN, RC2I)
	(DATE1, DATE2, X, Y, RC2I)
CALL iau_C2IXYS	(X, Y, S, RC2I)
CALL iau_C2T00A	(TTA, TTB, UTA, UTB, XP, YP, RC2T)
CALL iau_C2T00B	(TTA, TTB, UTA, UTB, XP, YP, RC2T)
CALL iau_C2T06A	(TTA, TTB, UTA, UTB, XP, YP, RC2T)
CALL iau C2TCEO	(RC2I, ERA, RPOM, RC2T)
CALL iau C2TCIO	(RC2I, ERA, RPOM, RC2T)
	(RBPN, GST, RPOM, RC2T)
	(TTA, TTB, UTA, UTB, DPSI, DEPS, XP, YP, RC2T)
	(TTA, TTB, UTA, UTB, X, Y, XP, YP, RC2T)
	(IY, IM, ID, DJM0, DJM, J)
	(SCALE, NDP, D1, D2, IY, IM, ID, IHMSF, J)
. —	
. —	
	(DATE1, DATE2, UT, ELONG, U, V)
	(SCALE, IY, IM, ID, IHR, IMN, SEC, D1, D2, J)
	(DATE1, DATE2, EPSA, DPSI)
	(DATE1, DATE2)
	(DATE1, DATE2)
	(DATE1, DATE2)
	(DATE1, DATE2)
	(N, A, F, J)
	(DATE1, DATE2)
	(RNPB, S)
D = iau_EPB	(DJ1, DJ2)
CALL iau_EPB2JD	(EPB, DJM0, DJM)
D = iau_EPJ	(DJ1, DJ2)
CALL iau_EPJ2JD	(EPJ, DJMO, DJM)
CALL iau_EPV00	(DJ1, DJ2, PVH, PVB, J)
D = iau_EQEQ94	(DATE1, DATE2)
D = iau ERA00	(DJ1, DJ2)
	(T)
	(T)
. –	(Т)
$D = iau_FAJU03$	
. –	(Т)
. —	(T)
. —	(T)
. —	(T)
$D = iau_FAME03$ $D = iau_FAME03$	(T)
	(I) (T)
$D = iau_FAPA03$	
$D = iau_FASA03$	
$D = iau_FAUR03$	
$D = iau_FAVE03$	
CALL iau_FK52H	(R5, D5, DR5, DD5, PX5, RV5,
:	RH, DH, DRH, DDH, PXH, RVH)
	(R5H, S5H)
	(R5, D5, DATE1, DATE2, RH, DH)
. — .	(GAMB, PHIB, PSI, EPS, R)
. — .	(GAMB, PHIB, PSI, EPS, X, Y)
CALL iau_GC2GD	(N, XYZ, ELONG, PHI, HEIGHT, J)

CALL <code>iau_GC2GDE</code> (<code>A, F, XYZ, ELONG, PHI, HEIGHT, J</code>) CALL <code>iau_GD2GC</code> (<code>N, ELONG, PHI, HEIGHT, XYZ, J</code>) CALL iau_GD2GCE (A, F, ELONG, PHI, HEIGHT, XYZ, J) UTA, UTB, TTA, TTB) UTA, UTB, TTA, TTB) D = iau_GMST00 (iau_GMST06 D = (UTA, UTB) UTA, UTB, TTA, TTB) UTA, UTB) D = iau_GMST82 (D iau_GST00A (iau_GST00B (D = UTA, UTB, TTA, TTB, RNPB) UTA, UTB, TTA, TTB) D = iau_GST06 (D = iau_GST06A ((UTA, UTB) iau_GST94 D = (RH, DH, DRH, DDH, PXH, RVH, CALL iau_H2FK5 R5, D5, DR5, DD5, PX5, RV5) (RH, DH, DATE1, DATE2, R5, D5, DR5, DD5) CALL iau HFK5Z CALL iau_JD2CAL (DJ1, DJ2, IY, IM, ID, FD, J) CALL iau_JDCALF (NDP, DJ1, DJ2, IYMDF, J) CALL iau_NUM00A (DATE1, DATE2, RMATN DATE1, DATE2, RMATN DATE1, DATE2, RMATN CALL iau_NUM00B (CALL iau_NUM06A (CALL iau NUMAT (EPSA, DPSI, DEPS, RMATN) CALL iau_NUT00A (DATE1, DATE2, DPSI, DEPS CALL iau_NUT00B (DATE1, DATE2, DPSI, DEPS CALL iau_NUT06A (DATE1, DATE2, DPSI, DEPS CALL iau_NUT80 (DATE1, DATE2, DPSI, DEPS) CALL iau_NUTM80 (DATE1, DATE2, RMATN) $D = iau_{OBL06}$ (DATE1, DATE2) (DATE1, DATE2) (DATE1, DATE2, BZETA, BZ, BTHETA) iau_OBL80 D =CALL iau PB06 (DATE1, DATE2, GAMB, PHIB, PSIB, EPSA) CALL iau_PFW06 CALL iau_PLAN94 (DATE1, DATE2, NP, PV, J) CALL iau_PMAT00 (DATE1, DATE2, RBP) CALL iau_PMAT06 (DATE1, DATE2, RBP) CALL iau_PMAT76 (DATE1, DATE2, RMATP) CALL iau_PN00 (DATE1, DATE2, DPSI, DEPS, EPSA, RB, RP, RBP, RN, RBPN) CALL iau PN00A (DATE1, DATE2, • DPSI, DEPS, EPSA, RB, RP, RBP, RN, RBPN) CALL iau_PN00B (DATE1, DATE2, DPSI, DEPS, EPSA, RB, RP, RBP, RN, RBPN) : (DATE1, DATE2, DPSI, DEPS, CALL iau_PN06 EPSA, RB, RP, RBP, RN, RBPN) : CALL iau_PN06A (DATE1, DATE2, DPSI, DEPS, RB, RP, RBP, RN, RBPN) CALL iau_PNM00A (DATE1, DATE2, RBPN) CALL iau_PNM00B (DATE1, DATE2, RBPN) CALL iau_PNM06A (DATE1, DATE2, RNPB) (DATE1, DATE2, RMATPN) (DATE1, DATE2, EPS0, PSIA, OMA, BPA, BQA, PIA, BPIA, CALL iau_PNM80 CALL iau_P06E $\mbox{EPSA},\mbox{CHIA},\mbox{ZA},\mbox{ZETAA},\mbox{THETAA},\mbox{PA},\mbox{GAM},\mbox{PHI},\mbox{PSI})$ CALL iau_POM00 (XP, YP, SP, RPOM) (DATE1, DATE2, DPSIPR, DEPSPR) CALL iau_PR00 CALL iau_PREC76 (EP01, EP02, EP11, EP12, ZETA, Z, THETA) CALL iau_PVSTAR (PV, RA, DEC, PMR, PMD, PX, RV, J) DATE1, DATE2, X, Y) D = iau_S00 (D = iau_S00A (DATE1, DATE2) (DATE1, DATE2) (DATE1, DATE2, X, Y) D = iau_S00B D = iau SO6 (DATE1, DATE2) (DATE1, DATE2) D = iau_S06A D = iau_SP00 CALL iau_STARPM (RA1, DEC1, PMR1, PMD1, PX1, RV1, : EP1A, EP1B, EP2A, EP2B, RA2, DEC2, PMR2, PMD2, PX2, RV2, J) CALL iau_STARPV (RA, DEC, PMR, PMD, PX, RV, PV, J) (TAI1, TAI2, TT1, TT2, J) (TAI1, TAI2, DTA, UT11, UT12, J) CALL iau_TAITT CALL iau_TAIUT1 CALL iau TAIUTC TAI1, TAI2, UTC1, UTC2, J) (TCB1, TCB2, TDB1, TDB2, J TCG1, TCG2, TT1, TT2, J) CALL iau_TCBTDB (J) CALL iau_TCGTT CALL iau_TDBTCB (TDB1, TDB2, TCB1, TCB2, J) CALL iau_TDBTT (TDB1, TDB2, DTR, TT1, TT2, J) CALL iau_TTTAI (TT1, TT2, TAI1, TAI2, J) CALL iau_TTTCG (TT1, TT2, TCG1, TCG2, J)

CALL iau_TTTDB	(TT1, TT2, DTR, TDB1, TDB2, J)
CALL iau_TTUT1	(TT1, TT2, DT, UT11, UT12, J)
CALL iau_UT1TAI	(UT11, UT12, TAI1, TAI2, J)
CALL iau_UT1TT	(UT11, UT12, DT, TT1, TT2, J)
CALL iau_UT1UTC	(UT11, UT12, DUT, UTC1, UTC2, J)
CALL iau_UTCTAI	(UTC1, UTC2, DTA, TAI1, TAI2, J)
CALL iau_UTCUT1	(UTC1, UTC2, DUT, UT11, UT12, J)
CALL iau_XY06	(DATE1, DATE2, X, Y)
CALL iau_XYSOOA	(DATE1, DATE2, X, Y, S)
CALL iau_XYS00B	(DATE1, DATE2, X, Y, S)
CALL iau_XYS06A	(DATE1, DATE2, X, Y, S)

CALLS: C VERSION

<pre>iauBp00 iauBp06 iauBp02 iauC2i00 iauC2i00 iauC2i00 iauC2ibp iauC2ibp iauC2ixy iauC2tx0 iauC2t00 iauC2t00 iauC2t00 iauC2tc2 i</pre>	<pre>(&dpsibi, &depsbi, &dra); (date1, date2, rb, rp, rbp); (date1, date2, rb, rp, rbp); y (rbpn, &x, &y); a (date1, date2, rc2i); b (date1, date2, rc2i); a (date1, date2, rc2i); a (date1, date2, rc2i); f (date1, date2, rc2i); f (date1, date2, rx, y, rc2i); f (date1, date2, x, y, rc2i); a (tta, ttb, uta, utb, xp, yp, rc2t); b (tta, ttb, uta, utb, xp, yp, rc2t); i (tta, ttb, uta, utb, xp, yp, rc2t); f (rc2i, era, rpom, rc2t); f (tta, ttb, uta, utb, dpsi, deps, xp, yp, rc2t); f (tta, ttb, uta, utb, x, y, xp, yp, rc2t); f (tta, ttb, uta, utb, x, y, xp, yp, rc2t); f (tta, ttb, uta, utb, x, y, xp, yp, rc2t); f (tta, ttb, uta, utb, x, y, xp, yp, rc2t); f (iy, im, id, &djm0, &djm); f (scale, ndp, d1, d2, &iy, &im, &id, ihmsf);</pre>
<pre>d = iauEe00a d = iauEe00a d = iauEe00a d = iauEe00a d = iauEe00a d = iauEe00a d = iauEo0a d = iauEo0a d = iauEpb iauEpb2: d = iauEpp iauEpj2; i = iauEpv00 d = iauEqe00 d = iauEqe00 d = iauFad00 d = iauFad00 d</pre>	<pre>1 (scale, iy, im, id, ihr, imn, sec, &dl, &d2); (datel, date2, epsa, dpsi); 1 (datel, date2); (datel, date2); (datel, date2); (datel, date2); (dn, &a, &f); (datel, date2); (rnpb, s); (dj1, dj2); (dj1, dj2); (dj1, dj2); (dj1, dj2 , pvh, pvb); (dj1, dj2, pvh, pvb); (dj1, dj2);</pre>

i = iauGc2gd (n, xyz, &elong, &phi, &height); i = iauGc2gde (a, f, xyz, &elong, &phi, &height); i = iauGd2gc (n, elong, phi, height, xyz); i = iauGd2gce (a, f, elong, phi, height, xyz); (uta, utb, tta, ttb); d = iauGmst00d = iauGmst06 (uta, utb, tta, ttb); d = iauGmst82 (uta, utb); d = iauGst00a (uta, utb, tta, ttb); d = iauGst00b (uta, utb); d = iauGst06 (uta, utb, tta, ttb, rnpb); d = iauGst06a (uta, utb, tta, ttb); d = iauGst94(uta, utb); (rh, dh, drh, ddh, pxh, rvh, &r5, &d5, &dr5, &dd5, &px5, &rv5); iauH2fk5 (rh, dh, date1, date2, &r5, &d5, &dr5, &dd5); iauHfk5z i = iauJd2cal (dj1, dj2, &iy, &im, &id, &fd); i = iauJdcalf (ndp, dj1, dj2, iymdf); iauNum00a (date1, date2, rmatn); iauNum00b (date1, date2, rmatn); iauNum06a (date1, date2, rmatn); iauNumat (epsa, dpsi, deps, rmatn); iauNut00a (date1, date2, &dpsi, &deps); iauNut00b (date1, date2, &dpsi, &deps); iauNut06a (date1, date2, &dpsi, &deps); (date1, date2, &dpsi, &deps); iauNut80 iauNutm80 (date1, date2, rmatn); iauObl06 (date1, date2); d = iauObl06(date1, date2); d = iauOb180(date1, date2, &bzeta, &bz, &btheta); (date1, date2, &gamb, &phib, &psib, &epsa); iauPb06 iauPfw06 i = iauPlan94 (date1, date2, np, pv); iauPmat00 (date1, date2, rbp); (date1, date2, rbp); iauPmat06 (date1, date2, rmatp); (date1, date2, dpsi, deps, iauPmat76 iauPn00 &epsa, rb, rp, rbp, rn, rbpn); (date1, date2, iauPn00a &dpsi, &deps, &epsa, rb, rp, rbp, rn, rbpn); iauPn00b (date1, date2, &dpsi, &deps, &epsa, rb, rp, rbp, rn, rbpn); (date1, date2, dpsi, deps, &epsa, rb, rp, rbp, rn, rbpn); iauPn06 iauPn06a (date1, date2, &dpsi, &deps, &epsa, rb, rp, rbp, rn, rbpn); iauPnm00a (date1, date2, rbpn); iauPnm00b (date1, date2, rbpn); iauPnm06a (date1, date2, rnpb); iauPnm80 (date1, date2, rmatpn); (date1, date2, &eps0, &psia, &oma, &bpa, &bqa, &pia, &bpia, &epsa, &chia, &za, &zetaa, &thetaa, &pa, iauP06e &gam, &phi, &psi); (xp, yp, sp, rpom); (date1, date2, &dpsipr, &depspr); iauPom00 iauPr00 iauPrec76 (ep01, ep02, ep11, ep12, &zeta, &z, &theta); i = iauPvstar (pv, &ra, &dec, &pmr, &pmd, &px, &rv); d = iauS00(date1, date2, x, y); (date1, date2); (date1, date2); (date1, date2, x, y); (date1, date2, x, y); d = iauS00ad = iauS00bd = iauS06(date1, date2); (date1, date2); d = iauS06ad = iauSp00i = iauStarpm (ral, decl, pmrl, pmdl, pxl, rvl, ep1a, ep1b, ep2a, ep2b, &ra2, &dec2, &pmr2, &pmd2, &px2, &rv2); i = iauStarpv (ra, dec, pmr, pmd, px, rv, pv); i = iauTaitt (tail, tai2, &ttl, &tt2); i = iauTaiutl (tail, tai2, dta, &utl1, &utl2); i = iauTaiutc (tai1, tai2, &utc1, &utc2); i = iauTcbtdb (tcb1, tcb2, &tdb1, &tdb2); i = iauTcgtt (tcg1, tcg2, &tt1, &tt2); i = iauTdbtcb (tdb1, tdb2, &tcb1, &tcb2);

i = iauTdbtt (tdb1, tdb2, dtr, &tt1, &tt2);
i = iauTttai (tt1, tt2, &tai1, &tai2);
i = iauTttcg (tt1, tt2, &tcg1, &tcg2);
i = iauTttdb (tt1, tt2, dtr, &tdb1, &tdb2);
i = iauTtut1 (tt1, tt2, dt, &ut11, &ut12);
i = iauUt1tai (ut11, ut12, &tai1, &tai2);
i = iauUt1tt (ut11, ut12, dt, &tt1, &tt2);
<pre>i = iauUt1utc (ut11, ut12, dut, &utc1, &utc2);</pre>
<pre>i = iauUtctai (utc1, utc2, dta, &tai1, &tai2);</pre>
<pre>i = iauUtcut1 (utc1, utc2, dut, &ut11, &ut12);</pre>
iauXy06 (date1, date2, &x, &y);
iauXys00a (date1, date2, &x, &y, &s);
iauXys00b (date1, date2, &x, &y, &s);
iauXys06a (date1, date2, &x, &y, &s);

SOFA Vector/Matrix Library

PREFACE

The routines described here comprise the SOFA vector/matrix library. Their general appearance and coding style conforms to conventions agreed by the SOFA Review Board, and their functions, names and algorithms have been ratified by the Board. Procedures for soliciting and agreeing additions to the library are still evolving.

PROGRAMMING LANGUAGES

The SOFA routines are available in two programming languages at present: Fortran 77 and ANSI C.

There is a one-to-one relationship between the two language versions. The naming convention is such that a SOFA routine referred to generically as "EXAMPL" exists as a Fortran subprogram iau_EXAMPL and a C function iauExampl. The calls for the two versions are very similar, with the same arguments in the same order. In a few cases, the C equivalent of a Fortran SUBROUTINE subprogram uses a return value rather than an argument.

GENERAL PRINCIPLES

The library consists mostly of routines which operate on ordinary Cartesian vectors (x,y,z) and 3x3 rotation matrices. However, there is also support for vectors which represent velocity as well as position and vectors which represent rotation instead of position. The vectors which represent both position and velocity may be considered still to have dimensions (3), but to comprise elements each of which is two numbers, representing the value itself and the time derivative. Thus:

- * "Position" or "p" vectors (or just plain 3-vectors) have dimension
 (3) in Fortran and [3] in C.
- * "Position/velocity" or "pv" vectors have dimensions (3,2) in Fortran and [2][3] in C.
- * "Rotation" or "r" matrices have dimensions (3,3) in Fortran and [3][3] in C. When used for rotation, they are "orthogonal"; the inverse of such a matrix is equal to the transpose. Most of the routines in this library do not assume that r-matrices are necessarily orthogonal and in fact work on any 3x3 matrix.
- * "Rotation" or "r" vectors have dimensions (3) in Fortran and [3] in C. Such vectors are a combination of the Euler axis and angle and are convertible to and from r-matrices. The direction is the axis of rotation and the magnitude is the angle of rotation, in radians. Because the amount of rotation can be scaled up and down simply by multiplying the vector by a scalar, r-vectors are useful for representing spins about an axis which is fixed.
- * The above rules mean that in terms of memory address, the three velocity components of a pv-vector follow the three position components. Application code is permitted to exploit this and all other knowledge of the internal layouts: that x, y and z appear in that order and are in a right-handed Cartesian coordinate system etc. For example, the cp function (copy a p-vector) can be used to copy the velocity component of a pv-vector (indeed, this is how the CPV routine is coded).
- * The routines provided do not completely fill the range of operations that link all the various vector and matrix options, but are confined to functions that are required by other parts of the SOFA software or which are likely to prove useful.

In addition to the vector/matrix routines, the library contains some routines related to spherical angles, including conversions to and from sexagesimal format.

Using the library requires knowledge of vector/matrix methods, spherical trigonometry, and methods of attitude representation. These topics are covered in many textbooks, including "Spacecraft Attitude Determination and Control", James R. Wertz (ed.), Astrophysics and Space Science Library, Vol. 73, D. Reidel Publishing Company, 1986.

OPERATIONS INVOLVING P-VECTORS AND R-MATRICES

Initialize

ZP	zero p-vector		
ZR	initialize r-matrix	to	null
IR	initialize r-matrix	to	identity

Copy/extend/extract

CP	сору	p-vector
CR	сору	r-matrix

Build rotations

RX	rotate	r-matrix	about	х
RY	rotate	r-matrix	about	У
RZ	rotate	r-matrix	about	z

Spherical/Cartesian conversions

S2C	spherical to unit vector
C2S	unit vector to spherical
S2P	spherical to p-vector
P2S	p-vector to spherical

Operations on vectors

PPP	p-vector plus p-vector
PMP	p-vector minus p-vector
PPSP	p-vector plus scaled p-vector
PDP	inner (=scalar=dot) product of two p-vectors
PXP	outer (=vector=cross) product of two p-vectors
PM	modulus of p-vector
PN	normalize p-vector returning modulus
SXP	multiply p-vector by scalar

Operations on matrices

RXR	r-matrix multiply
TR	transpose r-matrix

Matrix-vector products

RXP	product o	эf	r-matrix and	f	p-vector		
TRXP	product o	сf	transpose of	E	r-matrix	and	p-vector

Separation and position-angle

SEPP	angular separation from p-vectors
SEPS	angular separation from spherical coordinates
PAP	position-angle from p-vectors
PAS	position-angle from spherical coordinates

Rotation vectors

RV2M	r-vector	to	r-matrix
RM2V	r-matrix	to	r-vector

OPERATIONS INVOLVING PV-VECTORS

Initialize

ZPV zero pv-vector

Copy/extend/extract

CPV	copy pv-vector
P2PV	append zero velocity to p-vector
PV2P	discard velocity component of pv-vector

Spherical/Cartesian conversions

S2PV	spherical	to	pv-vector
PV2S	pv-vector	to	spherical

Operations on vectors

PVPPV	pv-vector plus pv-vector
PVMPV	pv-vector minus pv-vector
PVDPV	inner (=scalar=dot) product of two pv-vectors
PVXPV	outer (=vector=cross) product of two pv-vectors
PVM	modulus of pv-vector
SXPV	multiply pv-vector by scalar
S2XPV	multiply pv-vector by two scalars
PVU	update pv-vector
PVUP	update pv-vector discarding velocity

Matrix-vector products

RXPV	product c	Σf	r-matrix and	pv-v	vector	2	
TRXPV	product c	οf	transpose of	r-ma	atrix	and	pv-vector

OPERATIONS ON ANGLES

ANP	normalize radians to range 0 to 2pi
ANPM	normalize radians to range -pi to +pi
A2TF	decompose radians into hours, minutes, seconds
A2AF	decompose radians into degrees, arcminutes, arcseconds
AF2A	degrees, arcminutes, arcseconds to radians
D2TF	decompose days into hours, minutes, seconds
TF2A	hours, minutes, seconds to radians
TF2D	hours, minutes, seconds to days

CALLS: FORTRAN VERSION

. —	NDP, ANGLE, SIGN, IDMSF) NDP, ANGLE, SIGN, IHMSF)
. —	S, IDEG, IAMIN, ASEC, RAD, J)
$D = iau_ANP$ (
$D = iau_ANPM$ (
CALL iau_C2S (
CALL iau_CP (
CALL iau_CPV (
CALL iau_CR (
. —	NDP, DAYS, SIGN, IHMSF)
CALL iau IR (
CALL iau_P2PV (
	P, THETA, PHI, R)
CALL iau_PAP (
	AL, AP, BL, BP, THETA)
CALL iau_PDP (
CALL iau_PM (
CALL iau_PMP (
CALL iau_PN (
CALL iau_PPP (
CALL iau_PPSP (
CALL iau_PV2P (
. —	PV, THETA, PHI, R, TD, PD, RD)
CALL iau_PVDPV (A, B, ADB)
CALL iau_PVM (PV, R, S)
CALL iau_PVMPV (A, B, AMB)

CALL iau_PVUP (CALL iau_PXPV (CALL iau_RM2V (CALL iau_RM2V (CALL iau_RV2M (CALL iau_RXPV (CALL iau_RXPV (CALL iau_RXPV (CALL iau_RXPV (CALL iau_RXPV (CALL iau_RXR (CALL iau_S2P (CALL iau_S2P (CALL iau_S2PV (CALL iau_TF2A (CALL iau_TR2P (CALL iau_TRXPV (CALL iau_TRXPV (CALL iau_ZP (CALL iau_ZPV (CALL iau_Z	DT, PV, UPV) DT, PV, P) A, B, AXB) A, B, AXB) A, B, AXB) R, P) P, R) PHI, R) R, P, RP) R, PV, RPV) A, B, ATB) THETA, PHI, C) THETA, PHI, R, P) THETA, PHI, R, TD, PD, RD, PV) S1, S2, PV) A, B, S) AL, AP, BL, BP, S) S, PV, SPV) S, IHOUR, IMIN, SEC, RAD, J) S, IHOUR, IMIN, SEC, DAYS, J) R, RT) R, P, TRP)
CALLS: C VERSION	
<pre>iauA2af (iauA2tf (i = iauAf2a (d = iauAnp (d = iauAnpm (d = iauAnpm (iauCzs (iauCp (iauCp (iauCp (iauD2tf (iauD2tf (iauPv (iauP2s (d = iauPag (d = iauPag (d = iauPas (iauPvps (iauPvpp (iauPv2p (iauPv2p (iauPv2p (iauPv2p (iauPvdpv (iauPvppv (iauPvppv (iauPvpv (iauRvpv (iauRxp (iauRxp (iauRx) (iauRx (iauRy (iauRy (iauRy (iauRx (iauRy (iauRy (iauRx (iauRy (iauRx (iauRy (iauRx (iauRy (iauRx (iauRx (iauRy (iauRx (iauRx (iauRy (iauRx (iauRx (iauRx (iauRy (iauRx (iauRy (iauRx (iauRy (iauRx (iauR</pre>	<pre>ndp, angle, &sign, idmsf); ndp, angle, &sign, ihmsf); s, ideg, iamin, asec, &rad); a); a); p, θ, φ); p, c); pv, c); r, c); ndp, days, &sign, ihmsf); r); p, pv); p, θ, φ, &r); a, b); a, b ,i a, b, apb); a, b, amb); pv, θ, φ, &r, &td, &pd, &rd); a, b, apb); a, b, adb); pv, θ, φ, &r, &td, &pd, &rd); a, b, adb); pv, &r, &s); a, b, adb); pv, &r, &s); a, b, adb); pv, &r, &s); a, b, amb); dt, pv, p); a, b, axb); r, p); p, r); phi, r); r, pv, rpv); a, b, atb); theta, r); psi, r); theta, phi, c);</pre>

	iauS2p iauS2pv	<pre>(theta, phi, r, p); (theta, phi, r, td, pd, rd, pV);</pre>
	iauS2xpv	(s1, s2, pv);
d :	= iauSepp	(a,b);
d :	= iauSeps	(al, ap, bl, bp);
	iauSxp	(s, p, sp);
	iauSxpv	(s, pv, spv);
		(s, ihour, imin, sec, &rad);
i :	= iauTf2d	(s, ihour, imin, sec, &days);
	iauTr	(r, rt);
	iauTrxp	(r, p, trp);
	iauTrxpv	(r, pv, trpv);
	iauZp	(p);
	iauZpv	(pv);
	iauZr	(r);

SUBROUTINE iau_A2AF (NDP, ANGLE, SIGN, IDMSF) *+ * iau_A2AF * * _ _ _ _ _ * Decompose radians into degrees, arcminutes, arcseconds, fraction. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * Given: * NDP i resolution (Note 1) * ANGLE d angle in radians * * Returned: '+' or '-' * SIGN С i(4) * degrees, arcminutes, arcseconds, fraction IDMSF * * Called: * iau_D2TF decompose days to hms * * Notes: * * 1) NDP is interpreted as follows: * NDP resolution * ...0000 00 00 : * -7 1000 00 00 * 100 00 00 -6 * -5 10 00 00 * $^{-4}$ 1 00 00 * -3 0 10 00 * -2 0 01 00 * -1 0 00 10 * 0 0 00 01 * 0 00 00.1 1 * 2 0 00 00.01 * 0 00 00.001 3 0 00 00.000... : * 2) The largest positive useful value for NDP is determined by the size of ANGLE, the format of DOUBLE PRECISION floating-point * * numbers on the target platform, and the risk of overflowing IDMSF(4). On a typical platform, for ANGLE up to 2pi, the available floating-point precision might correspond to NDP=12. * * However, the practical limit is typically NDP=9, set by the * capacity of a 32-bit IDMSF(4). 3) The absolute value of ANGLE may exceed 2pi. In cases where it does not, it is up to the caller to test for and handle the * * case where ANGLE is very nearly 2pi and rounds up to 360 degrees, * by testing for IDMSF(1)=360 and setting IDMSF(1-4) to zero. * *_

SUBROUTINE iau_A2TF (NDP, ANGLE, SIGN, IHMSF) *+ * iau_A2TF * * _ _ _ _ _ * Decompose radians into hours, minutes, seconds, fraction. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * Given: * NDP i resolution (Note 1) * ANGLE d angle in radians * * Returned: '+' or '-' * SIGN С i(4) * hours, minutes, seconds, fraction IHMSF * * Called: * iau_D2TF decompose days to hms * * Notes: * * 1) NDP is interpreted as follows: * NDP resolution * ...0000 00 00 : * -7 1000 00 00 * 100 00 00 -6 * 10 00 00 -5 * $^{-4}$ 1 00 00 * -3 0 10 00 * -2 0 01 00 * -1 0 00 10 * 0 0 00 01 * 0 00 00.1 1 * 2 0 00 00.01 * 0 00 00.001 3 0 00 00.000... : * * 2) The largest useful value for NDP is determined by the size of ANGLE, the format of DOUBLE PRECISION floating-point numbers on the target platform, and the risk of overflowing IHMSF(4). On a typical platform, for ANGLE up to 2pi, the available * * floating-point precision might correspond to NDP=12. However, * the practical limit is typically NDP=9, set by the capacity of * a 32-bit IHMSF(4). 3) The absolute value of ANGLE may exceed 2pi. In cases where it does not, it is up to the caller to test for and handle the * * case where ANGLE is very nearly 2pi and rounds up to 24 hours, * by testing for IHMSF(1)=24 and setting IHMSF(1-4) to zero. * *_

```
SUBROUTINE iau_AF2A ( S, IDEG, IAMIN, ASEC, RAD, J )
*+
*
   iau_AF2A
*
*
           _ _ _ _
*
*
  Convert degrees, arcminutes, arcseconds to radians.
*
*
   This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: support routine.
*
  Given:
                        sign: '-' = negative, otherwise positive
*
      S
                 С
*
      IDEG
                i
                      degrees
*
      IAMIN
                 i
                       arcminutes
*
      ASEC
                 d
                       arcseconds
*
*
  Returned:
*
                        angle in radians
     RAD
                 d
*
      J
                        status: 0 = OK
                 i
                                 1 = IDEG outside range 0-359
                                 2 = IAMIN outside range 0-59
                                 3 = ASEC outside range 0-59.999...
*
  Notes:
*
*
   1) If the s argument is a string, only the leftmost character is
*
       used and no warning status is provided.
*
*
   2) The result is computed even if any of the range checks fail.
*
*
   3)
      Negative IDEG, IAMIN and/or ASEC produce a warning status, but
*
      the absolute value is used in the conversion.
*
*
   4) If there are multiple errors, the status value reflects only the
*
       first, the smallest taking precedence.
*
*
*_
```

DOUBLE PRECISION FUNCTION iau_ANP (A) *+ * _ _ _ * iau_ANP * _ _ _ _ _ _ _ _ _ * * Normalize angle into the range 0 <= A < 2pi. * This routine is part of the International Astronomical Union's SOFA (Standards of Fundamental Astronomy) software collection. * * * * Status: vector/matrix support routine. * * Given: * d angle (radians) Α * * Returned: * iau_ANP d angle in range 0-2pi * *_

DOUBLE PRECISION FUNCTION iau_ANPM (A) *+ * _ _ * iau_ANPM * - - - - - - - - -* * Normalize angle into the range -pi <= A < +pi. * This routine is part of the International Astronomical Union's SOFA (Standards of Fundamental Astronomy) software collection. * * * * Status: vector/matrix support routine. * * Given: * d angle (radians) А * * Returned: * iau_ANPM d angle in range +/-pi * *_

SUBROUTINE iau_BIOO (DPSIBI, DEPSBI, DRA) *+ * iau_BI00 * * * Frame bias components of IAU 2000 precession-nutation models (part of * MHB2000 with additions). * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * Returned: * DPSIBI, DEPSBI d longitude and obliquity corrections Ы the ICRS RA of the J2000.0 mean equinox DRA * Notes: 1) The frame bias corrections in longitude and obliquity (radians) are required in order to correct for the offset between the GCRS * GCRS frame, a J2000.0 mean pole. They define, with respect to the GCRS frame, a J2000.0 mean pole that is consistent with the rest of the IAU 2000A precession-nutation model. 2) In addition to the displacement of the pole, the complete description of the frame bias requires also an offset in right ascension. This is not part of the IAU 2000A model, and is from Chapront et al. (2002). It is returned in radians. 3) This is a supplemented implementation of one aspect of the IAU 2000A nutation model, formally adopted by the IAU General Assembly in 2000, namely MHB2000 (Mathews et al. 2002). References: Chapront, J., Chapront-Touze, M. & Francou, G., Astron.Astrophys., 387, 700, 2002. * * Mathews, P.M., Herring, T.A., Buffet, B.A., "Modeling of nutation and precession New nutation series for nonrigid Earth and * * insights into the Earth's interior", J.Geophys.Res., 107, B4, * 2002. The MHB2000 code itself was obtained on 9th September 2002 from ftp://maia.usno.navy.mil/conv2000/chapter5/IAU2000A. * *_

SUBROUTINE iau_BP00 (DATE1, DATE2, RB, RP, RBP) *+ * iau_BP00 * * - - - - -* Frame bias and precession, IAU 2000. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * d(3,3) frame bias matrix (Note 2) RB precession matrix (Note 3) * RP d(3,3) * RBP d(3,3) bias-precession matrix (Note 4) * * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0(JD method) 2451545D0 -1421.3D0 (J2000 method) 50123.2D0 2400000.5D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * 2) The matrix RB transforms vectors from GCRS to mean J2000.0 by * applying frame bias. * 3) The matrix RP transforms vectors from J2000.0 mean equator and * equinox to mean equator and equinox of date by applying precession. * 4) The matrix RBP transforms vectors from GCRS to mean equator and equinox of date by applying frame bias then precession. It is the * product RP x RB. Called: iau_BI00 frame bias components, IAU 2000 IAU 2000 precession adjustments * iau_PR00 initialize r-matrix to identity iau IR rotate around X-axis iau_RX rotate around Y-axis iau_RY rotate around Z-axis iau RZ iau_RXR product of two r-matrices Reference: Capitaine, N., Chapront, J., Lambert, S. and Wallace, P., "Expressions for the Celestial Intermediate Pole and Celestial Ephemeris Origin consistent with the IAU 2000A precession-nutation model", Astron.Astrophys. 400, 1145-1154 (2003) n.b. The celestial ephemeris origin (CEO) was renamed "celestial intermediate origin" (CIO) by IAU 2006 Resolution 2.

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SUBROUTINE iau_BP06 (DATE1, DATE2, RB, RP, RBP) *+ * iau_BP06 * * _ _ _ _ * Frame bias and precession, IAU 2006. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) Returned: * d(3,3) frame bias matrix (Note 2) RB precession matrix (Note 3) * RP d(3,3) * RBP d(3,3) bias-precession matrix (Note 4) * * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: * DATE1 DATE2 2450123.7D0 0D0(JD method) -1421.3D0 (J2000 method) 2451545D0 50123.2D0 2400000.5D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * 2) The matrix RB transforms vectors from GCRS to mean J2000.0 by * applying frame bias. * 3) The matrix RP transforms vectors from mean J2000.0 to mean of date * by applying precession. * 4) The matrix RBP transforms vectors from GCRS to mean of date by * applying frame bias then precession. It is the product RP x RB. * Called: iau_PFW06 bias-precession F-W angles, IAU 2006 * iau FW2M F-W angles to r-matrix * iau_PMAT06 PB matrix, IAU 2006 * iau_TR transpose r-matrix product of two r-matrices iau_RXR References: * Capitaine, N. & Wallace, P.T., 2006, Astron.Astrophys. 450, 855 * * Wallace, P.T. & Capitaine, N., 2006, Astron.Astrophys. 459, 981 *_

SUBROUTINE iau_BPN2XY (RBPN, X, Y) *+ * iau_BPN2XY * * * Extract from the bias-precession-nutation matrix the X,Y coordinates * of the Celestial Intermediate Pole. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * RBPN d(3,3) celestial-to-true matrix (Note 1) * * Returned: * Celestial Intermediate Pole (Note 2) Х.Ү d * * Notes: * 1) The matrix RBPN transforms vectors from GCRS to true equator (and CIO or equinox) of date, and therefore the Celestial Intermediate Pole unit vector is the bottom row of the matrix. 2) X,Y are components of the Celestial Intermediate Pole unit vector * * in the Geocentric Celestial Reference System. Reference: * Capitaine, N., Chapront, J., Lambert, S. and Wallace, P., * "Expressions for the Celestial Intermediate Pole and Celestial * Ephemeris Origin consistent with the IAU 2000A precession-nutation model", Astron.Astrophys. 400, 1145-1154 (2003) * * n.b. The celestial ephemeris origin (CEO) was renamed "celestial intermediate origin" (CIO) by IAU 2006 Resolution 2. * * *_

SUBROUTINE iau_C2I00A (DATE1, DATE2, RC2I) *+ * i a u _ C 2 I 0 0 A * Form the celestial-to-intermediate matrix for a given date using the * IAU 2000A precession-nutation model. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) Returned: RC2I d(3,3) celestial-to-intermediate matrix (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. 2) The matrix RC2I is the first stage in the transformation from celestial to terrestrial coordinates: [TRS] = RPOM * R_3(ERA) * RC2I * [CRS] = RC2T * [CRS] where [CRS] is a vector in the Geocentric Celestial Reference * System and [TRS] is a vector in the International Terrestrial Reference System (see IERS Conventions 2003), ERA is the Earth Rotation Angle and RPOM is the polar motion matrix. 3) A faster, but slightly less accurate result (about 1 mas), can be obtained by using instead the iau_C2I00B routine. Called: iau PNM00A classical NPB matrix, IAU 2000A celestial-to-intermediate matrix, given NPB matrix iau_C2IBPN References: Capitaine, N., Chapront, J., Lambert, S. and Wallace, P., "Expressions for the Celestial Intermediate Pole and Celestial Ephemeris Origin consistent with the IAU 2000A precession-nutation model", Astron.Astrophys. 400, 1145-1154 (2003) n.b. The celestial ephemeris origin (CEO) was renamed "celestial * intermediate origin" (CIO) by IAU 2006 Resolution 2. * McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)

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SUBROUTINE iau_C2I00B (DATE1, DATE2, RC2I) *+ * i a u _ C 2 I 0 0 B * Form the celestial-to-intermediate matrix for a given date using the * IAU 2000B precession-nutation model. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) Returned: RC2I d(3,3) celestial-to-intermediate matrix (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. 2) The matrix RC2I is the first stage in the transformation from celestial to terrestrial coordinates: [TRS] = RPOM * R_3(ERA) * RC2I * [CRS] = RC2T * [CRS] where [CRS] is a vector in the Geocentric Celestial Reference * System and [TRS] is a vector in the International Terrestrial Reference System (see IERS Conventions 2003), ERA is the Earth Rotation Angle and RPOM is the polar motion matrix. 3) The present routine is faster, but slightly less accurate (about 1 mas), than the iau_C2I00A routine. Called: iau_PNM00B classical NPB matrix, IAU 2000B celestial-to-intermediate matrix, given NPB matrix iau_C2IBPN References: Capitaine, N., Chapront, J., Lambert, S. and Wallace, P., "Expressions for the Celestial Intermediate Pole and Celestial Ephemeris Origin consistent with the IAU 2000A precession-nutation model", Astron.Astrophys. 400, 1145-1154 (2003) n.b. The celestial ephemeris origin (CEO) was renamed "celestial * intermediate origin" (CIO) by IAU 2006 Resolution 2. * McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)

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SUBROUTINE iau_C2I06A (DATE1, DATE2, RC2I) *+ * i a u _ C 2 I 0 6 A * * * Form the celestial-to-intermediate matrix for a given date using the * IAU 2006 precession and IAU 2000A nutation models. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * RC2I d(3,3) celestial-to-intermediate matrix (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. 2) The matrix RC2I is the first stage in the transformation from celestial to terrestrial coordinates: [TRS] = RPOM * R_3(ERA) * RC2I * [CRS] = RC2T * [CRS] where [CRS] is a vector in the Geocentric Celestial Reference * System and [TRS] is a vector in the International Terrestrial Reference System (see IERS Conventions 2003), ERA is the Earth * Rotation Angle and RPOM is the polar motion matrix. Called: iau_PNM06A classical NPB matrix, IAU 2006/2000A extract CIP X,Y coordinates from NPB matrix * iau_BPN2XY the CIO locator s, given X,Y, IAU 2006 iau SO6 iau_C2IXYS celestial-to-intermediate matrix, given X,Y and s References: McCarthy, D. D., Petit, G. (eds.), 2004, IERS Conventions (2003), IERS Technical Note No. 32, BKG Capitaine, N. & Wallace, P.T., 2006, Astron.Astrophys. 450, 855 * Wallace, P.T. & Capitaine, N., 2006, Astron.Astrophys. 459, 981 *_

SUBROUTINE iau_C2IBPN (DATE1, DATE2, RBPN, RC2I) *+ * iau_C2IBPN * * * Form the celestial-to-intermediate matrix for a given date given * the bias-precession-nutation matrix. IAU 2000. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) d(3,3) RBPN celestial-to-true matrix (Note 2) Returned: d(3,3) celestial-to-intermediate matrix (Note 3) RC2I * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, ${\tt JD(TT)=}2450123.7$ could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 50123.2D0 2400000.5D0 (MJD method) 0.2D0 (date & time method) 2450123.5D0 The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. 2) The matrix RBPN transforms vectors from GCRS to true equator (and CIO or equinox) of date. Only the CIP (bottom row) is used. 3) The matrix RC2I is the first stage in the transformation from celestial to terrestrial coordinates: [TRS] = RPOM * R_3(ERA) * RC2I * [CRS] * = RC2T * [CRS] where [CRS] is a vector in the Geocentric Celestial Reference System and [TRS] is a vector in the International Terrestrial Reference System (see IERS Conventions 2003), ERA is the Earth Rotation Angle and RPOM is the polar motion matrix. * 4) Although its name does not include "00", this routine is in fact specific to the IAU 2000 models. Called: iau_BPN2XY extract CIP X,Y coordinates from NPB matrix * celestial-to-intermediate matrix, given X,Y iau_C2IXY References: Capitaine, N., Chapront, J., Lambert, S. and Wallace, P. "Expressions for the Celestial Intermediate Pole and Celestial Ephemeris Origin consistent with the IAU 2000A precession-nutation * model", Astron.Astrophys. 400, 1145-1154 (2003) n.b. The celestial ephemeris origin (CEO) was renamed "celestial

* intermediate origin" (CIO) by IAU 2006 Resolution 2.
*
* McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003),
* IERS Technical Note No. 32, BKG (2004)
*
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SUBROUTINE iau_C2IXY (DATE1, DATE2, X, Y, RC2I) *+ * iau_C2IXY * * Form the celestial to intermediate-frame-of-date matrix for a given * date when the CIP X,Y coordinates are known. IAU 2000. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * Given: * DATE1,DATE2 TT as a 2-part Julian Date (Note 1) d Celestial Intermediate Pole (Note 2) X,Y d Returned: d(3,3) celestial-to-intermediate matrix (Note 3) RC2I * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0(JD method) 2451545D0 -1421.3D0 (J2000 method) 50123.2D0 2400000.5D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. 2) The Celestial Intermediate Pole coordinates are the x,y components of the unit vector in the Geocentric Celestial Reference System. 3) The matrix RC2I is the first stage in the transformation from celestial to terrestrial coordinates: [TRS] = RPOM * R_3(ERA) * RC2I * [CRS] * = RC2T * [CRS] where [CRS] is a vector in the Geocentric Celestial Reference System and [TRS] is a vector in the International Terrestrial Reference System (see IERS Conventions 2003), ERA is the Earth Rotation Angle and RPOM is the polar motion matrix. * 4) Although its name does not include "00", this routine is in fact specific to the IAU 2000 models. Called: * iau_C2IXYS celestial-to-intermediate matrix, given X,Y and s the CIO locator s, given X,Y, IAU 2000A iau_S00 * Reference: McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * * *_

```
SUBROUTINE iau_C2IXYS ( X, Y, S, RC2I )
*+
*
   iau_C2IXYS
*
*
*
  Form the celestial to intermediate-frame-of-date matrix given the CIP
*
  X,Y and the CIO locator s.
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
   Status: support routine.
*
*
   Given:
*
                  d
                           Celestial Intermediate Pole (Note 1)
     Х,Ү
*
      S
                  d
                          the CIO locator s (Note 2)
*
   Returned:
*
                d(3,3) celestial-to-intermediate matrix (Note 3)
      RC2I
*
*
  Notes:
*
   1) The Celestial Intermediate Pole coordinates are the x,y components
      of the unit vector in the Geocentric Celestial Reference System.
   2) The CIO locator s (in radians) positions the Celestial
      Intermediate Origin on the equator of the CIP.
*
   3) The matrix RC2I is the first stage in the transformation from
      celestial to terrestrial coordinates:
          [TRS] = RPOM * R_3(ERA) * RC2I * [CRS]
                 = RC2T * [CRS]
      where [CRS] is a vector in the Geocentric Celestial Reference
      System and [TRS] is a vector in the International Terrestrial Reference System (see IERS Conventions 2003), ERA is the Earth
*
      Rotation Angle and RPOM is the polar motion matrix.
*
   Called:
*
      iau_IR
                   initialize r-matrix to identity
*
      iau RZ
                    rotate around Z-axis
                    rotate around Y-axis
      iau_RY
*
*
  Reference:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
*_
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SUBROUTINE iau_C2S ( P, THETA, PHI )
*+
*
        _ _
  iau_C2S
*
*
  _ _ _ _ _ _ _ _ _
*
*
  P-vector to spherical coordinates.
*
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
  Status: vector/matrix support routine.
*
*
  Given:
*
              d(3)
                       p-vector
     Ρ
*
*
  Returned:
*
                        longitude angle (radians)
     THETA
              d
*
     PHI
                        latitude angle (radians)
              d
*
*
  Notes:
*
*
  1) P can have any magnitude; only its direction is used.
*
*
  2) If P is null, zero THETA and PHI are returned.
*
  3) At either pole, zero THETA is returned.
*
*
*_
```

SUBROUTINE iau_C2T00A (TTA, TTB, UTA, UTB, XP, YP, RC2T) *+ * i a u _ C 2 T 0 0 A * * Form the celestial to terrestrial matrix given the date, the UT1 and * the polar motion, using the IAU 2000A nutation model. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: TTA,TTB * d TT as a 2-part Julian Date (Note 1) UT1 as a 2-part Julian Date (Note 1) UTA,UTB d XP,YP d coordinates of the pole (radians, Note 2) * Returned: RC2T d(3,3) celestial-to-terrestrial matrix (Note 3) * Notes: 1) The TT and UT1 dates TTA+TTB and UTA+UTB are Julian Dates, apportioned in any convenient way between the arguments UTA and UTB. For example, JD(UT1)=2450123.7 could be expressed in any of these ways, among others: * UTA UTB 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is The J2000 and MJD methods are good compromises acceptable. between resolution and convenience. In the case of UTA, UTB, the date & time method is best matched to the Earth rotation angle * algorithm used: maximum accuracy (or, at least, minimum noise) is * delivered when the UTA argument is for Ohrs UT1 on the day in * question and the UTB argument lies in the range 0 to 1, or vice versa. 2) XP and YP are the coordinates (in radians) of the Celestial Intermediate Pole with respect to the International Terrestrial Reference System (see IERS Conventions 2003), measured along the * meridians to 0 and 90 deg west respectively. 3) The matrix RC2T transforms from celestial to terrestrial coordinates: [TRS] = RPOM * R_3(ERA) * RC2I * [CRS] = RC2T * [CRS] where [CRS] is a vector in the Geocentric Celestial Reference System and [TRS] is a vector in the International Terrestrial Reference System (see IERS Conventions 2003), RC2I is the celestial-to-intermediate matrix, ERA is the Earth rotation angle and RPOM is the polar motion matrix. 4) A faster, but slightly less accurate result (about 1 mas), can be obtained by using instead the iau_C2T00B routine. Called: * iau_C2I00A celestial-to-intermediate matrix, IAU 2000A iau_ERA00 Earth rotation angle, IAU 2000 * iau_SP00 the TIO locator s', IERS 2000 iau POM00 polar motion matrix

```
* iau_C2TCIO form CIO-based celestial-to-terrestrial matrix
*
* Reference:
*
* McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003),
* IERS Technical Note No. 32, BKG (2004)
*
*_
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SUBROUTINE iau_C2T00B (TTA, TTB, UTA, UTB, XP, YP, RC2T) *+ * i a u _ C 2 T 0 0 B * * Form the celestial to terrestrial matrix given the date, the UT1 and * the polar motion, using the IAU 2000B nutation model. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: TTA,TTB * d TT as a 2-part Julian Date (Note 1) UT1 as a 2-part Julian Date (Note 1) UTA,UTB d XP,YP d coordinates of the pole (radians, Note 2) * Returned: RC2T d(3,3) celestial-to-terrestrial matrix (Note 3) * Notes: 1) The TT and UT1 dates TTA+TTB and UTA+UTB are Julian Dates, apportioned in any convenient way between the arguments UTA and UTB. For example, JD(UT1)=2450123.7 could be expressed in any of these ways, among others: * UTA UTB 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is The J2000 and MJD methods are good compromises acceptable. between resolution and convenience. In the case of UTA, UTB, the date & time method is best matched to the Earth rotation angle * algorithm used: maximum accuracy (or, at least, minimum noise) is * delivered when the UTA argument is for Ohrs UT1 on the day in * question and the UTB argument lies in the range 0 to 1, or vice versa. 2) XP and YP are the coordinates (in radians) of the Celestial Intermediate Pole with respect to the International Terrestrial Reference System (see IERS Conventions 2003), measured along the * meridians to 0 and 90 deg west respectively. 3) The matrix RC2T transforms from celestial to terrestrial coordinates: [TRS] = RPOM * R_3(ERA) * RC2I * [CRS] = RC2T * [CRS] where [CRS] is a vector in the Geocentric Celestial Reference System and [TRS] is a vector in the International Terrestrial Reference System (see IERS Conventions 2003), RC2I is the celestial-to-intermediate matrix, ERA is the Earth rotation angle and RPOM is the polar motion matrix. 4) The present routine is faster, but slightly less accurate (about 1 mas), than the iau_C2T00A routine. Called: * iau_C2I00B celestial-to-intermediate matrix, IAU 2000B iau_ERA00 Earth rotation angle, IAU 2000 * iau_POM00 polar motion matrix form CIO-based celestial-to-terrestrial matrix iau_C2TCIO

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* 
* Reference:
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McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003),
* IERS Technical Note No. 32, BKG (2004)
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SUBROUTINE iau_C2T06A (TTA, TTB, UTA, UTB, XP, YP, RC2T) *+ * i a u _ C 2 T 0 6 A * Form the celestial to terrestrial matrix given the date, the UT1 and * the polar motion, using the IAU 2006 precession and IAU 2000A * nutation models. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: TT as a 2-part Julian Date (Note 1) TTA,TTB d UTA,UTB d UT1 as a 2-part Julian Date (Note 1) * XP,YP d coordinates of the pole (radians, Note 2) Returned: * RC2T d(3,3) celestial-to-terrestrial matrix (Note 3) * Notes: 1) The TT and UT1 dates TTA+TTB and UTA+UTB are Julian Dates, apportioned in any convenient way between the arguments UTA and UTB. For example, JD(UT1)=2450123.7 could be expressed in any of these ways, among others: UTA UTB 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 and MJD methods are good compromises * between resolution and convenience. In the case of UTA, UTB, the date & time method is best matched to the Earth rotation angle algorithm used: maximum accuracy (or, at least, minimum noise) is delivered when the UTA argument is for Ohrs UT1 on the day in question and the UTB argument lies in the range 0 to 1, or vice versa. 2) XP and YP are the coordinates (in radians) of the Celestial Intermediate Pole with respect to the International Terrestrial * Reference System (see IERS Conventions 2003), measured along the meridians to 0 and 90 deg west respectively. * 3) The matrix RC2T transforms from celestial to terrestrial coordinates: [TRS] = RPOM * R_3(ERA) * RC2I * [CRS] = RC2T * [CRS] where [CRS] is a vector in the Geocentric Celestial Reference System and [TRS] is a vector in the International Terrestrial Reference System (see IERS Conventions 2003), RC2I is the celestial-to-intermediate matrix, ERA is the Earth rotation angle and RPOM is the polar motion matrix. Called: iau_C2I06A celestial-to-intermediate matrix, IAU 2006/2000A Earth rotation angle, IAU 2000 iau_ERA00 * iau_SP00 the TIO locator s', IERS 2000 iau_POM00 polar motion matrix form CIO-based celestial-to-terrestrial matrix * iau_C2TCIO

```
* Reference:
*
* McCarthy, D. D., Petit, G. (eds.), 2004, IERS Conventions (2003),
* IERS Technical Note No. 32, BKG
*
*_
```

SUBROUTINE iau_C2TCEO (RC2I, ERA, RPOM, RC2T) *+ * iau_C2TCEO * * * Assemble the celestial to terrestrial matrix from CIO-based * components (the celestial-to-intermediate matrix, the Earth Rotation * Angle and the polar motion matrix). * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: obsolete routine. * * Given: * d(3,3) celestial-to-intermediate matrix RC2I * ERA d Earth rotation angle * d(3,3) RPOM polar-motion matrix * * Returned: * RC2T d(3,3) celestial-to-terrestrial matrix * Notes: * 1) The name of the present routine, iau_C2TCEO, reflects the original name of the celestial intermediate origin (CIO), which before the * * adoption of IAU 2006 Resolution 2 was called the "celestial ephemeris origin" (CEO). * 2) When the name change from CEO to CIO occurred, a new SOFA routine * called iau_C2TCIO was introduced as the successor to the existing * iau_C2TCEO. The present routine is merely a front end to the new * one. * 3) The present routine is included in the SOFA collection only to * support existing applications. It should not be used in new * applications. * * Called: * iau_C2TCI0 form CIO-based celestial-to-terrestrial matrix *_

SUBROUTINE iau_C2TCIO (RC2I, ERA, RPOM, RC2T) *+ * i a u _ C 2 T C I O * * * Assemble the celestial to terrestrial matrix from CIO-based * components (the celestial-to-intermediate matrix, the Earth Rotation * Angle and the polar motion matrix). This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * * Given: d(3,3) celestial-to-intermediate matrix RC2I * ERA d Earth rotation angle * RPOM d(3,3) polar-motion matrix * Returned: * RC2T d(3,3) celestial-to-terrestrial matrix * Notes: 1) This routine constructs the rotation matrix that transforms vectors in the celestial system into vectors in the terrestrial system. It does so starting from precomputed components, namely the matrix which rotates from celestial coordinates to the intermediate frame, the Earth rotation angle and the polar motion matrix. One use of the present routine is when generating a series of celestial-to-terrestrial matrices where only the Earth Rotation Angle changes, avoiding the considerable overhead of recomputing the precession-nutation more often than necessary to achieve given accuracy objectives. 2) The relationship between the arguments is as follows: [TRS] = RPOM * R_3(ERA) * RC2I * [CRS] * * = RC2T * [CRS] * where [CRS] is a vector in the Geocentric Celestial Reference * System and [TRS] is a vector in the International Terrestrial Reference System (see IERS Conventions 2003). * * Called: iau_CR copy r-matrix * iau_RZ rotate around Z-axis * iau_RXR product of two r-matrices * Reference: McCarthy, D. D., Petit, G. (eds.), 2004, IERS Conventions (2003), * IERS Technical Note No. 32, BKG * *_

SUBROUTINE iau_C2TEQX (RBPN, GST, RPOM, RC2T) *+ * iau_C2TEQX * * * Assemble the celestial to terrestrial matrix from equinox-based * components (the celestial-to-true matrix, the Greenwich Apparent * Sidereal Time and the polar motion matrix). * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * * Given: d(3,3) RBPN celestial-to-true matrix * GST Greenwich (apparent) Sidereal Time d * RPOM d(3,3) polar-motion matrix * Returned: * RC2T d(3,3) celestial-to-terrestrial matrix (Note 2) * Notes: 1) This routine constructs the rotation matrix that transforms vectors in the celestial system into vectors in the terrestrial system. It does so starting from precomputed components, namely the matrix which rotates from celestial coordinates to the true equator and equinox of date, the Greenwich Apparent Sidereal Time and the polar motion matrix. One use of the present routine is when generating a series of celestial-to-terrestrial matrices where only the Sidereal Time changes, avoiding the considerable overhead of recomputing the precession-nutation more often than necessary to achieve given accuracy objectives. 2) The relationship between the arguments is as follows: $[TRS] = RPOM * R_3(GST) * RBPN * [CRS]$ * = RC2T * [CRS] * * where [CRS] is a vector in the Geocentric Celestial Reference * System and [TRS] is a vector in the International Terrestrial Reference System (see IERS Conventions 2003). * Called: iau_CR copy r-matrix * iau_RZ rotate around Z-axis * iau_RXR product of two r-matrices * Reference: McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), * IERS Technical Note No. 32, BKG (2004) * *_

SUBROUTINE iau_C2TPE (TTA, TTB, UTA, UTB, DPSI, DEPS, XP, YP, : RC2T) *+ * _ _ * iau_C2TPE * * Form the celestial to terrestrial matrix given the date, the UT1, the * nutation and the polar motion. IAU 2000. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * * Given: TTA,TTB d TT as a 2-part Julian Date (Note 1) UTA,UTB * UT1 as a 2-part Julian Date (Note 1) d nutation (Note 2) * DPSI, DEPS d * coordinates of the pole (radians, Note 3) XP,YP d * * Returned: * d(3,3) celestial-to-terrestrial matrix (Note 4) RC2T * Notes: 1) The TT and UT1 dates TTA+TTB and UTA+UTB are Julian Dates, apportioned in any convenient way between the arguments $\stackrel{\cdot}{\text{UTA}}$ and * UTB. For example, JD(UT1)=2450123.7 could be expressed in any of these ways, among others: UTB UTA 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is * acceptable. The J2000 and MJD methods are good compromises between resolution and convenience. In the case of UTA, UTB, the * date & time method is best matched to the Earth rotation angle algorithm used: maximum accuracy (or, at least, minimum noise) is delivered when the UTA argument is for Ohrs UT1 on the day in * * question and the UTB argument lies in the range 0 to 1, or vice * versa. 2) The caller is responsible for providing the nutation components; they are in longitude and obliquity, in radians and are with * respect to the equinox and ecliptic of date. For high-accuracy * applications, free core nutation should be included as well as any other relevant corrections to the position of the CIP. 3) XP and YP are the coordinates (in radians) of the Celestial * Intermediate Pole with respect to the International Terrestrial Reference System (see IERS Conventions 2003), measured along the meridians to 0 and 90 deg west respectively. 4) The matrix RC2T transforms from celestial to terrestrial coordinates: $[TRS] = RPOM * R_3(GST) * RBPN * [CRS]$ = RC2T * [CRS] where [CRS] is a vector in the Geocentric Celestial Reference System and [TRS] is a vector in the International Terrestrial * Reference System (see IERS Conventions 2003), RBPN is the bias-precession-nutation matrix, GST is the Greenwich (apparent) Sidereal Time and RPOM is the polar motion matrix.

*	5) Although its name does not include "00", this routine is in fact
*	specific to the IAU 2000 models.
*	-
*	Called:
*	iau_PN00 bias/precession/nutation results, IAU 2000
*	iau GMST00 Greenwich mean sidereal time, IAU 2000
*	iau SP00 the TIO locator s', IERS 2000
*	iau EE00 equation of the equinoxes, IAU 2000
*	iau POM00 polar motion matrix
*	iau_C2TEQX form equinox-based celestial-to-terrestrial matrix
*	
*	Reference:
*	
*	McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003),
*	IERS Technical Note No. 32, BKG (2004)
*	
*_	

SUBROUTINE iau_C2TXY (TTA, TTB, UTA, UTB, X, Y, XP, YP, RC2T) *+ * іаи_С2ТХҮ * Form the celestial to terrestrial matrix given the date, the UT1, the * CIP coordinates and the polar motion. IAU 2000. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * TTA,TTB d TT as a 2-part Julian Date (Note 1) UTA,UTB d UT1 as a 2-part Julian Date (Note 1) X,Y d Celestial Intermediate Pole (Note 2) XP,YP d coordinates of the pole (radians, Note 3) Returned: * RC2T d(3,3) celestial-to-terrestrial matrix (Note 4) * Notes: 1) The TT and UT1 dates TTA+TTB and UTA+UTB are Julian Dates, apportioned in any convenient way between the arguments UTA and UTB. For example, JD(UT1)=2450123.7 could be expressed in any of these ways, among others: UTA UTB 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 and MJD methods are good compromises * between resolution and convenience. In the case of UTA, UTB, the date & time method is best matched to the Earth rotation angle algorithm used: maximum accuracy (or, at least, minimum noise) is delivered when the UTA argument is for Ohrs UT1 on the day in question and the UTB argument lies in the range 0 to 1, or vice versa. 2) The Celestial Intermediate Pole coordinates are the x,y components of the unit vector in the Geocentric Celestial Reference System. 3) XP and YP are the coordinates (in radians) of the Celestial Intermediate Pole with respect to the International Terrestrial Reference System (see IERS Conventions 2003), measured along the meridians to 0 and 90 deg west respectively. 4) The matrix RC2T transforms from celestial to terrestrial coordinates: [TRS] = RPOM * R_3(ERA) * RC2I * [CRS] = RC2T * [CRS] where [CRS] is a vector in the Geocentric Celestial Reference System and [TRS] is a vector in the International Terrestrial Reference System (see IERS Conventions 2003), ERA is the Earth Rotation Angle and RPOM is the polar motion matrix. 5) Although its name does not include "00", this routine is in fact * specific to the IAU 2000 models. * Called: iau_C2IXY celestial-to-intermediate matrix, given X,Y

```
* iau_ERA00 Earth rotation angle, IAU 2000
* iau_SP00 the TIO locator s', IERS 2000
* iau_POM00 polar motion matrix
* iau_C2TCIO form CIO-based celestial-to-terrestrial matrix
*
* Reference:
*
* McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003),
* IERS Technical Note No. 32, BKG (2004)
*
```

```
SUBROUTINE iau_CAL2JD ( IY, IM, ID, DJM0, DJM, J )
*+
*
   iau_CAL2JD
*
*
           _ _ _ _ _ _
*
*
   Gregorian Calendar to Julian Date.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: support routine.
*
  Given:
*
      IY,IM,ID
                  i
                        year, month, day in Gregorian calendar (Note 1)
*
*
   Returned:
*
      DJM0
                  d
                        MJD zero-point: always 2400000.5
*
                         Modified Julian Date for 0 hrs
      DJM
                  d
*
      J
                   i
                         status:
*
                             0 = OK
*
                            -1 = bad year
                                            (Note 3: JD not computed)
                            -2 = bad month (JD not computed)
                            -3 = bad day
                                            (JD computed)
*
*
  Notes:
*
*
   1) The algorithm used is valid from -4800 March 1, but this
      implementation rejects dates before -4799 January 1.
   2) The Julian Date is returned in two pieces, in the usual SOFA
*
*
      manner, which is designed to preserve time resolution.
                                                                The
*
      Julian Date is available as a single number by adding DJMO and
*
      DJM.
   3) In early eras the conversion is from the "Proleptic Gregorian
      Calendar"; no account is taken of the date(s) of adoption of
*
*
      the Gregorian Calendar, nor is the \ensuremath{\operatorname{AD/BC}} numbering convention
      observed.
*
*
   Reference:
*
      Explanatory Supplement to the Astronomical Almanac,
*
      P. Kenneth Seidelmann (ed), University Science Books (1992),
*
      Section 12.92 (p604).
*
*_
```

SUBROUTINE iau_CP (P, C) *+ * - - - - i a u _ C P * * * * Copy a p-vector. * This routine is part of the International Astronomical Union's SOFA (Standards of Fundamental Astronomy) software collection. * * * * Status: vector/matrix support routine. * * Given: * d(3) p-vector to be copied Ρ * * Returned: * d(3) copy С * *_

SUBROUTINE iau_CPV (PV, C) *+ * _ _ iau_CPV * * _ _ _ _ _ _ _ _ _ * * Copy a position/velocity vector. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * d(3,2) position/velocity vector to be copied PV * * Returned: * d(3,2) copy С * * Called: * iau_CP copy p-vector * *_

```
SUBROUTINE iau_CR ( R, C )
*+
*
    - - - - -
  iau_CR
*
*
*
*
  Copy an r-matrix.
*
*
  This routine is part of the International Astronomical Union's
* SOFA (Standards of Fundamental Astronomy) software collection.
*
*
  Status: vector/matrix support routine.
*
*
  Given:
*
            d(3,3) r-matrix to be copied
    R
*
*
  Returned:
*
            d(3,3) copy
   С
*
* Called:
*
    iau_CP copy p-vector
*
*_
```

```
SUBROUTINE iau_D2DTF ( SCALE, NDP, D1, D2, IY, IM, ID, IHMSF, J )
*+
*
   iau_D2DTF
*
*
*
  Format for output a 2-part Julian Date (or in the case of UTC a
*
   quasi-JD form that includes special provision for leap seconds).
*
   This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
  Status: support routine.
*
*
   Given:
*
      SCALE
                 C*(*)
                        time scale ID (Note 1)
      NDP
                 i
                        resolution (Note 2)
      D1,D2
                 d
                         time as a 2-part Julian Date (Notes 3,4)
*
  Returned:
                        year, month, day in Gregorian calendar (Note 5)
*
      IY, IM, ID
                 i
*
      IHMSF
                 i(4)
                        hours, minutes, seconds, fraction (Note 1)
                        status: +1 = dubious year (Note 5)
      ιT
                 i
                                 0 = OK
                                 -1 = unacceptable date (Note 6)
*
  Notes:
  1) SCALE identifies the time scale. Only the value 'UTC' (in upper
      case) is significant, and enables handling of leap seconds (see
      Note 4).
   2) NDP is the number of decimal places in the seconds field, and can
*
     have negative as well as positive values, such as:
      NDP
                  resolution
*
      ^{-4}
                    1 00 00
*
      -3
                    0 10 00
*
                    0 01 00
      -2
*
      -1
                    0 00 10
*
       0
                    0 00 01
                    0 00 00.1
       1
*
       2
                    0 00 00.01
*
       3
                    0 00 00.001
*
      The limits are platform dependent, but a safe range is -5 to +9.
   3) D1+D2 is Julian Date, apportioned in any convenient way between
      the two arguments, for example where D1 is the Julian Day Number
*
      and D2 is the fraction of a day. In the case of UTC, where the
      use of JD is problematical, special conventions apply: see the
*
      next note.
   4) JD cannot unambiguously represent UTC during a leap second unless
      special measures are taken. The SOFA internal convention is that
*
      the quasi-JD day represents UTC days whether the length is 86399,
*
      86400 or 86401 SI seconds.
   5) The warning status "dubious year" flags UTCs that predate the
      introduction of the time scale and that are too far in the future
      to be trusted. See iau_DAT for further details.
*
   6) For calendar conventions and limitations, see iau_CAL2JD.
*
   Called:
*
                   JD to Gregorian calendar
      iau JD2CAL
*
                   decompose days to hms
delta(AT) = TAI-UTC
      iau_D2TF
*
      iau_DAT
*
*_
```

SUBROUTINE iau_D2TF (NDP, DAYS, SIGN, IHMSF) *+ * iau_D2TF * * _ _ _ _ _ * * Decompose days to hours, minutes, seconds, fraction. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * Given: * NDP i resolution (Note 1) * DAYS d interval in days * * Returned: '+' or '-' * SIGN С i(4) * hours, minutes, seconds, fraction IHMSF * * Notes: * 1) NDP is interpreted as follows: * NDP resolution * ...0000 00 00 : * -7 1000 00 00 100 00 00 10 00 00 * -6 -5 * * $^{-4}$ 1 00 00 * -3 0 10 00 * 0 01 00 -2 * -1 0 00 10 * 0 0 00 01 * 0 00 00.1 1 * 2 0 00 00.01 * 3 0 00 00.001 * 0 00 00.000... * 2) The largest positive useful value for NDP is determined by the size of DAYS, the format of DOUBLE PRECISION floating-point * * numbers on the target platform, and the risk of overflowing * IHMSF(4). On a typical platform, for DAYS up to 1D0, the available floating-point precision might correspond to NDP=12. * However, the practical limit is typically NDP=9, set by the * capacity of a 32-bit IHMSF(4). * 3) The absolute value of DAYS may exceed 1D0. In cases where it does not, it is up to the caller to test for and handle the * case where DAYS is very nearly 1D0 and rounds up to 24 hours, * by testing for IHMSF(1)=24 and setting IHMSF(1-4) to zero. * *_

SUBROUTINE iau_DAT (IY, IM, ID, FD, DELTAT, J) *+ * iau_DAT * _ _ _ * For a given UTC date, calculate delta(AT) = TAI-UTC. * * :-----: IMPORTANT : * : A new version of this routine must be : produced whenever a new leap second is * : announced. There are five items to change on each such occasion: : : : 1) The parameter NDAT must be : increased by 1. : 2) The set of DATA statements that * : * : initialize the arrays IDAT and DATS must be extended by one line. : : : 3) The parameter IYV must be set to : : the current year. * : * 4) The "Latest leap second" comment : below must be set to the new leap : : : second date. * : 5) The "This revision" comment, later, : * : must be set to the current date. * : : Change (3) must also be carried out : whenever the routine is re-issued, * even if no leap seconds have been : * : added. * : * : Latest leap second: 2012 June 30 * : * * This routine is part of the International Astronomical Union's SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * Given: * ΙY i UTC: year (Notes 1 and 2) month (Note 2) ΤМ i * ID i day (Notes 2 and 3) FD fraction of day (Note 4) * d * Returned: DELTAT d TAI minus UTC, seconds * status (Note 5): J i 1 = dubious year (Note 1) 0 = OK -1 = bad year-2 = bad month-3 = bad day (Note 3)-4 = bad fraction (Note 4) * Notes: 1) UTC began at 1960 January 1.0 (JD 2436934.5) and it is improper * to call the routine with an earlier date. If this is attempted, * zero is returned together with a warning status. Because leap seconds cannot, in principle, be predicted in advance, a reliable check for dates beyond the valid range is *

impossible. To guard against gross errors, a year five or more after the release year of the present routine (see parameter IYV) * is considered dubious. In this case a warning status is returned * but the result is computed in the normal way. * For both too-early and too-late years, the warning status is J=+1. This is distinct from the error status J=-1, which signifies a * year so early that JD could not be computed. * 2) If the specified date is for a day which ends with a leap second, the UTC-TAI value returned is for the period leading up to the leap second. If the date is for a day which begins as a leap second ends, the UTC-TAI returned is for the period following the leap second. 3) The day number must be in the normal calendar range, for example * 1 through 30 for April. The "almanac" convention of allowing such dates as January 0 and December 32 is not supported in this routine, in order to avoid confusion near leap seconds. 4) The fraction of day is used only for dates before the introduction of leap seconds, the first of which occurred at the end of 1971. It is tested for validity (0 to 1 is the valid range) even if not used; if invalid, zero is used and status J=-4 is returned. For many applications, setting FD to zero is acceptable; the resulting error is always less than 3 ms (and occurs only pre-1972). 5) The status value returned in the case where there are multiple errors refers to the first error detected. For example, if the month and day are 13 and 32 respectively, J=-2 (bad month) will be * returned. 6) In cases where a valid result is not available, zero is returned. * References: 1) For dates from 1961 January 1 onwards, the expressions from the file ftp://maia.usno.navy.mil/ser7/tai-utc.dat are used. 2) The 5ms timestep at 1961 January 1 is taken from 2.58.1 (p87) of * the 1992 Explanatory Supplement. * Called: * iau CAL2JD Gregorian calendar to Julian Day number *_

DOUBLE PRECISION FUNCTION iau_DTDB (DATE1, DATE2, : UT, ELONG, U, V) *+ * _ _ _ * iau_DTDB * * An approximation to TDB-TT, the difference between barycentric * dynamical time and terrestrial time, for an observer on the Earth. * The different time scales - proper, coordinate and realized - are * related to each other: TAI <- physically realized * offset <- observed (nominally +32.184s) : TT<- terrestrial time : <- definition of TT rate adjustment (L_G) * TCG <- time scale for GCRS : <- iau_DTDB is an implementation "periodic" terms rate adjustment (L_C) <- function of solar-system ephemeris : * TCB <- time scale for BCRS : rate adjustment (-L_B) <- definition of TDB : TDB <- TCB scaled to track TT • "periodic" terms <- -iau_DTDB is an approximation * : <- terrestrial time TT* Adopted values for the various constants can be found in the IERS * Conventions (McCarthy & Petit 2003). * This routine is part of the International Astronomical Union's SOFA (Standards of Fundamental Astronomy) software collection. * * Status: support routine. * * Given: * DATE1,DATE2 d date, TDB (Notes 1-3) universal time (UT1, fraction of one day) UΤ d * ELONG d longitude (east positive, radians) * IJ d distance from Earth spin axis (km) distance north of equatorial plane (km) V d * * Returned: * d TDB-TT (seconds) iau_DTDB * * Notes: * 1) The date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the arguments DATE1 and DATE2. For * example, JD(TDB)=2450123.7 could be expressed in any of these ways, among others: * DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 50123.2D0 2400000.5D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) * * The JD method is the most natural and convenient to use in cases * where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the

argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both * good compromises between resolution and convenience. * Although the date is, formally, barycentric dynamical time (TDB), * the terrestrial dynamical time (TT) can be used with no practical effect on the accuracy of the prediction. 2) TT can be regarded as a coordinate time that is realized as an offset of 32.184s from International Atomic Time, TAI. TT is a specific linear transformation of geocentric coordinate time TCG, which is the time scale for the Geocentric Celestial Reference System, GCRS. 3) TDB is a coordinate time, and is a specific linear transformation * of barycentric coordinate time TCB, which is the time scale for the Barycentric Celestial Reference System, BCRS. 4) The difference TCG-TCB depends on the masses and positions of the bodies of the solar system and the velocity of the Earth. It is dominated by a rate difference, the residual being of a periodic character. The latter, which is modeled by the present routine, comprises a main (annual) sinusoidal term of amplitude approximately 0.00166 seconds, plus planetary terms up to about 20 microseconds, and lunar and diurnal terms up to 2 microseconds. These effects come from the changing transverse Doppler effect and gravitational red-shift as the observer (on the Earth's surface) experiences variations in speed (with respect to the BCRS) and gravitational potential. 5) TDB can be regarded as the same as TCB but with a rate adjustment to keep it close to TT, which is convenient for many applications. The history of successive attempts to define TDB is set out in Resolution 3 adopted by the IAU General Assembly in 2006, which defines a fixed TDB(TCB) transformation that is consistent with contemporary solar-system ephemerides. Future ephemerides will imply slightly changed transformations between TCG and TCB, which could introduce a linear drift between TDB and TT; however, any such drift is unlikely to exceed 1 nanosecond per century. 6) The geocentric TDB-TT model used in the present routine is that of Fairhead & Bretagnon (1990), in its full form. It was originally supplied by Fairhead (private communications with P.T.Wallace, 1990) as a Fortran subroutine. The present routine contains an adaptation of the Fairhead code. The numerical results are essentially unaffected by the changes, the differences with respect to the Fairhead & Bretagnon original being at the 1D-20 s * level. The topocentric part of the model is from Moyer (1981) and Murray (1983), with fundamental arguments adapted from * Simon et al. 1994. It is an approximation to the expression (v / c) . (r / c), where v is the barycentric velocity of the Earth, r is the geocentric position of the observer and * * c is the speed of light. By supplying zeroes for U and V, the topocentric part of the model can be nullified, and the routine will return the Fairhead & Bretagnon result alone. 7) During the interval 1950-2050, the absolute accuracy is better than +/-3 nanoseconds relative to time ephemerides obtained by direct numerical integrations based on the JPL DE405 solar system ephemeris. 8) It must be stressed that the present routine is merely a model, and that numerical integration of solar-system ephemerides is the definitive method for predicting the relationship between TCG and TCB and hence between TT and TDB. References: Fairhead, L., & Bretagnon, P., Astron.Astrophys., 229, 240-247

* (1990). * * IAU 2006 Resolution 3. * McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * * * Moyer, T.D., Cel.Mech., 23, 33 (1981). * * Murray, C.A., Vectorial Astrometry, Adam Hilger (1983). * * Seidelmann, P.K. et al., Explanatory Supplement to the Astronomical Almanac, Chapter 2, University Science Books (1992). * * * Simon, J.L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G. & Laskar, J., Astron.Astrophys., 282, 663-683 (1994). * * *_

```
SUBROUTINE iau_DTF2D ( SCALE, IY, IM, ID, IHR, IMN, SEC,
     :
                              D1, D2, J )
*+
*
*
   iau_DTF2D
*
*
   Encode date and time fields into 2-part Julian Date (or in the case
*
   of UTC a quasi-JD form that includes special provision for leap
   seconds).
   This routine is part of the International Astronomical Union's
   SOFA (Standards of Fundamental Astronomy) software collection.
*
   Status: support routine.
*
*
   Given:
*
                  c*(*) time scale ID (Note 1)
      SCALE
                         year, month, day in Gregorian calendar (Note 2)
*
      IY, IM, ID
                  i
*
      IHR, IMN
                  i
                         hour, minute
      SEC
                  d
                         seconds
*
   Returned:
      D1,D2
                  d
                         2-part Julian Date (Notes 3,4)
                         status: +3 = both of next two
      J
                                  +2 = time is after end of day (Note 5)
                                  +1 = dubious year (Note 6)
                                   0 = OK
                                  -1 = bad year
                                  -2 = bad month
                                  -3 = bad day
                                  -4 = bad hour
                                  -5 = bad minute
                                  -6 = bad second (<0)
  Notes:
   1) SCALE identifies the time scale. Only the value 'UTC' (in upper case) is significant, and enables handling of leap seconds (see
*
      Note 4).
   2) For calendar conventions and limitations, see iau_CAL2JD.
   3) The sum of the results, D1+D2, is Julian Date, where normally D1
      is the Julian Day Number and D2 is the fraction of a day. In the
      case of UTC, where the use of JD is problematical, special
      conventions apply: see the next note.
   4) JD cannot unambiguously represent UTC during a leap second unless
*
      special measures are taken. The SOFA internal convention is that
      the quasi-JD day represents UTC days whether the length is 86399,
*
      86400 or 86401 SI seconds.
   5) The warning status "time is after end of day" usually means that
      the SEC argument is greater than 60D0. However, in a day ending
      in a leap second the limit changes to 61D0 (or 59D0 in the case of
*
      a negative leap second).
   6) The warning status "dubious year" flags UTCs that predate the
      introduction of the time scale and that are too far in the future
      to be trusted. See iau_DAT for further details.
   7) Only in the case of continuous and regular time scales (TAI, TT,
      TCG, TCB and TDB) is the result D1+D2 a Julian Date, strictly
      speaking. In the other cases (UT1 and UTC) the result must be
      used with circumspection; in particular the difference between
two such results cannot be interpreted as a precise time
      interval.
*
*
   Called:
*
      iau_CAL2JD
                    Gregorian calendar to JD
      iau DAT
                    delta(AT) = TAI-UTC
```

* * *_ iau_JD2CAL JD to Gregorian calendar

DOUBLE PRECISION FUNCTION iau_EE00 (DATE1, DATE2, EPSA, DPSI) *+ * * iau_EE00 * * The equation of the equinoxes, compatible with IAU 2000 resolutions, * given the nutation in longitude and the mean obliquity. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * EPSA d mean obliquity (Note 2) * DPSI d nutation in longitude (Note 3) * * Returned: * iau_EE00 d equation of the equinoxes (Note 4) * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 2450123.7D0 000 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) The obliquity, in radians, is mean of date. * 3) The result, which is in radians, operates in the following sense: Greenwich apparent ST = GMST + equation of the equinoxes * 4) The result is compatible with the IAU 2000 resolutions. For further details, see IERS Conventions 2003 and Capitaine et al. * (2002).Called: * iau_EECT00 equation of the equinoxes complementary terms * References: Capitaine, N., Wallace, P.T. and McCarthy, D.D., "Expressions to implement the IAU 2000 definition of UT1", Astronomy &* Astrophysics, 406, 1135-1149 (2003) * McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), * IERS Technical Note No. 32, BKG (2004) *_

DOUBLE PRECISION FUNCTION iau_EE00A (DATE1, DATE2) *+ * iau_EE00A * * _ _ _ _ _ * Equation of the equinoxes, compatible with IAU 2000 resolutions. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * * Returned: * iau EE00A d equation of the equinoxes (Note 2) * * Notes: * * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 * 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.200 (date & time method) * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. * 2) The result, which is in radians, operates in the following sense: Greenwich apparent ST = GMST + equation of the equinoxes 3) The result is compatible with the IAU 2000 resolutions. For * further details, see IERS Conventions 2003 and Capitaine et al. * (2002).* Called: * iau_PR00 IAU 2000 precession adjustments mean obliquity, IAU 1980 iau_OBL80 * iau_NUT00A nutation, IAU 2000A equation of the equinoxes, IAU 2000 * iau EE00 * References: * Capitaine, N., Wallace, P.T. and McCarthy, D.D., "Expressions to * implement the IAU 2000 definition of UT1", Astronomy & * Astrophysics, 406, 1135-1149 (2003) McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * *_

DOUBLE PRECISION FUNCTION iau_EE00B (DATE1, DATE2) *+ * * i a u _ E E O O B * * Equation of the equinoxes, compatible with IAU 2000 resolutions but * using the truncated nutation model IAU 2000B. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * iau EE00B d equation of the equinoxes (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. 2) The result, which is in radians, operates in the following sense: Greenwich apparent ST = GMST + equation of the equinoxes * 3) The result is compatible with the IAU 2000 resolutions except that accuracy has been compromised for the sake of speed. For further details, see McCarthy & Luzum (2001), IERS Conventions 2003 and Capitaine et al. (2003). * Called: * iau_PR00 IAU 2000 precession adjustments * iau_OBL80 mean obliquity, IAU 1980 iau NUT00B nutation, IAU 2000B iau_EE00 equation of the equinoxes, IAU 2000 References: Capitaine, N., Wallace, P.T. and McCarthy, D.D., "Expressions to implement the IAU 2000 definition of UT1", Astronomy &Astrophysics, 406, 1135-1149 (2003) * McCarthy, D.D. & Luzum, B.J., "An abridged model of the precession-nutation of the celestial pole", Celestial Mechanics & * Dynamical Astronomy, 85, 37-49 (2003) McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * * *_

DOUBLE PRECISION FUNCTION iau_EE06A (DATE1, DATE2) *+ * i a u _ E E O 6 A * * * Equation of the equinoxes, compatible with IAU 2000 resolutions and * IAU 2006/2000A precession-nutation. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * iau EE06A d equation of the equinoxes (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others: * DATE1 * DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) The result, which is in radians, operates in the following sense: Greenwich apparent ST = GMST + equation of the equinoxes * Called: iau_ANPM normalize angle into range +/- pi iau_GST06A Greenwich apparent sidereal time, IAU 2006/2000A * * iau_GMST06 Greenwich mean sidereal time, IAU 2006 * Reference: * McCarthy, D. D., Petit, G. (eds.), 2004, IERS Conventions (2003), IERS Technical Note No. 32, BKG * * *_

DOUBLE PRECISION FUNCTION iau_EECT00 (DATE1, DATE2) *+ * iau_EECT00 * * Equation of the equinoxes complementary terms, consistent with * IAU 2000 resolutions. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * iau EECT00 d complementary terms (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. 2) The "complementary terms" are part of the equation of the equinoxes (EE), classically the difference between apparent and mean Sidereal Time: * GAST = GMST + EE* with: EE = dpsi * cos(eps) where dpsi is the nutation in longitude and eps is the obliquity of date. However, if the rotation of the Earth were constant in an inertial frame the classical formulation would lead to apparent irregularities in the UT1 timescale traceable to side-effects of precession-nutation. In order to eliminate these effects from UT1, "complementary terms" were introduced in 1994 (IAU, 1994) and took effect from 1997 (Capitaine and Gontier, 1993): GAST = GMST + CT + EEBy convention, the complementary terms are included as part of the equation of the equinoxes rather than as part of the mean Sidereal Time. This slightly compromises the "geometrical" interpretation of mean sidereal time but is otherwise inconsequential. The present routine computes CT in the above expression, compatible with IAU 2000 resolutions (Capitaine et al., 2002, and * IERS Conventions 2003). * Called: iau_FAL03 mean anomaly of the Moon

mean anomaly of the Sun mean argument of the latitude of the Moon iau_FALP03 * iau_FAF03 mean argument of the latitude of the Sun iau_FAD03 mean elongation of the Moon's ascending no * * iau_FAOM03 mean longitude of the Moon's ascending node iau_FAVE03 mean longitude of Venus * * * mean longitude of Earth iau_FAE03 * iau_FAPA03 general accumulated precession in longitude * * References: * Capitaine, N. & Gontier, A.-M., Astron. Astrophys., 275, 645-650 (1993) * * * Capitaine, N., Wallace, P.T. and McCarthy, D.D., "Expressions to implement the IAU 2000 definition of UT1", Astronomy & Astrophysics, 406, 1135-1149 (2003) * * * * * IAU Resolution C7, Recommendation 3 (1994) * McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * * *_

```
SUBROUTINE iau_EFORM ( N, A, F, J )
*+
*
   iau_EFORM
*
*
           _ _ _ _ _
*
*
  Earth reference ellipsoids.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
   Status: canonical.
*
   Given:
*
     Ν
                i
                      ellipsoid identifier (Note 1)
*
*
   Returned:
*
                d
                       equatorial radius (meters, Note 2)
      Α
*
                      flattening (Note 2)
      F
                d
*
                      status: 0 = OK
-1 = illegal identifier (Note 3)
      ъT
                i
*
*
  Notes:
   1) The identifier N is a number that specifies the choice of
*
      reference ellipsoid. The following are supported:
             ellipsoid
*
         Ν
              WGS84
         1
              GRS80
         2
         3
              WGS72
*
      The number N has no significance outside the SOFA software.
   2) The ellipsoid parameters are returned in the form of equatorial
*
      radius in meters (A) and flattening (F). The latter is a number
      around 0.00335, i.e. around 1/298.
   3) For the case where an unsupported N value is supplied, zero A and
*
      F are returned, as well as error status.
*
*
   References:
*
      Department of Defense World Geodetic System 1984, National Imagery
*
      and Mapping Agency Technical Report 8350.2, Third Edition, p3-2.
*
      Moritz, H., Bull. Geodesique 66-2, 187 (1992).
*
*
      The Department of Defense World Geodetic System 1972, World
      Geodetic System Committee, May 1974.
*
      Explanatory Supplement to the Astronomical Almanac,
*
      P. Kenneth Seidelmann (ed), University Science Books (1992),
*
      p220.
*
*_
```

DOUBLE PRECISION FUNCTION iau_EO06A (DATE1, DATE2) *+ * i a u _ E O O 6 A * * _ _ _ _ _ * Equation of the origins, IAU 2006 precession and IAU 2000A nutation. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * iau EO06A d equation of the origins in radians * * Notes: * * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.200 (date & time method) The JD method is the most natural and convenient to use in * cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the * * optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. * 2) The equation of the origins is the distance between the true equinox and the celestial intermediate origin and, equivalently, the difference between Earth rotation angle and Greenwich * apparent sidereal time (ERA-GST). It comprises the precession (since J2000.0) in right ascension plus the equation of the equinoxes (including the small correction terms). Called: * classical NPB matrix, IAU 2006/2000A iau_PNM06A extract CIP X,Y coordinates from NPB matrix * iau_BPN2XY the CIO locator s, given X,Y, IAU 2006 iau SO6 * iau_EORS equation of the origins, given NPB matrix and s References: * Capitaine, N. & Wallace, P.T., 2006, Astron.Astrophys. 450, 855 * * Wallace, P.T. & Capitaine, N., 2006, Astron.Astrophys. 459, 981 * *_

DOUBLE PRECISION FUNCTION iau_EORS (RNPB, S) *+ * * iau_EORS * _ _ _ _ * * Equation of the origins, given the classical NPB matrix and the * quantity s. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * * Given: * RNPB d(3,3) classical nutation x precession x bias matrix * d the quantity s (the CIO locator) S * Returned: * iau_EORS d the equation of the origins in radians. * * Notes: * 1) The equation of the origins is the distance between the true equinox and the celestial intermediate origin and, equivalently, the difference between Earth rotation angle and Greenwich apparent sidereal time (ERA-GST). It comprises the precession (since J2000.0) in right ascension plus the equation of the * equinoxes (including the small correction terms). 2) The algorithm is from Wallace & Capitaine (2006). * References: * Capitaine, N. & Wallace, P.T., 2006, Astron.Astrophys. 450, 855 * * Wallace, P. & Capitaine, N., 2006, Astron.Astrophys. 459, 981 * *_

DOUBLE PRECISION FUNCTION iau_EPB (DJ1, DJ2) *+ * * iau_EPB * - - - - - - - -* * Julian Date to Besselian Epoch. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: support routine. * * Given: * DJ1,DJ2 d Julian Date (see note) * * The result is the Besselian Epoch. * * Note: * The Julian Date is supplied in two pieces, in the usual SOFA * * manner, which is designed to preserve time resolution. The * Julian Date is available as a single number by adding DJ1 and DJ2. The maximum resolution is achieved if DJ1 is 2451545D0 * * (J2000.0). * * Reference: * * Lieske, J.H., 1979, Astron.Astrophys. 73, 282. * *_

SUBROUTINE iau_EPB2JD (EPB, DJM0, DJM) *+ * * iau_EPB2JD * _ _ _ _ _ _ _ _ _ _ _ _ * * Besselian Epoch to Julian Date. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: support routine. * * Given: * Besselian Epoch (e.g. 1957.3D0) EPB d * * Returned: * MJD zero-point: always 2400000.5 DJM0 d * DJM Modified Julian Date d * * Note: * * The Julian Date is returned in two pieces, in the usual SOFA * manner, which is designed to preserve time resolution. The * Julian Date is available as a single number by adding DJMO and * DJM. * * Reference: * * Lieske, J.H., 1979, Astron.Astrophys. 73, 282. * *_

DOUBLE PRECISION FUNCTION iau_EPJ (DJ1, DJ2) *+ * * iau_EPJ * - - - - - - - -* * Julian Date to Julian Epoch. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: support routine. * * Given: * Julian Date (see note) DJ1,DJ2 d * * The result is the Julian Epoch. * * Note: * The Julian Date is supplied in two pieces, in the usual SOFA * * manner, which is designed to preserve time resolution. The * Julian Date is available as a single number by adding DJ1 and DJ2. The maximum resolution is achieved if DJ1 is 2451545D0 * * (J2000.0). * * Reference: * * Lieske, J.H., 1979, Astron.Astrophys. 73, 282. * *_

SUBROUTINE iau_EPJ2JD (EPJ, DJM0, DJM) *+ * * iau_EPJ2JD * _ _ _ _ _ _ _ _ _ _ _ * * Julian Epoch to Julian Date. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: support routine. * * Given: * d Julian Epoch (e.g. 1996.8D0) EPJ * * Returned: * MJD zero-point: always 2400000.5 DJM0 d * DJM Modified Julian Date d * * Note: * * The Julian Date is returned in two pieces, in the usual SOFA * manner, which is designed to preserve time resolution. The * Julian Date is available as a single number by adding DJMO and * DJM. * * Reference: * * Lieske, J.H., 1979, Astron.Astrophys. 73, 282. * *_

SUBROUTINE iau_EPV00 (DATE1, DATE2, PVH, PVB, JSTAT) *+ * iau_EPV00 * * * Earth position and velocity, heliocentric and barycentric, with * respect to the Barycentric Celestial Reference System. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * DATE1 d TDB date part A (Note 1) * DATE2 d TDB date part B (Note 1) * Returned: d(3,2) PVH heliocentric Earth position/velocity (AU,AU/day) PVB d(3,2) barycentric Earth position/velocity (AU,AU/day) * JSTAT status: 0 = OKi +1 = warning: date outside 1900-2100 AD * Notes: 1) The epoch EPOCH1+EPOCH2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TDB)=2450123.7 could be expressed in any of these ways, among others: EPOCH2 EPOCH1 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution * is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. However, the accuracy of the result is more likely to be * limited by the algorithm itself than the way the epoch has been * expressed. * 2) On return, the arrays PVH and PVB contain the following: * PVH(1,1) х * PVH(2,1) heliocentric position, AU У * PVH(3,1) 7. * PVH(1,2) xdot * PVH(2,2) ydot heliocentric velocity, AU/d zdot PVH(3,2) * * PVB(1,1) х PVB(2,1) barycentric position, AU У * PVB(3,1) z * PVB(1,2) xdot PVB(2,2) ydot PVB(3,2) zdot barycentric velocity, AU/d * PVB(3,2) The vectors are with respect to the Barycentric Celestial Reference System. The time unit is one day in TDB. * * 3) The routine is a SIMPLIFIED SOLUTION from the planetary theory VSOP2000 (X. Moisson, P. Bretagnon, 2001, Celes. Mechanics & Dyn. Astron., 80, 3/4, 205-213) and is an adaptation of original *

* Fortran code supplied by P. Bretagnon (private comm., 2000). * * 4) Comparisons over the time span 1900-2100 with this simplified * solution and the JPL DE405 ephemeris give the following results: * * RMS max * Heliocentric: * position error 3.7 11.2 km * 1.4 5.0 mm/s velocity error * * Barycentric: * position error 4.6 13.4 km * velocity error 1.4 4.9 mm/s * * Comparisons with the JPL DE406 ephemeris show that by 1800 and 2200 the position errors are approximately double their 1900-2100 size. By 1500 and 2500 the deterioration is a factor of 10 and by 1000 and 3000 a factor of 60. The velocity accuracy falls off at about half that rate. * * * * * *_

DOUBLE PRECISION FUNCTION iau_EQEQ94 (DATE1, DATE2) *+ * iau_EQEQ94 * * _ _ * Equation of the equinoxes, IAU 1994 model. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: canonical model. * Given: * DATE1,DATE2 d TDB date (Note 1) * * Returned: * iau_EQEQ94 d equation of the equinoxes (Note 2) * * Notes: * * 1) The date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, * JD(TDB)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 * 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. * 2) The result, which is in radians, operates in the following sense: Greenwich apparent ST = GMST + equation of the equinoxes * Called: * iau_NUT80 nutation, IAU 1980 mean obliquity, IAU 1980
normalize angle into range +/- pi iau_OBL80 * iau ANPM * * References: * IAU Resolution C7, Recommendation 3 (1994) * Capitaine, N. & Gontier, A.-M., Astron. Astrophys., 275, * 645-650 (1993) * *_

DOUBLE PRECISION FUNCTION iau_ERA00 (DJ1, DJ2) *+ * * iau_ERA00 * _ _ _ _ _ * Earth rotation angle (IAU 2000 model). * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: canonical model. * Given: * DJ1,DJ2 d UT1 as a 2-part Julian Date (see note) * * The result is the Earth rotation angle (radians), in the range 0 to * 2pi. * * Notes: * * 1) The UT1 date DJ1+DJ2 is a Julian Date, apportioned in any convenient way between the arguments DJ1 and DJ2. For example, JD(UT1)=2450123.7 could be expressed in any of these ways, among others: DJ1 DJ2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.200 (date & time method) The JD method is the most natural and convenient to use in * cases where the loss of several decimal digits of resolution is acceptable. The J2000 and MJD methods are good compromises between resolution and convenience. The date & time method is best matched to the algorithm used: maximum accuracy (or, at * least, minimum noise) is delivered when the DJ1 argument is for * Ohrs UT1 on the day in question and the DJ2 argument lies in the * range 0 to 1, or vice versa. * 2) The algorithm is adapted from Expression 22 of Capitaine et al. 2000. The time argument has been expressed in days directly, * and, to retain precision, integer contributions have been * eliminated. The same formulation is given in IERS Conventions * (2003), Chap. 5, Eq. 14. * Called: * iau_ANP normalize angle into range 0 to 2pi * References: Capitaine N., Guinot B. and McCarthy D.D, 2000, Astron. * Astrophys., 355, 398-405. * McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), * IERS Technical Note No. 32, BKG (2004) *_

```
DOUBLE PRECISION FUNCTION iau_FAD03 ( T )
*+
*
*
   iau_FAD03
*
*
*
  Fundamental argument, IERS Conventions (2003):
*
   mean elongation of the Moon from the Sun.
*
   This routine is part of the International Astronomical Union's
*
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical model.
*
*
   Given:
*
                   d
                       TDB, Julian centuries since J2000.0 (Note 1)
     т
*
*
   Returned:
*
      iau_FAD03 d D, radians (Note 2)
*
*
  Notes:
*
*
   1) Though T is strictly TDB, it is usually more convenient to use TT,
      which makes no significant difference.
   2) The expression used is as adopted in IERS Conventions (2003) and
*
      is from Simon et al. (1994).
*
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
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*
      Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683
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*_
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```
DOUBLE PRECISION FUNCTION iau_FAE03 ( T )
*+
*
    iau_FAE03
*
*
            - - - - - -
*
*
   Fundamental argument, IERS Conventions (2003):
*
   mean longitude of Earth.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical model.
*
*
   Given:
*
                   d
                        TDB, Julian centuries since J2000.0 (Note 1)
     т
*
*
   Returned:
*
      iau_FAE03 d mean longitude of Earth, radians (Note 2)
*
*
  Notes:
*
   1) Though T is strictly TDB, it is usually more convenient to use TT,
      which makes no significant difference.
   2) The expression used is as adopted in IERS Conventions (2003) and
*
      comes from Souchay et al. (1999) after Simon et al. (1994).
*
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
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*
*
      Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683
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*
      Souchay, J., Loysel, B., Kinoshita, H., Folgueira, M. 1999,
*
      Astron.Astrophys.Supp.Ser. 135, 111
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DOUBLE PRECISION FUNCTION iau_FAF03 (T) *+ * iau_FAF03 * * * * Fundamental argument, IERS Conventions (2003): * mean longitude of the Moon minus mean longitude of the ascending * node. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: canonical model. * * Given: * d TDB, Julian centuries since J2000.0 (Note 1) Т * * Returned: * iau_FAF03 d F, radians (Note 2) * * Notes: 1) Though T is strictly TDB, it is usually more convenient to use TT, * which makes no significant difference. * * 2) The expression used is as adopted in IERS Conventions (2003) and * * is from Simon et al. (1994). * References: * McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * * Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683 * * * *_

```
DOUBLE PRECISION FUNCTION iau_FAJU03 ( T )
*+
*
    iau_FAJU03
*
*
*
*
   Fundamental argument, IERS Conventions (2003):
*
   mean longitude of Jupiter.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical model.
*
*
   Given:
*
                   d
                        TDB, Julian centuries since J2000.0 (Note 1)
     т
*
*
   Returned:
*
      iau_FAJU03 d mean longitude of Jupiter, radians (Note 2)
*
*
   Notes:
*
   1) Though T is strictly TDB, it is usually more convenient to use TT,
      which makes no significant difference.
   2) The expression used is as adopted in IERS Conventions (2003) and
*
      comes from Souchay et al. (1999) after Simon et al. (1994).
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
*
      Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683
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*
      Souchay, J., Loysel, B., Kinoshita, H., Folgueira, M. 1999,
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      Astron.Astrophys.Supp.Ser. 135, 111
*
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```
DOUBLE PRECISION FUNCTION iau_FAL03 ( T )
*+
*
*
   iau_FAL03
*
*
*
  Fundamental argument, IERS Conventions (2003):
*
   mean anomaly of the Moon.
*
   This routine is part of the International Astronomical Union's
*
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical model.
*
*
   Given:
*
                   d
                       TDB, Julian centuries since J2000.0 (Note 1)
     т
*
*
   Returned:
*
      iau_FAL03 d l, radians (Note 2)
*
*
  Notes:
*
*
   1) Though T is strictly TDB, it is usually more convenient to use TT,
      which makes no significant difference.
   2) The expression used is as adopted in IERS Conventions (2003) and
*
      is from Simon et al. (1994).
*
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
*
      Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683
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*
*_
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```
DOUBLE PRECISION FUNCTION iau_FALP03 ( T )
*+
*
*
   iau_FALP03
*
*
*
  Fundamental argument, IERS Conventions (2003):
*
   mean anomaly of the Sun.
*
   This routine is part of the International Astronomical Union's
*
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical model.
*
*
   Given:
*
                        TDB, Julian centuries since J2000.0 (Note 1)
                   d
     т
*
*
   Returned:
*
      iau_FALP03 d l', radians (Note 2)
*
*
  Notes:
*
*
   1) Though T is strictly TDB, it is usually more convenient to use TT,
      which makes no significant difference.
   2) The expression used is as adopted in IERS Conventions (2003) and
*
      is from Simon et al. (1994).
*
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
*
      Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683
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*_
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```
DOUBLE PRECISION FUNCTION iau_FAMA03 ( T )
*+
*
    iau_FAMA03
*
*
*
*
   Fundamental argument, IERS Conventions (2003):
*
   mean longitude of Mars.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical model.
*
*
   Given:
*
                   d
                        TDB, Julian centuries since J2000.0 (Note 1)
     т
*
*
   Returned:
*
      iau_FAMA03 d mean longitude of Mars, radians (Note 2)
*
*
  Notes:
*
   1) Though T is strictly TDB, it is usually more convenient to use TT,
      which makes no significant difference.
   2) The expression used is as adopted in IERS Conventions (2003) and
*
      comes from Souchay et al. (1999) after Simon et al. (1994).
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
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*
      Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683
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*
      Souchay, J., Loysel, B., Kinoshita, H., Folgueira, M. 1999,
*
      Astron.Astrophys.Supp.Ser. 135, 111
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```
DOUBLE PRECISION FUNCTION iau_FAME03 ( T )
*+
*
    iau_FAME03
*
*
*
   Fundamental argument, IERS Conventions (2003):
*
*
   mean longitude of Mercury.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical model.
*
*
   Given:
*
                   d
                        TDB, Julian centuries since J2000.0 (Note 1)
     т
*
*
   Returned:
*
      iau_FAME03 d mean longitude of Mercury, radians (Note 2)
*
*
  Notes:
*
   1) Though T is strictly TDB, it is usually more convenient to use TT,
      which makes no significant difference.
   2) The expression used is as adopted in IERS Conventions (2003) and
*
      comes from Souchay et al. (1999) after Simon et al. (1994).
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
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*
      Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683
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      Souchay, J., Loysel, B., Kinoshita, H., Folgueira, M. 1999,
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      Astron.Astrophys.Supp.Ser. 135, 111
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```
DOUBLE PRECISION FUNCTION iau_FANE03 ( T )
*+
*
   iau_FANE03
*
*
            _ _ _ _ _
*
*
   Fundamental argument, IERS Conventions (2003):
*
   mean longitude of Neptune.
*
   This routine is part of the International Astronomical Union's
*
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical model.
*
*
   Given:
*
                   d
                        TDB, Julian centuries since J2000.0 (Note 1)
     т
*
*
   Returned:
*
      iau_FANE03 d mean longitude of Neptune, radians (Note 2)
*
*
  Notes:
*
   1) Though T is strictly TDB, it is usually more convenient to use TT,
      which makes no significant difference.
   2) The expression used is as adopted in IERS Conventions (2003) and
*
      is adapted from Simon et al. (1994).
*
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
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*
      Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683
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*
*_
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```
DOUBLE PRECISION FUNCTION iau_FAOM03 ( T )
*+
*
   iau_FAOM03
*
*
*
*
   Fundamental argument, IERS Conventions (2003):
*
   mean longitude of the Moon's ascending node.
*
   This routine is part of the International Astronomical Union's
*
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical model.
*
*
   Given:
*
                   d
                       TDB, Julian centuries since J2000.0 (Note 1)
     т
*
*
   Returned:
*
      iau_FAOM03 d Omega, radians (Note 2)
*
*
  Notes:
*
   1) Though T is strictly TDB, it is usually more convenient to use TT,
      which makes no significant difference.
   2) The expression used is as adopted in IERS Conventions (2003) and
*
      is from Simon et al. (1994).
*
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
*
      Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683
*
*
*_
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DOUBLE PRECISION FUNCTION iau_FAPA03 (T) *+ * iau_FAPA03 * * * Fundamental argument, IERS Conventions (2003): * * general accumulated precession in longitude. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: canonical model. * * Given: * d TDB, Julian centuries since J2000.0 (Note 1) т * * Returned: * iau_FAPA03 d general precession in longitude, radians (Note 2) * * Notes: * * 1) Though T is strictly TDB, it is usually more convenient to use TT, which makes no significant difference. 2) The expression used is as adopted in IERS Conventions (2003). It is taken from Kinoshita & Souchay (1990) and comes originally from * * Lieske et al. (1977). * References: * * Kinoshita, H. and Souchay J. 1990, Celest.Mech. and Dyn.Astron. * 48, 187 * Lieske, J.H., Lederle, T., Fricke, W. & Morando, B. 1977, Astron.Astrophys. 58, 1-16 * * * McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * * *_

```
DOUBLE PRECISION FUNCTION iau_FASA03 ( T )
*+
*
    iau_FASA03
*
*
*
*
   Fundamental argument, IERS Conventions (2003):
*
   mean longitude of Saturn.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical model.
*
*
   Given:
*
                   d
                        TDB, Julian centuries since J2000.0 (Note 1)
     т
*
*
   Returned:
*
      iau_FASA03 d mean longitude of Saturn, radians (Note 2)
*
*
  Notes:
*
   1) Though T is strictly TDB, it is usually more convenient to use TT,
      which makes no significant difference.
   2) The expression used is as adopted in IERS Conventions (2003) and
*
      comes from Souchay et al. (1999) after Simon et al. (1994).
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
*
      Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683
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*
      Souchay, J., Loysel, B., Kinoshita, H., Folgueira, M. 1999,
*
      Astron.Astrophys.Supp.Ser. 135, 111
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*_
```

```
DOUBLE PRECISION FUNCTION iau_FAUR03 ( T )
*+
*
   iau_FAUR03
*
*
*
*
  Fundamental argument, IERS Conventions (2003):
*
   mean longitude of Uranus.
*
   This routine is part of the International Astronomical Union's
*
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical model.
*
*
   Given:
*
                   d
                        TDB, Julian centuries since J2000.0 (Note 1)
     т
*
*
   Returned:
*
      iau_FAUR03 d mean longitude of Uranus, radians (Note 2)
*
*
  Notes:
*
   1) Though T is strictly TDB, it is usually more convenient to use TT,
      which makes no significant difference.
   2) The expression used is as adopted in IERS Conventions (2003) and
*
      is adapted from Simon et al. (1994).
*
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
*
      Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683
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*_
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```
DOUBLE PRECISION FUNCTION iau_FAVE03 ( T )
*+
*
   iau_FAVE03
*
*
*
*
   Fundamental argument, IERS Conventions (2003):
*
   mean longitude of Venus.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical model.
*
*
   Given:
*
                   d
                        TDB, Julian centuries since J2000.0 (Note 1)
     т
*
*
   Returned:
*
      iau_FAVE03 d mean longitude of Venus, radians (Note 2)
*
*
  Notes:
*
   1) Though T is strictly TDB, it is usually more convenient to use TT,
      which makes no significant difference.
   2) The expression used is as adopted in IERS Conventions (2003) and
*
      comes from Souchay et al. (1999) after Simon et al. (1994).
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
*
      Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683
*
*
*
      Souchay, J., Loysel, B., Kinoshita, H., Folgueira, M. 1999,
*
      Astron.Astrophys.Supp.Ser. 135, 111
*
*_
```

SUBROUTINE iau_FK52H (R5, D5, DR5, DD5, PX5, RV5, : RH, DH, DRH, DDH, PXH, RVH) *+ * * іаи_ F К 5 2 Н * * Transform FK5 (J2000.0) star data into the Hipparcos system. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given (all FK5, equinox J2000.0, epoch J2000.0): RA (radians) * R5 d D5 d Dec (radians) DR5 d proper motion in RA (dRA/dt, rad/Jyear) proper motion in Dec (dDec/dt, rad/Jyear) DD5 d PX5 d parallax (arcsec) RV5 d radial velocity (km/s, positive = receding) * Returned (all Hipparcos, epoch J2000.0): RA (radians) RH d Dec (radians) DH d DRH proper motion in RA (dRA/dt, rad/Jyear) d DDH d proper motion in Dec (dDec/dt, rad/Jyear) parallax (arcsec) * PXH d RVH d radial velocity (km/s, positive = receding) * Notes: 1) This routine transforms FK5 star positions and proper motions into * the system of the Hipparcos catalog. 2) The proper motions in RA are dRA/dt rather than cos(Dec)*dRA/dt, and are per year rather than per century. 3) The FK5 to Hipparcos transformation is modeled as a pure rotation * and spin; zonal errors in the FK5 catalog are not taken into * account. * 4) See also iau_H2FK5, iau_FK5HZ, iau_HFK5Z. * Called: * iau_STARPV star catalog data to space motion pv-vector iau_FK5HIP * FK5 to Hipparcos rotation and spin iau RXP product of r-matrix and p-vector iau_PXP vector product of two p-vectors * iau_PPP p-vector plus p-vector space motion pv-vector to star catalog data iau_PVSTAR * Reference: * F.Mignard & M.Froeschle, Astron. Astrophys. 354, 732-739 (2000). * *_

```
SUBROUTINE iau_FK5HIP ( R5H, S5H )
*+
*
   іаи_ F К 5 Н I Р
*
*
            _ _ _ _ _ _
*
*
  FK5 to Hipparcos rotation and spin.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: support routine.
*
   Returned:
*
      R5H
               d(3,3)
                        r-matrix: FK5 rotation wrt Hipparcos (Note 2)
                        r-vector: FK5 spin wrt Hipparcos (Note 3)
*
      S5H
               d(3)
*
*
   Notes:
*
*
   1) This routine models the FK5 to Hipparcos transformation as a pure rotation and spin; zonal errors in the FK5 catalogue are
*
*
      not taken into account.
   2) The r-matrix R5H operates in the sense:
*
             P_Hipparcos = R5H x P_FK5
*
*
      where P_FK5 is a p-vector in the FK5 frame, and P_Hipparcos is
*
      the equivalent Hipparcos p-vector.
*
   3) The r-vector S5H represents the time derivative of the FK5 to
      Hipparcos rotation. The units are radians per year (Julian,
*
*
      TDB).
*
*
   Called:
*
      iau_RV2M
                   r-vector to r-matrix
*
*
  Reference:
*
*
      F.Mignard & M.Froeschle, Astron. Astrophys. 354, 732-739 (2000).
*
*_
```

SUBROUTINE iau_FK5HZ (R5, D5, DATE1, DATE2, RH, DH) *+ * іаи_ F К 5 Н Z * * * Transform an FK5 (J2000.0) star position into the system of the * Hipparcos catalogue, assuming zero Hipparcos proper motion. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * R5 d FK5 RA (radians), equinox J2000.0, at date D5d FK5 Dec (radians), equinox J2000.0, at date DATE1,DATE2 d TDB date (Notes 1,2) * Returned: * RH d Hipparcos RA (radians) * DH d Hipparcos Dec (radians) * Notes: 1) This routine converts a star position from the FK5 system to the Hipparcos system, in such a way that the Hipparcos proper motion is zero. Because such a star has, in general, a non-zero proper motion in the FK5 system, the routine requires the date at which the position in the FK5 system was determined. 2) The date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TDB)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) * 2451545D0 -1421.3D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.200 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 3) The FK5 to Hipparcos transformation is modeled as a pure rotation and spin; zonal errors in the FK5 catalogue are not taken into account. 4) The position returned by this routine is in the Hipparcos reference system but at date DATE1+DATE2. * * 5) See also iau_FK52H, iau_H2FK5, iau_HFK5Z. Called: iau_S2C spherical coordinates to unit vector iau_FK5HIP FK5 to Hipparcos rotation and spin iau_SXP multiply p-vector by scalar iau_RV2M r-vector to r-matrix iau_TRXP product of transpose of r-matrix and p-vector vector product of two p-vectors iau PXP iau_C2S p-vector to spherical iau_ANP normalize angle into range 0 to 2pi Reference: F.Mignard & M.Froeschle, Astron. Astrophys. 354, 732-739 (2000).

*_____

```
SUBROUTINE iau_FW2M ( GAMB, PHIB, PSI, EPS, R )
*+
*
    iau_FW2M
*
*
           _ _ _ _
*
   Form rotation matrix given the Fukushima-Williams angles.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
   Status: support routine.
*
   Given:
*
      GAMB
                  d
                         F-W angle gamma_bar (radians)
*
      PHIB
                         F-W angle phi_bar (radians)
                 d
      PSI
                  d
                         F-W angle psi (radians)
      EPS
                  d
                         F-W angle epsilon (radians)
*
*
*
  Returned:
               d(3,3) rotation matrix
*
      R
*
*
  Notes:
   1) Naming the following points:
            e = J2000.0 ecliptic pole,
*
            p = GCRS pole,
            E = ecliptic pole of date,
      and
           P = CIP,
      the four Fukushima-Williams angles are as follows:
         GAMB = gamma = epE
         PHIB = phi = pE
         PSI = psi = pEP
         EPS = epsilon = EP
   2) The matrix representing the combined effects of frame bias,
      precession and nutation is:
         NxPxB = R_1(-EPS).R_3(-PSI).R_1(PHIB).R_3(GAMB)
   3) Three different matrices can be constructed, depending on the
      supplied angles:
*
      o To obtain the nutation x precession x frame bias matrix,
         generate the four precession angles, generate the nutation
         components and add them to the psi_bar and epsilon_A angles,
         and call the present routine.
      o To obtain the precession x frame bias matrix, generate the
         four precession angles and call the present routine.
      o To obtain the frame bias matrix, generate the four precession angles for date J2000.0 and call the present routine.
      The nutation-only and precession-only matrices can if necessary
      be obtained by combining these three appropriately.
*
   Called:
*
      iau_IR
                    initialize r-matrix to identity
      iau RZ
                   rotate around Z-axis
      iau_RX
                   rotate around X-axis
*
   Reference:
*
      Hilton, J. et al., 2006, Celest.Mech.Dyn.Astron. 94, 351
*
*
*_
```

```
SUBROUTINE iau_FW2XY ( GAMB, PHIB, PSI, EPS, X, Y )
*+
*
   iau_FW2XY
*
*
           _ _ _ _ _
*
*
  CIP X,Y given Fukushima-Williams bias-precession-nutation angles.
*
*
   This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: support routine.
*
   Given:
*
      GAMB
                 d
                        F-W angle gamma_bar (radians)
*
      PHIB
                        F-W angle phi_bar (radians)
                 d
*
      PSI
                        F-W angle psi (radians)
                 d
*
      EPS
                 d
                        F-W angle epsilon (radians)
*
*
  Returned:
*
                 d
                       CIP X,Y ("radians")
     X,Y
*
*
  Notes:
  1) Naming the following points:
*
            e = J2000.0 ecliptic pole,
*
            p = GCRS pole
*
            E = ecliptic pole of date,
*
      and
          P = CIP,
*
      the four Fukushima-Williams angles are as follows:
         GAMB = gamma = epE
*
         PHIB = phi = pE
         PSI = psi = pEP
         EPS = epsilon = EP
   2) The matrix representing the combined effects of frame bias,
*
     precession and nutation is:
*
         NxPxB = R_1(-EPSA).R_3(-PSI).R_1(PHIB).R_3(GAMB)
*
*
      X, Y are elements (3, 1) and (3, 2) of the matrix.
*
   Called:
*
      iau FW2M
                   F-W angles to r-matrix
      iau_BPN2XY extract CIP X,Y coordinates from NPB matrix
*
*
  Reference:
*
      Hilton, J. et al., 2006, Celest.Mech.Dyn.Astron. 94, 351
*
*_
```

```
SUBROUTINE iau_GC2GD ( N, XYZ, ELONG, PHI, HEIGHT, J )
*+
*
    i a u _ G C 2 G D
*
*
*
   Transform geocentric coordinates to geodetic using the specified
*
   reference ellipsoid.
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
   Status: canonical transformation.
*
*
   Given:
*
                  i
                         ellipsoid identifier (Note 1)
      Ν
*
                  d(3) geocentric vector (Note 2)
      XYZ
*
   Returned:
                         longitude (radians, east +ve)
latitude (geodetic, radians, Note 3)
*
      ELONG
                  d
*
      PHI
                  d
*
      HEIGHT
                  d
                         height above ellipsoid (geodetic, Notes 2,3)
                         status: 0 = OK
      ιT
                  i
                                  -1 = illegal identifier (Note 3)
                                  -2 = internal error (Note 3)
*
   Notes:
   1) The identifier N is a number that specifies the choice of reference ellipsoid. The following are supported:
*
              ellipsoid
          Ν
*
          1
               WGS84
*
          2
               GRS80
          3
               WGS72
      The number N has no significance outside the SOFA software.
*
   2) The geocentric vector (XYZ, given) and height (HEIGHT, returned)
*
      are in meters.
*
   3) An error status J=-1 means that the identifier N is illegal.
                                                                             An
      error status J=-2 is theoretically impossible. In all error cases, PHI and HEIGHT are both set to -1D9.
*
*
   4) The inverse transformation is performed in the routine iau_GD2GC.
*
   Called:
*
      iau_EFORM
                     Earth reference ellipsoids
*
                     geocentric to geodetic transformation, general
      iau_GC2GDE
*
*_
```

SUBROUTINE iau_GC2GDE (A, F, XYZ, ELONG, PHI, HEIGHT, J) *+ * iau_GC2GDE * * * Transform geocentric coordinates to geodetic for a reference * ellipsoid of specified form. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * d equatorial radius (Notes 2,4) А * d flattening (Note 3) d(3) geocentric vector (Note 4) F * XYZ * * Returned: * ELONG d longitude (radians, east +ve) * PHI d latitude (geodetic, radians) HEIGHT d height above ellipsoid (geodetic, Note 4) J i status: 0 = OK-1 = illegal F-2 = illegal A* Notes: 1) This routine is closely based on the GCONV2H subroutine by Toshio Fukushima (see reference). 2) The equatorial radius, A, can be in any units, but meters is * the conventional choice. 3) The flattening, F, is (for the Earth) a value around 0.00335, i.e. around 1/298. 4) The equatorial radius, A, and the geocentric vector, XYZ, * must be given in the same units, and determine the units of * the returned height, HEIGHT. * 5) If an error occurs (J<0), ELONG, PHI and HEIGHT are unchanged. * 6) The inverse transformation is performed in the routine iau_GD2GCE. 7) The transformation for a standard ellipsoid (such as WGS84) can more conveniently be performed by calling iau_GC2GD, which uses a * * numerical code (1 for WGS84) to identify the required A and F * values. * Reference: Fukushima, T., "Transformation from Cartesian to geodetic * coordinates accelerated by Halley's method", J.Geodesy (2006) * 79: 689-693 * *_

```
SUBROUTINE iau_GD2GC ( N, ELONG, PHI, HEIGHT, XYZ, J )
*+
*
    iau_GD2GC
*
*
*
   Transform geodetic coordinates to geocentric using the specified
*
  reference ellipsoid.
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
   Status: canonical transformation.
*
*
   Given:
*
                i
                       ellipsoid identifier (Note 1)
     Ν
*
      ELONG
                d
                       longitude (radians, east +ve)
      PHI
                d
                       latitude (geodetic, radians, Note 3)
*
      HEIGHT
                d
                       height above ellipsoid (geodetic, Notes 2,3)
*
  Returned:
*
      XYZ
                d(3) geocentric vector (Note 2)
      ιT
                i
                       status: 0 = OK
                                -1 = illegal identifier (Note 3)
                                -2 = illegal case (Note 3)
*
  Notes:
   1) The identifier N is a number that specifies the choice of reference ellipsoid. The following are supported:
*
             ellipsoid
         Ν
*
         1
              WGS84
*
         2
              GRS80
         3
              WGS72
      The number N has no significance outside the SOFA software.
*
   2) The height (HEIGHT, given) and the geocentric vector (XYZ,
*
      returned) are in meters.
*
   3) No validation is performed on the arguments ELONG, PHI and HEIGHT.
*
      An error status J=-1 means that the identifier N is illegal. An
      error status J=-2 protects against cases that would lead to
*
      arithmetic exceptions. In all error cases, XYZ is set to zeros.
   4) The inverse transformation is performed in the routine iau_GC2GD.
*
*
   Called:
      iau_EFORM
                   Earth reference ellipsoids
*
      iau_GD2GCE
                   geodetic to geocentric transformation, general
*
      iau_ZP
                   zero p-vector
*
*_
```

```
SUBROUTINE iau_GD2GCE ( A, F, ELONG, PHI, HEIGHT, XYZ, J )
*+
*
   iau_GD2GCE
*
*
*
  Transform geodetic coordinates to geocentric for a reference
*
  ellipsoid of specified form.
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
  Status: support routine.
*
*
   Given:
*
                d
                      equatorial radius (Notes 1,4)
     А
      ਜ
                      flattening (Notes 2,4)
                d
                      longitude (radians, east +ve)
latitude (geodetic, radians, Note 4)
*
      ELONG
                d
*
      PHI
                d
     HEIGHT
                      height above ellipsoid (geodetic, Notes 3,4)
                d
*
  Returned:
     XYZ
                d(3) geocentric vector (Note 3)
      J
                i
                      status: 0 = OK
                              -1 = illegal case (Note 4)
*
  Notes:
*
*
  1) The equatorial radius, A, can be in any units, but meters is
      the conventional choice.
   2) The flattening, F, is (for the Earth) a value around 0.00335,
      i.e. around 1/298.
   3) The equatorial radius, A, and the height, HEIGHT, must be
      given in the same units, and determine the units of the
      returned geocentric vector, XYZ.
   4) No validation is performed on individual arguments. The error
*
      status J=-1 protects against (unrealistic) cases that would lead
      to arithmetic exceptions. If an error occurs, XYZ is unchanged.
*
*
  5) The inverse transformation is performed in the routine iau_GC2GDE.
*
   6) The transformation for a standard ellipsoid (such as WGS84) can
      more conveniently be performed by calling iau_GD2GC, which uses a
      numerical code (1 for WGS84) to identify the required A and F
*
      values.
*
  References:
*
      Green, R.M., Spherical Astronomy, Cambridge University Press,
*
      (1985) Section 4.5, p96.
      Explanatory Supplement to the Astronomical Almanac,
*
*
      P. Kenneth Seidelmann (ed), University Science Books (1992),
      Section 4.22, p202.
*
*
```

DOUBLE PRECISION FUNCTION iau_GMST00 (UTA, UTB, TTA, TTB) *+ * iau_GMST00 * * Greenwich Mean Sidereal Time (model consistent with IAU 2000 * resolutions). * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * Given: * UTA, UTB d UT1 as a 2-part Julian Date (Notes 1,2) d TTA, TTB TT as a 2-part Julian Date (Notes 1,2) Returned: Greenwich mean sidereal time (radians) iau_GMST00 d * * Notes: 1) The UT1 and TT dates UTA+UTB and TTA+TTB respectively, are both Julian Dates, apportioned in any convenient way between the argument pairs. For example, JD=2450123.7 could be expressed in any of these ways, among others: Part A Part B 2450123.7D0 0D0(JD method) 2451545D0 -1421.3D0 (J2000 method) 50123.2D0 2400000.5D0 (MJD method) 2450123.5D0 (date & time method) 0.2D0 The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable (in the case of UT; the TT is not at all critical in this respect). The J2000 and MJD methods are good compromises between resolution and convenience. For UT, the date & time method is best matched to the algorithm that is used by the Earth * Rotation Angle routine, called internally: maximum accuracy (or, * at least, minimum noise) is delivered when the UTA argument is for Ohrs UT1 on the day in question and the UTB argument lies in the range 0 to 1, or vice versa. 2) Both UT1 and TT are required, UT1 to predict the Earth rotation and TT to predict the effects of precession. If UT1 is used for * both purposes, errors of order 100 microarcseconds result. * 3) This GMST is compatible with the IAU 2000 resolutions and must be used only in conjunction with other IAU 2000 compatible components such as precession-nutation and equation of the equinoxes. 4) The result is returned in the range 0 to 2pi. * 5) The algorithm is from Capitaine et al. (2003) and IERS Conventions * 2003. Called: iau_ERA00 Earth rotation angle, IAU 2000 normalize angle into range 0 to 2pi iau ANP References: Capitaine, N., Wallace, P.T. and McCarthy, D.D., "Expressions to implement the IAU 2000 definition of UT1", Astronomy & Astrophysics, 406, 1135-1149 (2003) * McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)

*_

DOUBLE PRECISION FUNCTION iau_GMST06 (UTA, UTB, TTA, TTB) *+ * іаи_ G M S T 0 б * * * Greenwich mean sidereal time (consistent with IAU 2006 precession). * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * * Given: UTA, UTB * d UT1 as a 2-part Julian Date (Notes 1,2) TT as a 2-part Julian Date (Notes 1,2) TTA, TTB * d Returned: * * iau GMST06 d Greenwich mean sidereal time (radians) Notes: * 1) The UT1 and TT dates UTA+UTB and TTA+TTB respectively, are both Julian Dates, apportioned in any convenient way between the argument pairs. For example, JD=2450123.7 could be expressed in any of these ways, among others: Part A Part B 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable (in the case of UT; the TT is not at all critical in this respect). The J2000 and MJD methods are good compromises between resolution and convenience. For UT, the date & time method is best matched to the algorithm that is used by the Earth rotation angle routine, called internally: maximum accuracy (or, at least, minimum noise) is delivered when the UTA argument is for * Ohrs UT1 on the day in question and the UTB argument lies in the range 0 to 1, or vice versa. 2) Both UT1 and TT are required, UT1 to predict the Earth rotation and TT to predict the effects of precession. If UT1 is used for both purposes, errors of order 100 microarcseconds result. 3) This GMST is compatible with the IAU 2006 precession and must not be used with other precession models. * 4) The result is returned in the range 0 to 2pi. * Called: * iau_ERA00 Earth rotation angle, IAU 2000 iau_ANP normalize angle into range 0 to 2pi * Reference: Capitaine, N., Wallace, P.T. & Chapront, J., 2005, Astron.Astrophys. 432, 355 * * * *_

DOUBLE PRECISION FUNCTION iau_GMST82 (DJ1, DJ2) *+ * iau_GMST82 * * * Universal Time to Greenwich Mean Sidereal Time (IAU 1982 model). * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * * Given: * DJ1, DJ2 d UT1 Julian Date (see note) * * Returned: * iau_GMST82 d Greenwich mean sidereal time (radians) * * Notes: * * 1) The UT1 epoch DJ1+DJ2 is a Julian Date, apportioned in any convenient way between the arguments DJ1 and DJ2. For example, JD(UT1)=2450123.7 could be expressed in any of these ways, among others: DJ1 DJ2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.200 (date & time method) The JD method is the most natural and convenient to use in * cases where the loss of several decimal digits of resolution is acceptable. The J2000 and MJD methods are good compromises between resolution and convenience. The date & time method is best matched to the algorithm used: maximum accuracy (or, at * least, minimum noise) is delivered when the DJ1 argument is for * Ohrs UT1 on the day in question and the DJ2 argument lies in the * range 0 to 1, or vice versa. * 2) The algorithm is based on the IAU 1982 expression. This is always described as giving the GMST at 0 hours UT1. In fact, it gives the * difference between the GMST and the UT, the steady 4-minutes-per-day * drawing-ahead of ST with respect to UT. When whole days are ignored, the expression happens to equal the GMST at 0 hours UT1 each day. * * 3) In this routine, the entire UT1 (the sum of the two arguments DJ1 and DJ2) is used directly as the argument for the standard formula, the constant term of which is adjusted by 12 hours to take account * of the noon phasing of Julian Date. The UT1 is then added, but * * omitting whole days to conserve accuracy. * 4) The result is returned in the range 0 to 2pi. * Called: * iau_ANP normalize angle into range 0 to 2pi References: Transactions of the International Astronomical Union, * * XVIII B, 67 (1983). * Aoki et al., Astron. Astrophys. 105, 359-361 (1982). * *_

DOUBLE PRECISION FUNCTION iau_GST00A (UTA, UTB, TTA, TTB) *+ * i a u _ G S T 0 0 A * * Greenwich Apparent Sidereal Time (consistent with IAU 2000 * resolutions). * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * Given: * UTA, UTB d UT1 as a 2-part Julian Date (Notes 1,2) d TTA, TTB TT as a 2-part Julian Date (Notes 1,2) * Returned: iau_GST00A d Greenwich apparent sidereal time (radians) * * Notes: 1) The UT1 and TT dates UTA+UTB and TTA+TTB respectively, are both Julian Dates, apportioned in any convenient way between the argument pairs. For example, JD=2450123.7 could be expressed in any of these ways, among others: Part A Part B 2450123.7D0 0D0(JD method) -1421.3D0 (J2000 method) 2451545D0 50123.2D0 2400000.5D0 (MJD method) (date & time method) 2450123.5D0 0.2D0 The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable (in the case of UT; the TT is not at all critical in this respect). The J2000 and MJD methods are good compromises between resolution and convenience. For UT, the date & time method is best matched to the algorithm that is used by the Earth * Rotation Angle routine, called internally: maximum accuracy (or, * at least, minimum noise) is delivered when the UTA argument is for Ohrs UT1 on the day in question and the UTB argument lies in the range 0 to 1, or vice versa. 2) Both UT1 and TT are required, UT1 to predict the Earth rotation and TT to predict the effects of precession-nutation. If UT1 is * used for both purposes, errors of order 100 microarcseconds * result. 3) This GAST is compatible with the IAU 2000 resolutions and must be used only in conjunction with other IAU 2000 compatible components such as precession-nutation. 4) The result is returned in the range 0 to 2pi. * 5) The algorithm is from Capitaine et al. (2003) and IERS Conventions * 2003. Called: * iau_GMST00 Greenwich mean sidereal time, IAU 2000 iau EE00A equation of the equinoxes, IAU 2000A iau_ANP normalize angle into range 0 to 2pi References: Capitaine, N., Wallace, P.T. and McCarthy, D.D., "Expressions to * implement the IAU 2000 definition of UT1", Astronomy & Astrophysics, 406, 1135-1149 (2003) McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003),

IERS Technical Note No. 32, BKG (2004)

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DOUBLE PRECISION FUNCTION iau_GST00B (UTA, UTB) *+ * i a u _ G S T 0 0 B * Greenwich Apparent Sidereal Time (consistent with IAU 2000 * resolutions but using the truncated nutation model IAU 2000B). This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * Given: * UTA, UTB d UT1 as a 2-part Julian Date (Notes 1,2) Returned: iau GST00B d Greenwich apparent sidereal time (radians) Notes: * 1) The UT1 date UTA+UTB is a Julian Date, apportioned in any convenient way between the argument pair. For example, JD=2450123.7 could be expressed in any of these ways, among others: UTA UTB 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 and MJD methods are good compromises between resolution and convenience. For UT, the date & time method is best matched to the algorithm that is used by the Earth Rotation Angle routine, called internally: maximum accuracy (or, at least, minimum noise) is delivered when the UTA argument is for Ohrs UT1 on the day in question and the UTB argument lies in the range 0 to 1, or vice versa. 2) The result is compatible with the IAU 2000 resolutions, except that accuracy has been compromised for the sake of speed and * convenience in two respects: . UT is used instead of TDB (or TT) to compute the precession * component of GMST and the equation of the equinoxes. This results in errors of order 0.1 mas at present. . The IAU 2000B abridged nutation model (McCarthy & Luzum, 2001) is used, introducing errors of up to 1 mas. 3) This GAST is compatible with the IAU 2000 resolutions and must be used only in conjunction with other IAU 2000 compatible components such as precession-nutation. 4) The result is returned in the range 0 to 2pi. 5) The algorithm is from Capitaine et al. (2003) and IERS Conventions 2003. Called: iau GMST00 Greenwich mean sidereal time, IAU 2000 equation of the equinoxes, IAU 2000B iau_EE00B iau_ANP normalize angle into range 0 to 2pi References: Capitaine, N., Wallace, P.T. and McCarthy, D.D., "Expressions to

* implement the IAU 2000 definition of UT1", Astronomy & * Astrophysics, 406, 1135-1149 (2003) * * McCarthy, D.D. & Luzum, B.J., "An abridged model of the * precession-nutation of the celestial pole", Celestial Mechanics & * Dynamical Astronomy, 85, 37-49 (2003) * * McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), * IERS Technical Note No. 32, BKG (2004) *

DOUBLE PRECISION FUNCTION iau_GST06 (UTA, UTB, TTA, TTB, RNPB) *+ * iau_GST06 * * * Greenwich apparent sidereal time, IAU 2006, given the NPB matrix. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * Given: * UTA, UTB d UT1 as a 2-part Julian Date (Notes 1,2) TT as a 2-part Julian Date (Notes 1,2) TTA, TTB d d(3,3) RNPB nutation x precession x bias matrix * Returned: * iau_GST06 d Greenwich apparent sidereal time (radians) * * Notes: * 1) The UT1 and TT dates UTA+UTB and TTA+TTB respectively, are both Julian Dates, apportioned in any convenient way between the argument pairs. For example, JD=2450123.7 could be expressed in any of these ways, among others: Part A Part B 2450123.7D0 0D0 (JD method) -1421.3D0 (J2000 method) 2451545D0 50123.2D0 2400000.5D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable (in the case of UT; the TT is not at all critical in this respect). The J2000 and MJD methods are good compromises * between resolution and convenience. For UT, the date & time * method is best matched to the algorithm that is used by the Earth rotation angle routine, called internally: maximum accuracy (or, * at least, minimum noise) is delivered when the UTA argument is for * Ohrs UT1 on the day in question and the UTB argument lies in the range 0 to 1, or vice versa. 2) Both UT1 and TT are required, UT1 to predict the Earth rotation and TT to predict the effects of precession-nutation. If UT1 is * used for both purposes, errors of order 100 microarcseconds * result. 3) Although the routine uses the IAU 2006 series for s+XY/2, it is otherwise independent of the precession-nutation model and can in practice be used with any equinox-based NPB matrix. 4) The result is returned in the range 0 to 2pi. * * Called: iau_BPN2XY * extract CIP X,Y coordinates from NPB matrix the CIO locator s, given X,Y, IAU 2006 iau_S06 normalize angle into range 0 to 2pi iau_ANP iau_ERA00 Earth rotation angle, IAU 2000 equation of the origins, given NPB matrix and s iau_EORS * Reference: * Wallace, P.T. & Capitaine, N., 2006, Astron.Astrophys. 459, 981 * *_

DOUBLE PRECISION FUNCTION iau_GST06A (UTA, UTB, TTA, TTB) *+ * i a u _ G S T 0 6 A * * * Greenwich apparent sidereal time (consistent with IAU 2000 and 2006 * resolutions). * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * Given: * UTA, UTB d UT1 as a 2-part Julian Date (Notes 1,2) TTA, TTB TT as a 2-part Julian Date (Notes 1,2) d Returned: Greenwich apparent sidereal time (radians) iau_GST06A d * * Notes: 1) The UT1 and TT dates UTA+UTB and TTA+TTB respectively, are both Julian Dates, apportioned in any convenient way between the argument pairs. For example, JD=2450123.7 could be expressed in any of these ways, among others: Part A Part B 2450123.7D0 0D0(JD method) -1421.3D0 (J2000 method) 2451545D0 50123.2D0 2400000.5D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable (in the case of UT; the TT is not at all critical in this respect). The J2000 and MJD methods are good compromises between resolution and convenience. For UT, the date & time method is best matched to the algorithm that is used by the Earth * rotation angle routine, called internally: maximum accuracy (or, at least, minimum noise) is delivered when the UTA argument is for Ohrs UT1 on the day in question and the UTB argument lies in the range 0 to 1, or vice versa. 2) Both UT1 and TT are required, UT1 to predict the Earth rotation and TT to predict the effects of precession-nutation. If UT1 is used for both purposes, errors of order 100 microarcseconds 4 result. 3) This GAST is compatible with the IAU 2000/2006 resolutions and must be used only in conjunction with IAU 2006 precession and IAU 2000A nutation. 4) The result is returned in the range 0 to 2pi. * Called: iau_PNM06A classical NPB matrix, IAU 2006/2000A Greenwich apparent ST, IAU 2006, given NPB matrix iau GST06 Reference: * Wallace, P.T. & Capitaine, N., 2006, Astron.Astrophys. 459, 981 *_

DOUBLE PRECISION FUNCTION iau_GST94 (UTA, UTB) *+ * iau_GST94 * * Greenwich Apparent Sidereal Time (consistent with IAU 1982/94 * resolutions). * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * Given: * UTA, UTB d UT1 as a 2-part Julian Date (Notes 1,2) Returned: * iau GST94 d Greenwich apparent sidereal time (radians) Notes: * 1) The UT1 date UTA+UTB is a Julian Date, apportioned in any convenient way between the argument pair. For example, JD=2450123.7 could be expressed in any of these ways, among others: UTA UTB 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 and MJD methods are good compromises between resolution and convenience. For UT, the date & time method is best matched to the algorithm that is used by the Earth Rotation Angle routine, called internally: maximum accuracy (or, at least, minimum noise) is delivered when the UTA argument is for * Ohrs UT1 on the day in question and the UTB argument lies in the range 0 to 1, or vice versa. 2) The result is compatible with the IAU 1982 and 1994 resolutions, except that accuracy has been compromised for the sake of * convenience in that UT is used instead of TDB (or TT) to compute the equation of the equinoxes. 3) This GAST must be used only in conjunction with contemporaneous IAU standards such as 1976 precession, 1980 obliquity and 1982 * nutation. It is not compatible with the IAU 2000 resolutions. 4) The result is returned in the range 0 to 2pi. Called: * iau_GMST82 Greenwich mean sidereal time, IAU 1982 iau_EQEQ94 equation of the equinoxes, IAU 1994 normalize angle into range 0 to 2pi iau_ANP References: Explanatory Supplement to the Astronomical Almanac, * P. Kenneth Seidelmann (ed), University Science Books (1992) * IAU Resolution C7, Recommendation 3 (1994) * *_

SUBROUTINE iau_H2FK5 (RH, DH, DRH, DDH, PXH, RVH, : R5, D5, DR5, DD5, PX5, RV5) *+ * * іаи_Н2FК5 * * Transform Hipparcos star data into the FK5 (J2000.0) system. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given (all Hipparcos, epoch J2000.0): * RA (radians) RH d d DH Dec (radians) DRH d proper motion in RA (dRA/dt, rad/Jyear) DDH d proper motion in Dec (dDec/dt, rad/Jyear) PXH d parallax (arcsec) RVH d radial velocity (km/s, positive = receding) Returned (all FK5, equinox J2000.0, epoch J2000.0): RA (radians) R5 d Dec (radians) D5 d DR 5 d proper motion in RA (dRA/dt, rad/Jyear) DD5 d proper motion in Dec (dDec/dt, rad/Jyear) parallax (arcsec) * PX5 d RV5 d radial velocity (km/s, positive = receding) * Notes: 1) This routine transforms Hipparcos star positions and proper * motions into FK5 J2000.0. 2) The proper motions in RA are dRA/dt rather than cos(Dec)*dRA/dt, and are per year rather than per century. 3) The FK5 to Hipparcos transformation is modeled as a pure rotation * and spin; zonal errors in the FK5 catalog are not taken into * account. 4) See also iau_FK52H, iau_FK5HZ, iau_HFK5Z. * Called: iau_STARPV star catalog data to space motion pv-vector iau_FK5HIP * FK5 to Hipparcos rotation and spin iau RV2M r-vector to r-matrix iau_RXP product of r-matrix and p-vector * iau_TRXP product of transpose of r-matrix and p-vector iau_PXP vector product of two p-vectors * iau_PMP p-vector minus p-vector space motion pv-vector to star catalog data iau PVSTAR * Reference: * F.Mignard & M.Froeschle, Astron. Astrophys. 354, 732-739 (2000). * *

SUBROUTINE iau_HFK5Z (RH, DH, DATE1, DATE2, R5, D5, DR5, DD5) *+ * iau_HFK5Z * * * Transform a Hipparcos star position into FK5 J2000.0, assuming * zero Hipparcos proper motion. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * d Hipparcos RA (radians) RH Hipparcos Dec (radians) DH d DATE1, DATE2 d TDB date (Note 1) Returned (all FK5, equinox J2000.0, date DATE1+DATE2): R5 d RA (radians) * D5 d Dec (radians) FK5 RA proper motion (rad/year, Note 4) DR 5 d DD5 d Dec proper motion (rad/year, Note 4) Notes: 1) The date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TDB)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 2450123.7D0 (JD method) 0D0 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * 2) The proper motion in RA is dRA/dt rather than cos(Dec)*dRA/dt. 3) The FK5 to Hipparcos transformation is modeled as a pure * rotation and spin; zonal errors in the FK5 catalogue are not taken into account. 4) It was the intention that Hipparcos should be a close * approximation to an inertial frame, so that distant objects have zero proper motion; such objects have (in general) non-zero proper motion in FK5, and this routine returns those * fictitious proper motions. 5) The position returned by this routine is in the FK5 J2000.0 reference system but at date DATE1+DATE2. 6) See also iau_FK52H, iau_H2FK5, iau_FK5ZHZ. Called: iau S2C spherical coordinates to unit vector iau FK5HIP FK5 to Hipparcos rotation and spin iau_RXP product of r-matrix and p-vector iau_SXP multiply p-vector by scalar * iau_RXR product of two r-matrices iau_TRXP product of transpose of r-matrix and p-vector iau_PXP vector product of two p-vectors pv-vector to spherical iau PV2S

* Reference:

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* F.Mignard & M.Froeschle, Astron. Astrophys. 354, 732-739 (2000).
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SUBROUTINE iau_IR ( R )
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*
          - - -
*
   iau_IR
*
   _ _ _ _ _ _ _ _
*
*
  Initialize an r-matrix to the identity matrix.
*

    * This routine is part of the International Astronomical Union's
    * SOFA (Standards of Fundamental Astronomy) software collection.

*
*
   Status: vector/matrix support routine.
*
* Returned:
*
                d(3,3) r-matrix
      R
*
*_
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SUBROUTINE iau_JD2CAL (DJ1, DJ2, IY, IM, ID, FD, J) *+ * iau_JD2CAL * * _ _ _ _ _ _ * Julian Date to Gregorian year, month, day, and fraction of a day. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * Given: * DJ1,DJ2 d Julian Date (Notes 1, 2) * * Returned: * ΙY i year * ΙM i month * ΤD i day fraction of day * FDd * J i status: 0 = OK -1 = unacceptable date (Note 3) * * Notes: * * 1) The earliest valid date is -68569.5 (-4900 March 1). The largest value accepted is 10⁹. 2) The Julian Date is apportioned in any convenient way between * * the arguments DJ1 and DJ2. For example, JD=2450123.7 could * be expressed in any of these ways, among others: * DJ1 DJ2 * 2450123.7D0 0D0 (JD method) -1421.3D0 50123.2D0 * 2451545D0 (J2000 method) 2400000.5D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) * 3) In early eras the conversion is from the "Proleptic Gregorian Calendar"; no account is taken of the date(s) of adoption of * * the Gregorian Calendar, nor is the AD/BC numbering convention observed. * * Reference: * Explanatory Supplement to the Astronomical Almanac, * P. Kenneth Seidelmann (ed), University Science Books (1992), * Section 12.92 (p604). * *_

SUBROUTINE iau_JDCALF (NDP, DJ1, DJ2, IYMDF, J) *+ * iau_JDCALF * * * Julian Date to Gregorian Calendar, expressed in a form convenient for formatting messages: rounded to a specified precision, and with * * the fields stored in a single array. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * * Given: * number of decimal places of days in fraction NDP i * DJ1,DJ2 d DJ1+DJ2 = Julian Date (Note 1) * * Returned: i(4) year, month, day, fraction in Gregorian * IYMDF * calendar ъT i status: -1 = date out of range 0 = OK +1 = NDP not 0-9 (interpreted as 0)* Notes: 1) The Julian Date is apportioned in any convenient way between the arguments DJ1 and DJ2. For example, JD=2450123.7 could be expressed in any of these ways, among others: DJ2 * DJ1 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) 2) In early eras the conversion is from the "Proleptic Gregorian Calendar"; no account is taken of the date(s) of adoption of * * the Gregorian Calendar, nor is the AD/BC numbering convention observed. * 3) Refer to the routine iau JD2CAL. 4) NDP should be 4 or less if internal overflows are to be * avoided on machines which use 16-bit integers. * Called: * iau_JD2CAL JD to Gregorian calendar * Reference: * Explanatory Supplement to the Astronomical Almanac, * * P. Kenneth Seidelmann (ed), University Science Books (1992), * Section 12.92 (p604). * *_

SUBROUTINE iau_NUM00A (DATE1, DATE2, RMATN) *+ * iau_NUM00A * * - - - - - -* Form the matrix of nutation for a given date, IAU 2000A model. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * * Returned: * RMATN d(3,3) nutation matrix * * Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any * convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.200 (date & time method) The JD method is the most natural and convenient to use in * cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. * 2) The matrix operates in the sense V(true) = RMATN * V(mean), where the p-vector V(true) is with respect to the true * equatorial triad of date and the p-vector V(mean) is with respect to the mean equatorial triad of date. * * 3) A faster, but slightly less accurate result (about 1 mas), can be * obtained by using instead the iau_NUM00B routine. * Called: * iau_PN00A bias/precession/nutation, IAU 2000A * Reference: Explanatory Supplement to the Astronomical Almanac, * P. Kenneth Seidelmann (ed), University Science Books (1992), * Section 3.222-3 (p114). * *_

SUBROUTINE iau_NUM00B (DATE1, DATE2, RMATN) *+ * i a u _ N U M O O B * * _ _ _ _ _ _ * Form the matrix of nutation for a given date, IAU 2000B model. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * * Returned: * RMATN d(3,3) nutation matrix * * Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any * convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.200 (date & time method) The JD method is the most natural and convenient to use in * cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. * 2) The matrix operates in the sense V(true) = RMATN * V(mean), where the p-vector V(true) is with respect to the true * equatorial triad of date and the p-vector V(mean) is with * respect to the mean equatorial triad of date. * 3) The present routine is faster, but slightly less accurate (about * 1 mas), than the iau_NUM00A routine. * Called: * iau_PN00B bias/precession/nutation, IAU 2000B * Reference: Explanatory Supplement to the Astronomical Almanac, * P. Kenneth Seidelmann (ed), University Science Books (1992), * Section 3.222-3 (p114). * *_

SUBROUTINE iau_NUM06A (DATE1, DATE2, RMATN) *+ * iau_NUM06A * * _ _ _ _ _ _ * Form the matrix of nutation for a given date, IAU 2006/2000A model. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * * Returned: * RMATN d(3,3) nutation matrix * * Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any * convenient way between the two arguments. For example, * JD(TT)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 * 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. * 2) The matrix operates in the sense V(true) = RMATN * V(mean), where the p-vector V(true) is with respect to the true * equatorial triad of date and the p-vector V(mean) is with * respect to the mean equatorial triad of date. * Called: mean obliquity, IAU 2006 nutation, IAU 2006/2000A * iau OBL06 iau NUT06A * iau_NUMAT form nutation matrix * References: * Capitaine, N., Wallace, P.T. & Chapront, J., 2005, Astron. Astrophys. 432, 355 * * * Wallace, P.T. & Capitaine, N., 2006, Astron.Astrophys. 459, 981 *_

```
SUBROUTINE iau_NUMAT ( EPSA, DPSI, DEPS, RMATN )
*+
*
    iau_NUMAT
*
*
            _ _ _ _ _ _
*
   Form the matrix of nutation.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: support routine.
*
   Given:
*
      EPSA
                     d
                              mean obliquity of date (Note 1)
*
                              nutation (Note 2)
      DPSI,DEPS
                     d
*
   Returned:
                   d(3,3) nutation matrix (Note 3)
*
      RMATN
*
*
   Notes:
*
*
   1) The supplied mean obliquity EPSA, must be consistent with the
      precession-nutation models from which DPSI and DEPS were obtained.
*
   2) The caller is responsible for providing the nutation components;
      they are in longitude and obliquity, in radians and are with
respect to the equinox and ecliptic of date.
*
   3) The matrix operates in the sense V(true) = RMATN * V(mean),
*
      where the p-vector V(true) is with respect to the true
      equatorial triad of date and the p-vector V(mean) is with
*
      respect to the mean equatorial triad of date.
   Called:
*
      iau_IR
                   initialize r-matrix to identity
                    rotate around X-axis
rotate around Z-axis
*
      iau_RX
      iau_RZ
*
*
   Reference:
*
      Explanatory Supplement to the Astronomical Almanac,
*
      P. Kenneth Seidelmann (ed), University Science Books (1992),
*
      Section 3.222-3 (p114).
*
*_
```

SUBROUTINE iau_NUT00A (DATE1, DATE2, DPSI, DEPS) *+ * iau_NUT00A * Nutation, IAU 2000A model (MHB2000 luni-solar and planetary nutation * with free core nutation omitted). This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) Returned: DPSI,DEPS d nutation, luni-solar + planetary (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) -1421.3D0 (J2000 method) 2451545D0 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. 2) The nutation components in longitude and obliquity are in radians and with respect to the equinox and ecliptic of date. The obliquity at J2000.0 is assumed to be the Lieske et al. (1977) value of 84381.448 arcsec. Both the luni-solar and planetary nutations are included. The latter are due to direct planetary nutations and the perturbations of the lunar and terrestrial orbits. 3) The routine computes the MHB2000 nutation series with the associated corrections for planetary nutations. It is an implementation of the nutation part of the IAU 2000A precessionnutation model, formally adopted by the IAU General Assembly in 2000, namely MHB2000 (Mathews et al. 2002), but with the free core nutation (FCN - see Note 4) omitted. 4) The full MHB2000 model also contains contributions to the nutations in longitude and obliquity due to the free-excitation of the free-core-nutation during the period 1979-2000. These FCN terms, which are time-dependent and unpredictable, are NOT included in the present routine and, if required, must be independently computed. With the FCN corrections included, the present routine delivers a pole which is at current epochs accurate to a few hundred microarcseconds. The omission of FCN introduces further errors of about that size. 5) The present routine provides classical nutation. The MHB2000 * algorithm, from which it is adapted, deals also with (i) the offsets between the GCRS and mean poles and (ii) the adjustments in longitude and obliquity due to the changed precession rates. These additional functions, namely frame bias and precession

* adjustments, are supported by the SOFA routines iau BI00 and iau PR00. * * 6) The MHB2000 algorithm also provides "total" nutations, comprising the arithmetic sum of the frame bias, precession adjustments, * luni-solar nutation and planetary nutation. These total nutations can be used in combination with an existing IAU 1976 precession implementation, such as iau_PMAT76, to deliver GCRS-to-true predictions of sub-mas accuracy at current epochs. However, there * are three shortcomings in the MHB2000 model that must be taken into account if more accurate or definitive results are required (see Wallace 2002): (i) The MHB2000 total nutations are simply arithmetic sums, yet in reality the various components are successive Euler rotations. This slight lack of rigor leads to cross terms that exceed 1 mas after a century. The rigorous procedure is to form the GCRS-to-true rotation matrix by applying the bias, precession and nutation in that order. (ii) Although the precession adjustments are stated to be with respect to Lieske et al. (1977), the MHB2000 model does not specify which set of Euler angles are to be used and how the adjustments are to be applied. The most literal and * straightforward procedure is to adopt the 4-rotation epsilon_0, psi_A, omega_A, xi_A option, and to add DPSIPR to psi_A and DEPSPR to both omega_A and eps_A. (iii) The MHB2000 model predates the determination by Chapront et al. (2002) of a 14.6 mas displacement between the $\tt J2000.0$ mean equinox and the origin of the ICRS frame. It should, however, be noted that neglecting this displacement when calculating star coordinates does not lead to a 14.6 mas change in right ascension, only a small second-order distortion in the pattern of the precession-nutation effect. For these reasons, the SOFA routines do not generate the "total nutations" directly, though they can of course easily be generated by calling iau_BI00, iau_PR00 and the present routine and adding the results. 7) The MHB2000 model contains 41 instances where the same frequency appears multiple times, of which 38 are duplicates and three are * triplicates. To keep the present code close to the original MHB algorithm, this small inefficiency has not been corrected. * Called: mean anomaly of the Moon mean argument of the latitude of the Moon * iau FAL03 iau_FAF03 iau_FAOM03 mean longitude of the Moon's ascending node * iau_FAME03 mean longitude of Mercury mean longitude of Venus iau FAVE03 * iau_FAE03 mean longitude of Earth mean longitude of Mars * iau FAMA03 mean longitude of Jupiter iau FAJU03 * iau_FASA03 mean longitude of Saturn iau_FAUR03 mean longitude of Uranus * iau_FAPA03 general accumulated precession in longitude References: * Chapront, J., Chapront-Touze, M. & Francou, G. 2002, * Astron.Astrophys. 387, 700 Lieske, J.H., Lederle, T., Fricke, W. & Morando, B. 1977, Astron.Astrophys. 58, 1-16 Mathews, P.M., Herring, T.A., Buffet, B.A. 2002, J.Geophys.Res. 107, B4. The MHB_2000 code itself was obtained on 9th September * 2002 from ftp//maia.usno.navy.mil/conv2000/chapter5/IAU2000A. Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., Francou, G., Laskar, J. 1994, Astron.Astrophys. 282, 663-683

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* * Souchay, J., Loysel, B., Kinoshita, H., Folgueira, M. 1999,
* Astron.Astrophys.Supp.Ser. 135, 111
* * Wallace, P.T., "Software for Implementing the IAU 2000
* Resolutions", in IERS Workshop 5.1 (2002)
*
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*_

SUBROUTINE iau_NUT00B (DATE1, DATE2, DPSI, DEPS) *+ * i a u _ N U T O O B * * - - - - - -* Nutation, IAU 2000B model. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) Returned: * DPSI,DEPS d nutation, luni-solar + planetary (Note 2) * * Notes: * * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 50123.2D0 (MJD method) 2400000.5D0 2450123.5D0 0.200 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. 2) The nutation components in longitude and obliquity are in radians * and with respect to the equinox and ecliptic of date. The obliquity at J2000.0 is assumed to be the Lieske et al. (1977) value of 84381.448 arcsec. (The errors that result from using this routine with the IAU 2006 value of 84381.406 arcsec can be neglected.) The nutation model consists only of luni-solar terms, but includes also a fixed offset which compensates for certain long-period planetary terms (Note 7). 3) This routine is an implementation of the IAU 2000B abridged nutation model formally adopted by the IAU General Assembly in 2000. The routine computes the MHB_2000_SHORT luni-solar nutation series (Luzum 2001), but without the associated corrections for the precession rate adjustments and the offset between the GCRS * and J2000.0 mean poles. 4) The full IAU 2000A (MHB2000) nutation model contains nearly 1400 terms. The IAU 2000B model (McCarthy & Luzum 2003) contains only 77 terms, plus additional simplifications, yet still delivers results of 1 mas accuracy at present epochs. This combination of * accuracy and size makes the IAU 2000B abridged nutation model suitable for most practical applications. The routine delivers a pole accurate to 1 mas from 1900 to 2100 (usually better than 1 mas, very occasionally just outside 1 mas). The full IAU 2000A model, which is implemented in the routine * iau_NUT00A (q.v.), delivers considerably greater accuracy at current epochs; however, to realize this improved accuracy, corrections for the essentially unpredictable free-core-nutation (FCN) must also be included.

5) The present routine provides classical nutation. The * MHB_2000_SHORT algorithm, from which it is adapted, deals also * with (i) the offsets between the GCRS and mean poles and (ii) the adjustments in longitude and obliquity due to the changed * precession rates. These additional functions, namely frame bias and precession adjustments, are supported by the SOFA routines iau_BI00 and iau_PR00. 6) The MHB_2000_SHORT algorithm also provides "total" nutations, comprising the arithmetic sum of the frame bias, precession adjustments, and nutation (luni-solar + planetary). These total nutations can be used in combination with an existing IAU 1976 precession implementation, such as iau PMAT76, to deliver GCRS-totrue predictions of mas accuracy at current epochs. However, for * symmetry with the iau_NUT00A routine (q.v. for the reasons), the SOFA routines do not generate the "total nutations" directly. Should they be required, they could of course easily be generated by calling iau_BI00, iau_PR00 and the present routine and adding the results. 7) The IAU 2000B model includes "planetary bias" terms that are fixed * in size but compensate for long-period nutations. The amplitudes quoted in McCarthy & Luzum (2003), namely Dpsi = -1.5835 mas and Depsilon = +1.6339 mas, are optimized for the "total nutations" method described in Note 6. The Luzum (2001) values used in this SOFA implementation, namely -0.135 mas and +0.388 mas, are optimized for the "rigorous" method, where frame bias, precession and nutation are applied separately and in that order. During the interval 1995-2050, the SOFA implementation delivers a maximum error of 1.001 mas (not including FCN). References: Lieske, J.H., Lederle, T., Fricke, W., Morando, B., "Expressions for the precession quantities based upon the IAU /1976/ system of astronomical constants", Astron.Astrophys. 58, 1-2, 1-16. (1977) Luzum, B., private communication, 2001 (Fortran code MHB_2000_SHORT) * McCarthy, D.D. & Luzum, B.J., "An abridged model of the precession-nutation of the celestial pole", Cel.Mech.Dyn.Astron. * * 85, 37-49 (2003) * Simon, J.-L., Bretagnon, P., Chapront, J., Chapront-Touze, M., * Francou, G., Laskar, J., Astron.Astrophys. 282, 663-683 (1994) * *_

SUBROUTINE iau_NUT06A (DATE1, DATE2, DPSI, DEPS) *+ * iau_NUT06A * * * IAU 2000A nutation with adjustments to match the IAU 2006 precession. * * Given: * DATE1, DATE2 d TT as a 2-part Julian Date (Note 1) * * Returned: DPST.DEPS d nutation, luni-solar + planetary (Note 2) * * Status: canonical model. * * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any * * convenient way between the two arguments. For example, * JD(TT)=2450123.7 could be expressed in any of these ways, * among others DATE1 DATE2 (JD method) 2450123.7D0 000 -1421.3D0 2451545D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in * cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. * 2) The nutation components in longitude and obliquity are in radians * and with respect to the mean equinox and ecliptic of date, * IAU 2006 precession model (Hilton et al. 2006, Capitaine et al. * 2005). * 3) The routine first computes the IAU 2000A nutation, then applies * adjustments for (i) the consequences of the change in obliquity from the IAU 1980 ecliptic to the IAU 2006 ecliptic and (ii) the * secular variation in the Earth's dynamical form factor J2. 4) The present routine provides classical nutation, complementing * the IAU 2000 frame bias and IAU 2006 precession. It delivers a * pole which is at current epochs accurate to a few tens of microarcseconds, apart from the free core nutation. * Called: iau_NUT00A nutation, IAU 2000A * * Reference: * Wallace, P.T. & Capitaine, N., 2006, Astron.Astrophys. 459, 981 * *_

SUBROUTINE iau_NUT80 (DATE1, DATE2, DPSI, DEPS) *+ * iau_NUT80 * * _ _ _ _ _ * Nutation, IAU 1980 model. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: canonical model. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * * Returned: * DPSI d nutation in longitude (radians) * nutation in obliquity (radians) DEPS d * * Notes: * 1) The DATE DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TDB)=2450123.7 could be expressed in any of these ways, * among others: * DATE1 DATE2 * 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution * is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * 2) The nutation components are with respect to the ecliptic of * date. * Called: * iau_ANPM normalize angle into range +/- pi * Reference: * * Explanatory Supplement to the Astronomical Almanac, P. Kenneth Seidelmann (ed), University Science Books (1992), * Section 3.222 (p111). * *_

SUBROUTINE iau_NUTM80 (DATE1, DATE2, RMATN) *+ * iau_NUTM80 * * _ _ _ _ _ _ * * Form the matrix of nutation for a given date, IAU 1980 model. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: support routine. * Given: * DATE1,DATE2 d TDB date (Note 1) * * Returned: * RMATN d(3,3) nutation matrix * * Notes: * 1) The date DATE1+DATE2 is a Julian Date, apportioned in any * convenient way between the two arguments. For example, * JD(TDB)=2450123.7 could be expressed in any of these ways, * among others: DATE2 * DATE1 * * 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 (date & time method) 0.2D0 * * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the * * optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. * 2) The matrix operates in the sense V(true) = RMATN * V(mean), where the p-vector V(true) is with respect to the true * equatorial triad of date and the p-vector V(mean) is with * respect to the mean equatorial triad of date. * Called: nutation, IAU 1980 mean obliquity, IAU 1980 * iau NUT80 * iau_OBL80 * form nutation matrix iau_NUMAT * *_

DOUBLE PRECISION FUNCTION iau_OBL06 (DATE1, DATE2) *+ * iau_OBL06 * * _ _ _ _ _ * * Mean obliquity of the ecliptic, IAU 2006 precession model. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: canonical model. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * * Returned: * iau_OBL06 d obliquity of the ecliptic (radians, Note 2) * * Notes: * 1) The date DATE1+DATE2 is a Julian Date, apportioned in any * convenient way between the two arguments. For example, * JD(TT)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 * * 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) * * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. * 2) The result is the angle between the ecliptic and mean equator of date DATE1+DATE2. * * Reference: * * Hilton, J. et al., 2006, Celest.Mech.Dyn.Astron. 94, 351 * *_

DOUBLE PRECISION FUNCTION iau_OBL80 (DATE1, DATE2) *+ * iau_OBL80 * * _ _ _ _ _ * * Mean obliquity of the ecliptic, IAU 1980 model. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: canonical model. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * * Returned: * iau_OBL80 d obliquity of the ecliptic (radians, Note 2) * * Notes: * * 1) The date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, * JD(TDB)=2450123.7 could be expressed in any of these ways, * among others: DATE2 DATE1 * * 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) * * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. * 2) The result is the angle between the ecliptic and mean equator of date DATE1+DATE2. * * Reference: * Explanatory Supplement to the Astronomical Almanac, P. Kenneth Seidelmann (ed), University Science Books (1992), Expression 3.222-1 (p114). * * * *_

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SUBROUTINE iau_P06E ( DATE1, DATE2,
                               EPSO, PSIA, OMA, BPA, BQA, PIA, BPIA,
EPSA, CHIA, ZA, ZETAA, THETAA, PA,
GAM, PHI, PSI )
     :
      :
      :
*+
*
    i a u _ P 0 6 E
*
*
   _ _ _ _ _ _ _ _ _
*
*
   Precession angles, IAU 2006, equinox based.
*
*
   This routine is part of the International Astronomical Union's
   SOFA (Standards of Fundamental Astronomy) software collection.
*
   Status: canonical models.
*
*
   Given:
*
      DATE1,DATE2 d
                         TT as a 2-part Julian Date (Note 1)
*
*
   Returned (see Note 2):
*
      EPS0
                      d
                            epsilon_0
*
      PSIA
                      d
                            psi_A
      OMA
                      d
                            omega A
*
      BPA
                      d
                            P_A
*
      BQA
                      d
                            ΟA
      PIA
                      d
                            pi_A
*
      BPIA
                      d
                            Pi A
*
      EPSA
                      d
                            obliquity epsilon_A
*
      CHIA
                            chi_A
                      d
      ΖA
                      d
                            z_A
*
      ZETAA
                      d
                            zeta_A
*
      THETAA
                      d
                            theta_A
*
      ΡA
                      d
                            p_A
*
                            F-W angle gamma J2000
      GAM
                      d
      PHI
                      d
                            F-W angle phi_J2000
      PSI
                      d
                            F-W angle psi_J2000
*
   Notes:
*
*
   1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any
      convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways,
*
*
      among others
*
              DATE1
                               DATE2
*
*
           2450123.7D0
                                0D0
                                             (JD method)
            2451545D0
                             -1421.3D0
                                             (J2000 method)
                             50123.2D0
           2400000.5D0
                                             (MJD method)
*
           2450123.5D0
                               0.2D0
                                             (date & time method)
*
      The JD method is the most natural and convenient to use in
      cases where the loss of several decimal digits of resolution
*
      is acceptable. The J2000 method is best matched to the way
*
      the argument is handled internally and will deliver the
      optimum resolution. The MJD method and the date & time methods
*
      are both good compromises between resolution and convenience.
   2) This routine returns the set of equinox based angles for the
*
      Capitaine et al. "PO3" precession theory, adopted by the IAU in 2006. The angles are set out in Table 1 of Hilton et al. (2006):
*
       EPS0
               epsilon_0
                             obliquity at J2000.0
*
        PSIA
               psi_A
                             luni-solar precession
*
        OMA
                omega_A
                             inclination of equator wrt J2000.0 ecliptic
                             ecliptic pole x, J2000.0 ecliptic triad
*
        BPA
                P_A
*
                             ecliptic pole -y, J2000.0 ecliptic triad
       BQA
                Q_A
*
        PIA
               pi_A
                             angle between moving and J2000.0 ecliptics
*
       BPIA
               Pi_A
                             longitude of ascending node of the ecliptic
*
       EPSA
               epsilon_A
                             obliquity of the ecliptic
               chi_A
                             planetary precession
equatorial precession: -3rd 323 Euler angle
*
        CHIA
        ZA
               zΑ
```

equatorial precession: -1st 323 Euler angle equatorial precession: 2nd 323 Euler angle * ZETAA zeta A THETAA theta_A * * PA p_A general precession * gamma_J2000 J2000.0 RA difference of ecliptic poles GAM J2000.0 codeclination of ecliptic pole phi_J2000 PHT * PSI psi_J2000 longitude difference of equator poles, J2000.0 The returned values are all radians. * 3) Hilton et al. (2006) Table 1 also contains angles that depend on models distinct from the PO3 precession theory itself, namely the * IAU 2000A frame bias and nutation. The quoted polynomials are * used in other SOFA routines: . iau_XY06 contains the polynomial parts of the X and Y series. * . iau_S06 contains the polynomial part of the s+XY/2 series. . iau_PFW06 implements the series for the Fukushima-Williams angles that are with respect to the GCRS pole (i.e. the variants that include frame bias). * 4) The IAU resolution stipulated that the choice of parameterization was left to the user, and so an IAU compliant precession implementation can be constructed using various combinations of the angles returned by the present routine. 5) The parameterization used by SOFA is the version of the Fukushima-Williams angles that refers directly to the GCRS pole. These angles may be calculated by calling the routine iau_PFW06. SOFA also supports the direct computation of the CIP GCRS X,Y by series, available by calling iau_XY06. 6) The agreement between the different parameterizations is at the 1 microarcsecond level in the present era. 7) When constructing a precession formulation that refers to the GCRS pole rather than the dynamical pole, it may (depending on the choice of angles) be necessary to introduce the frame bias explicitly. * Reference: * Hilton, J. et al., 2006, Celest.Mech.Dyn.Astron. 94, 351 * Called: * iau OBL06 mean obliquity, IAU 2006 * *_

SUBROUTINE iau_P2PV (P, PV) *+ * _ _ iau_P2PV * * _ _ _ _ _ _ _ _ _ _ * * Extend a p-vector to a pv-vector by appending a zero velocity. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * d(3) p-vector Ρ * * Returned: d(3,2) pv-vector * PV * Called: iau_CP * * iau_CP copy p-vector iau_ZP zero p-vector * * *_

SUBROUTINE iau_P2S (P, THETA, PHI, R) *+ * * iau_P2S * - - - - - - - -* * P-vector to spherical polar coordinates. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * d(3) Ρ p-vector * * Returned: * d longitude angle (radians) THETA * PHI latitude angle (radians) d * radial distance d R * * Notes: * 1) If P is null, zero THETA, PHI and R are returned. * * * 2) At either pole, zero THETA is returned. * * Called: iau_C2S p-vector to spherical modulus of p-vector * * iau_PM * *_

SUBROUTINE iau_PAP (A, B, THETA) *+ * iau_PAP * * _ _ _ _ * * Position-angle from two p-vectors. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * Given: * А d(3) direction of reference point direction of point whose PA is required * В d(3) * * Returned: position angle of B with respect to A (radians) * THETA d * * Notes: * 1) The result is the position angle, in radians, of direction B with respect to direction A. It is in the range -pi to +pi. The sense is such that if B is a small distance "north" of A the position * * angle is approximately zero, and if B is a small distance "east" of * A the position angle is approximately +pi/2. * 2) A and B need not be unit vectors. * 3) Zero is returned if the two directions are the same or if either * vector is null. * * 4) If A is at a pole, the result is ill-defined. * Called: * iau_PN decompose p-vector into modulus and direction modulus of p-vector * iau_PM * vector product of two p-vectors iau_PXP * iau_PMP p-vector minus p-vector * iau_PDP scalar product of two p-vectors *_

SUBROUTINE iau_PAS (AL, AP, BL, BP, THETA) *+ * * iau_PAS * - - - - - - - -* * Position-angle from spherical coordinates. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * AL d longitude of point A (e.g. RA) in radians * latitude of point A (e.g. Dec) in radians AP d * ΒL d longitude of point B * ΒP d latitude of point B * * Returned: * position angle of B with respect to A THETA d * * Notes: * * 1) The result is the bearing (position angle), in radians, of point B with respect to point A. It is in the range -pi to +pi. The sense is such that if B is a small distance "east" of point A, * * * the bearing is approximately +pi/2. * * 2) Zero is returned if the two points are coincident. * *_

SUBROUTINE iau_PB06 (DATE1, DATE2, BZETA, BZ, BTHETA) *+ * i a u _ P B 0 6 * This routine forms three Euler angles which implement general * precession from epoch J2000.0, using the IAU 2006 model. Frame * bias (the offset between ICRS and mean J2000.0) is included. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: 1st rotation: radians clockwise around z 3rd rotation: radians clockwise around z BZETA d B7 d * BTHETA d 2nd rotation: radians counterclockwise around y * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the arguments DATE1 and DATE2. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution * is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. 2) The traditional accumulated precession angles zeta_A, z_A, theta_A cannot be obtained in the usual way, namely through polynomial expressions, because of the frame bias. The latter means that two of the angles undergo rapid changes near this date. They are * instead the results of decomposing the precession-bias matrix * obtained by using the Fukushima-Williams method, which does not suffer from the problem. The decomposition returns values which * can be used in the conventional formulation and which include frame bias. * 3) The three angles are returned in the conventional order, which * is not the same as the order of the corresponding Euler rotations. The precession-bias matrix is $R_3(-z) \propto R_2(+theta) \propto R_3(-zeta)$. 4) Should zeta_A, z_A, theta_A angles be required that do not contain frame bias, they are available by calling the SOFA routine iau_P06E. * * Called: * iau_PMAT06 PB matrix, IAU 2006 * iau RZ rotate around Z-axis * *_

SUBROUTINE iau_PDP (A, B, ADB) *+ * - - -* iau_PDP * _ _ _ _ _ _ _ _ _ * * p-vector inner (=scalar=dot) product. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: d(3) first p-vector d(3) second p-vector * A * В * * Returned: * d A.B ADB * *_

SUBROUTINE iau_PFW06 (DATE1, DATE2, GAMB, PHIB, PSIB, EPSA) *+ * iau_PFW06 * * * Precession angles, IAU 2006 (Fukushima-Williams 4-angle formulation). * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * * Returned: * d F-W angle gamma_bar (radians) GAMB * PHIB F-W angle phi_bar (radians) d * F-W angle psi_bar (radians) PSTB d * EPSA d F-W angle epsilon_A (radians) * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others DATE1 DATE2 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) 2400000.5D0 50123.2D0 * (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution * is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) Naming the following points: e = J2000.0 ecliptic pole, p = GCRS pole, E = mean ecliptic pole of date, * and P = mean pole of date, * the four Fukushima-Williams angles are as follows: $GAMB = gamma_bar = epE$ PHIB = phi_bar = pE PSIB = psi_bar = pEP EPSA = epsilon_A = EP 3) The matrix representing the combined effects of frame bias and precession is: PxB = R_1(-EPSA).R_3(-PSIB).R_1(PHIB).R_3(GAMB) 4) The matrix representing the combined effects of frame bias, precession and nutation is simply: $NxPxB = R_1(-EPSA-dE).R_3(-PSIB-dP).R_1(PHIB).R_3(GAMB)$ where dP and dE are the nutation components with respect to the * ecliptic of date. * Reference:

```
* Hilton, J. et al., 2006, Celest.Mech.Dyn.Astron. 94, 351
* Called:
* iau_OBL06 mean obliquity, IAU 2006
*
*_
```

```
SUBROUTINE iau_PLAN94 ( DATE1, DATE2, NP, PV, J )
*+
*
   iau_PLAN94
*
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
  Status: support routine.
*
  Approximate heliocentric position and velocity of a nominated major
  planet: Mercury, Venus, EMB, Mars, Jupiter, Saturn, Uranus or
  Neptune (but not the Earth itself).
  Given:
                       TDB date part A (Note 1)
      DATE1
               d
      DATE2
               d
                       TDB date part B (Note 1)
     NP
                       planet (1=Mercury, 2=Venus, 3=EMB ... 8=Neptune)
               i
  Returned:
*
     ΡV
               d(3,2) planet pos,vel (heliocentric, J2000.0, AU, AU/d)
      ιT
               i
                       status: -1 = illegal NP (outside 1-8)
                                0 = OK
                               +1 = warning: date outside 1000-3000 AD
                               +2 = warning: solution failed to converge
*
  Notes:
   1) The date DATE1+DATE2 is in the TDB timescale and is a Julian Date,
      apportioned in any convenient way between the two arguments. For
      example, JD(TDB)=2450123.7 could be expressed in any of these
      ways, among others:
             DATE1
                           date2
          2450123.7D0
                                        (JD method)
                            0D0
                          -1421.3D0
50123.2D0
           2451545D0
                                        (J2000 method)
          2400000.500
                                        (MJD method)
          2450123.5D0
                            0.2D0
                                        (date & time method)
     The JD method is the most natural and convenient to use in
      cases where the loss of several decimal digits of resolution
      is acceptable. The J2000 method is best matched to the way
      the argument is handled internally and will deliver the
      optimum resolution. The MJD method and the date & time methods
      are both good compromises between resolution and convenience.
      The limited accuracy of the present algorithm is such that any
      of the methods is satisfactory.
*
   2) If an NP value outside the range 1-8 is supplied, an error
*
      status (J = -1) is returned and the PV vector set to zeroes.
   3) For NP=3 the result is for the Earth-Moon Barycenter. To
      obtain the heliocentric position and velocity of the Earth,
      use instead the SOFA routine iau_EPV00.
   4) On successful return, the array PV contains the following:
         PV(1,1) x
         PV(2,1) y
                            heliocentric position, AU
         PV(3,1)
                 z
         PV(1,2) xdot
        PV(2,2) ydot
PV(3,2) zdot
                           heliocentric velocity, AU/d
      The reference frame is equatorial and is with respect to the
*
      mean equator and equinox of epoch J2000.0.
   5) The algorithm is due to J.L. Simon, P. Bretagnon, J. Chapront,
*
      M. Chapront-Touze, G. Francou and J. Laskar (Bureau des
```

Longitudes, Paris, France). From comparisons with JPL ephemeris DE102, they quote the following maximum errors over the interval 1800-2050: * * * * L (arcsec) B (arcsec) R (km) * * 4 300 Mercury 1 * 5 800 Venus 1 * б EMB 1 1000 * Mars 17 1 7700 Jupiter * 71 5 76000 * 267000 Saturn 81 13 * 7 712000 Uranus 86 Neptune 11 1 253000 * * Over the interval 1000-3000, they report that the accuracy is no worse than 1.5 times that over 1800-2050. Outside 1000-3000 the * * accuracy declines. * Comparisons of the present routine with the JPL DE200 ephemeris * give the following RMS errors over the interval 1960-2025: * * position (km) velocity (m/s) * Mercury 334 0.437 * 1060 Venus 0.855 * 0.815 2010 EMB * Mars 7690 1.98 * Jupiter 71700 7.70 * 199000 19.4 Saturn Uranus 564000 16.4 * 158000 Neptune 14.4 * * Comparisons against DE200 over the interval 1800-2100 gave the * following maximum absolute differences. (The results using * DE406 were essentially the same.) * L (arcsec) B (arcsec) R (km) Rdot (m/s) * 500 * 7 0.7 Mercurv 1 * 7 1 1100 0.9 Venus * 9 EMB 1 1300 1.0 26 9000 2.5 Mars 1 * Jupiter 78 6 82000 8.2 * Saturn 87 14 263000 24.6 86 7 661000 27.4 Uranus * Neptune 11 2 248000 21.4 * 6) The present SOFA re-implementation of the original Simon et al. * Fortran code differs from the original in the following respects: * The date is supplied in two parts. * The result is returned only in equatorial Cartesian form; * the ecliptic longitude, latitude and radius vector are not returned. * The result is in the J2000.0 equatorial frame, not ecliptic. More is done in-line: there are fewer calls to other * * routines. * Different error/warning status values are used. A different Kepler's-equation-solver is used (avoiding use of COMPLEX*16). * * Polynomials in T are nested to minimize rounding errors. * * Explicit double-precision constants are used to avoid mixed-mode expressions. * There are other, cosmetic, changes to comply with SOFA

*	style conventions.		
*			
*	None of the above changes affects the result significantly.		
*			
*	7) The returned status, J, indicates the most serious condition		
*	encountered during execution of the routine. Illegal NP is		
*	considered the most serious, overriding failure to converge,		
*	which in turn takes precedence over the remote epoch warning.		
*			
*	Called:		
*	iau_ANP normalize angle into range 0 to 2pi		
*			
*	Reference: Simon, J.L, Bretagnon, P., Chapront, J.,		
*	Chapront-Touze, M., Francou, G., and Laskar, J.,		
*	Astron. Astrophys. 282, 663 (1994).		
*			
*_			

SUBROUTINE iau_PM (P, R) *+ * _ _ _ _ _ _ iau_PM _____ * * * * Modulus of p-vector. * * This routine is part of the International Astronomical Union's
 * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * d(3) p-vector Ρ * * Returned: * d modulus R * *_

SUBROUTINE iau_PMAT00 (DATE1, DATE2, RBP) *+ * iau_PMAT00 * * * Precession matrix (including frame bias) from GCRS to a specified * date, IAU 2000 model. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * RBP d(3,3) bias-precession matrix (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the arguments DATE1 and DATE2. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) The matrix operates in the sense V(date) = RBP * V(GCRS), where * the p-vector V(GCRS) is with respect to the Geocentric Celestial Reference System (IAU, 2000) and the p-vector V(date) is with * respect to the mean equatorial triad of the given date. * * Called: iau_BP00 frame bias and precession matrices, IAU 2000 * Reference: * IAU: Trans. International Astronomical Union, Vol. XXIVB; Proc. * 24th General Assembly, Manchester, UK. Resolutions B1.3, B1.6. (2000)* *_

SUBROUTINE iau_PMAT06 (DATE1, DATE2, RBP) *+ * i a u _ P M A T 0 6 * * * Precession matrix (including frame bias) from GCRS to a specified * date, IAU 2006 model. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * RBP d(3,3) bias-precession matrix (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the arguments DATE1 and DATE2. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) The matrix operates in the sense V(date) = RBP * V(GCRS), where * the p-vector V(GCRS) is with respect to the Geocentric Celestial * Reference System (IAU, 2000) and the p-vector $\ensuremath{\mathtt{V}}(\ensuremath{\mathtt{date}})$ is with respect to the mean equatorial triad of the given date. * Called: iau_PFW06 bias-precession F-W angles, IAU 2006 * iau_FW2M F-W angles to r-matrix * References: * Capitaine, N. & Wallace, P.T., 2006, Astron.Astrophys. 450, 855 * Wallace, P.T. & Capitaine, N., 2006, Astron.Astrophys. 459, 981 * *_

SUBROUTINE iau_PMAT76 (DATE1, DATE2, RMATP) *+ * i a u _ P M A T 7 6 * * * Precession matrix from J2000.0 to a specified date, IAU 1976 model. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * Given: * DATE1,DATE2 d ending date, TT (Note 1) * Returned: * RMATP d(3,3) precession matrix, J2000.0 -> DATE1+DATE2 * * Notes: * 1) The ending date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the arguments DATE1 and DATE2. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.200 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. * 2) The matrix operates in the sense V(date) = RMATP * V(J2000), where the p-vector V(J2000) is with respect to the mean * equatorial triad of epoch J2000.0 and the p-vector V(date) * is with respect to the mean equatorial triad of the given date. 3) Though the matrix method itself is rigorous, the precession angles are expressed through canonical polynomials which are valid only for a limited time span. In addition, the IAU 1976 * precession rate is known to be imperfect. The absolute accuracy of the present formulation is better than 0.1 arcsec from 1960AD to 2040AD, better than 1 arcsec from 1640AD to 2360AD, and remains below 3 arcsec for the whole of the period 500BC to 3000AD. The errors exceed 10 arcsec outside the range 1200BC to 3900AD, exceed 100 arcsec outside 4200BC to 5600AD and exceed 1000 arcsec outside 6800BC to 8200AD. * Called: iau_PREC76 * accumulated precession angles, IAU 1976 initialize r-matrix to identity iau_IR rotate around Z-axis iau_RZ rotate around Y-axis iau RY iau_CR copy r-matrix References: * Lieske, J.H., 1979, Astron.Astrophys. 73, 282. equations (6) & (7), p283. * * * Kaplan, G.H., 1981, USNO circular no. 163, pA2. *_

SUBROUTINE iau_PMP (A, B, AMB) *+ * _ _ _ * iau_PMP * _ _ _ _ _ _ _ _ _ * * P-vector subtraction. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: d(3) first p-vector d(3) second p-vector * A * В * * Returned: * d(3) A - B AMB * *_

```
SUBROUTINE iau_PN ( P, R, U )
*+
*
         _ _ -
*
   iau_PN
*
  _ _ _ _ _ _ _
*
*
  Convert a p-vector into modulus and unit vector.
*
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
  Status: vector/matrix support routine.
*
*
  Given:
*
                d(3)
     Ρ
                         p-vector
*
*
   Returned:
                d
               d modulus
d(3) unit vector
*
    R
*
      U
*
*
  Note:
     If P is null, the result is null. Otherwise the result is a unit vector.
*
*
*
*
  Called:
     iau_PM modulus of p-vector
iau_ZP zero p-vector
iau_SXP multiply p-vector by scalar
*
*
*
*
*_
```

```
SUBROUTINE iau_PN00 ( DATE1, DATE2, DPSI, DEPS,
     :
                             EPSA, RB, RP, RBP, RN, RBPN )
*+
*
*
   iau_PN00
*
  Precession-nutation, IAU 2000 model: a multi-purpose routine,
*
*
   supporting classical (equinox-based) use directly and CIO-based
   use indirectly.
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
   Status: support routine.
*
*
   Given:
*
      DATE1,DATE2
                     d
                             TT as a 2-part Julian Date (Note 1)
*
      DPSI,DEPS
                    d
                             nutation (Note 2)
*
   Returned:
*
     EPSA
                     d
                             mean obliquity (Note 3)
                   d(3,3)
      RB
                             frame bias matrix (Note 4)
                             precession matrix (Note 5)
      RP
                   d(3,3)
      RBP
                   d(3,3)
                             bias-precession matrix (Note 6)
      RN
                   d(3,3)
d(3,3)
                             nutation matrix (Note 7)
      RBPN
                             GCRS-to-true matrix (Note 8)
   Notes:
   1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any
*
      convenient way between the two arguments. For example,
      JD(TT)=2450123.7 could be expressed in any of these ways,
*
      among others:
             DATE1
                             DATE2
          2450123.7D0
                              0D0
                                          (JD method)
                           -1421.3D0
           2451545D0
                                          (J2000 method)
*
          2400000.5D0
                           50123.2D0
                                          (MJD method)
*
          2450123.5D0
                              0.2D0
                                          (date & time method)
      The JD method is the most natural and convenient to use in
      cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way % f(x) = 0
*
      the argument is handled internally and will deliver the
      optimum resolution. The MJD method and the date & time methods
*
      are both good compromises between resolution and convenience.
   2) The caller is responsible for providing the nutation components; they are in longitude and obliquity, in radians and are with
*
*
      respect to the equinox and ecliptic of date. For high-accuracy
      applications, free core nutation should be included as well as
*
      any other relevant corrections to the position of the CIP.
*
   3) The returned mean obliquity is consistent with the IAU 2000
      precession-nutation models.
   4) The matrix RB transforms vectors from GCRS to J2000.0 mean equator
      and equinox by applying frame bias.
   5) The matrix RP transforms vectors from J2000.0 mean equator and
*
*
      equinox to mean equator and equinox of date by applying
      precession.
   6) The matrix RBP transforms vectors from GCRS to mean equator and
      equinox of date by applying frame bias then precession. It is the
      product RP x RB.
*
*
   7) The matrix RN transforms vectors from mean equator and equinox of
*
      date to true equator and equinox of date by applying the nutation
      (luni-solar + planetary).
```

* * * *	8) The matrix RBPN transforms vectors from GCRS to true equator and equinox of date. It is the product RN x RBP, applying frame bias, precession and nutation in that order.		
* * * * *	Called: iau_PR00 IAU 2000 precession adjustments iau_OBL80 mean obliquity, IAU 1980 iau_BP00 frame bias and precession matrices, IAU 2000 iau_NUMAT form nutation matrix iau_RXR product of two r-matrices		
*	Reference:		
* * * *	Capitaine, N., Chapront, J., Lambert, S. and Wallace, P., "Expressions for the Celestial Intermediate Pole and Celestial Ephemeris Origin consistent with the IAU 2000A precession-nutation model", Astron.Astrophys. 400, 1145-1154 (2003)		
* * * *_	n.b. The celestial ephemeris origin (CEO) was renamed "celestial intermediate origin" (CIO) by IAU 2006 Resolution 2.		

SUBROUTINE iau_PN00A (DATE1, DATE2, : DPSI, DEPS, EPSA, RB, RP, RBP, RN, RBPN) *+ * * iau_PN00A * Precession-nutation, IAU 2000A model: a multi-purpose routine, * * supporting classical (equinox-based) use directly and CIO-based use indirectly. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * * Given: DATE1,DATE2 d * TT as a 2-part Julian Date (Note 1) * * Returned: DPSI,DEPS d nutation (Note 2) EPSA * d mean obliquity (Note 3) RB d(3,3) frame bias matrix (Note 4) precession matrix (Note 5) RP d(3,3) bias-precession matrix (Note 6) RBP d(3,3) RN d(3,3) d(3,3) nutation matrix (Note 7) RBPN GCRS-to-true matrix (Notes 8,9) Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any * convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way % f(x) = 0the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * 2) The nutation components (luni-solar + planetary, IAU 2000A) in longitude and obliquity are in radians and with respect to the * equinox and ecliptic of date. Free core nutation is omitted; for the utmost accuracy, use the iau_PN00 routine, where the nutation components are caller-specified. For faster but slightly less * accurate results, use the iau_PN00B routine. 3) The mean obliquity is consistent with the IAU 2000 precession. 4) The matrix RB transforms vectors from GCRS to J2000.0 mean equator and equinox by applying frame bias. 5) The matrix RP transforms vectors from J2000.0 mean equator and * equinox to mean equator and equinox of date by applying precession. 6) The matrix RBP transforms vectors from GCRS to mean equator and equinox of date by applying frame bias then precession. It is the product RP x RB. * * 7) The matrix RN transforms vectors from mean equator and equinox of * date to true equator and equinox of date by applying the nutation (luni-solar + planetary).

	8) The matrix RBPN transforms vectors from GCRS to true equator and equinox of date. It is the product RN x RBP, applying frame bias, precession and nutation in that order.		
:	9) The X,Y,Z coordinates of the IAU 2000A Celestial Intermediate Pole are elements (3,1-3) of the matrix RBPN.		
•	carred.		
:	Reference:		
-	Capitaine, N., Chapront, J., Lambert, S. and Wallace, P., "Expressions for the Celestial Intermediate Pole and Celestial Ephemeris Origin consistent with the IAU 2000A precession-nutation model", Astron.Astrophys. 400, 1145-1154 (2003).		
	n.b. The celestial ephemeris origin (CEO) was renamed "celestial intermediate origin" (CIO) by IAU 2006 Resolution 2.		

SUBROUTINE iau_PN00B (DATE1, DATE2, : DPSI, DEPS, EPSA, RB, RP, RBP, RN, RBPN) *+ * * iau_PN00B * Precession-nutation, IAU 2000B model: a multi-purpose routine, * * supporting classical (equinox-based) use directly and CIO-based use indirectly. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * * Given: DATE1,DATE2 d * TT as a 2-part Julian Date (Note 1) * * Returned: DPSI,DEPS d nutation (Note 2) EPSA * d mean obliquity (Note 3) RB d(3,3) frame bias matrix (Note 4) precession matrix (Note 5) RP d(3,3) bias-precession matrix (Note 6) RBP d(3,3) RN d(3,3) d(3,3) nutation matrix (Note 7) RBPN GCRS-to-true matrix (Notes 8,9) Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any * convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way % f(x) = 0the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * 2) The nutation components (luni-solar + planetary, IAU 2000B) in longitude and obliquity are in radians and with respect to the * equinox and ecliptic of date. For more accurate results, but at the cost of increased computation, use the iau_PN00A routine. For the utmost accuracy, use the iau_PN00 routine, where the nutation components are caller-specified. 3) The mean obliquity is consistent with the IAU 2000 precession. 4) The matrix RB transforms vectors from GCRS to J2000.0 mean equator and equinox by applying frame bias. 5) The matrix RP transforms vectors from J2000.0 mean equator and equinox to mean equator and equinox of date by applying precession. 6) The matrix RBP transforms vectors from GCRS to mean equator and equinox of date by applying frame bias then precession. It is the product RP x RB. * * 7) The matrix RN transforms vectors from mean equator and equinox of date to true equator and equinox of date by applying the nutation (luni-solar + planetary).

equinox of dat	PN transforms vectors from GCRS to true equator and te. It is the product RN x RBP, applying frame bias, d nutation in that order.	
	rdinates of the IAU 2000B Celestial Intermediate Pole (3,1-3) of the matrix RBPN.	
Iau_NOIOOD I	nutation, IAU 2000B pias/precession/nutation results, IAU 2000	
Reference:		
Capitaine, N., Chapront, J., Lambert, S. and Wallace, P., "Expressions for the Celestial Intermediate Pole and Celestial Ephemeris Origin consistent with the IAU 2000A precession-nutation model", Astron.Astrophys. 400, 1145-1154 (2003).		
	stial ephemeris origin (CEO) was renamed "celestial iate origin" (CIO) by IAU 2006 Resolution 2.	

SUBROUTINE iau_PN06 (DATE1, DATE2, DPSI, DEPS, : EPSA, RB, RP, RBP, RN, RBPN) *+ * * іаи_РМОб * Precession-nutation, IAU 2006 model: a multi-purpose routine, * * supporting classical (equinox-based) use directly and CIO-based use indirectly. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * DPSI,DEPS d nutation (Note 2) Returned: * EPSA d mean obliquity (Note 3) d(3,3) RB frame bias matrix (Note 4) precession matrix (Note 5) RP d(3,3) RBP d(3,3) bias-precession matrix (Note 6) d(3,3) d(3,3) RN nutation matrix (Note 7) RBPN GCRS-to-true matrix (Note 8) Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any * convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way % f(x) = 0* the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) The caller is responsible for providing the nutation components; they are in longitude and obliquity, in radians and are with * * respect to the equinox and ecliptic of date. For high-accuracy applications, free core nutation should be included as well as * any other relevant corrections to the position of the CIP. 3) The returned mean obliquity is consistent with the IAU 2006 * precession. 4) The matrix RB transforms vectors from GCRS to mean J2000.0 by applying frame bias. 5) The matrix RP transforms vectors from mean J2000.0 to mean of date by applying precession. 6) The matrix RBP transforms vectors from GCRS to mean of date by applying frame bias then precession. It is the product RP x RB. 7) The matrix RN transforms vectors from mean of date to true of date * by applying the nutation (luni-solar + planetary). * 8) The matrix RBPN transforms vectors from GCRS to true of date (CIP/equinox). It is the product RN x RBP, applying frame bias,

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*
      precession and nutation in that order.
*
*
  9) The X,Y,Z coordinates of the IAU 2006/2000A Celestial Intermediate
*
      Pole are elements (3,1-3) of the matrix RBPN.
*
*
  Called:
                  bias-precession F-W angles, IAU 2006
F-W angles to r-matrix
transpose r-matrix
      iau_PFW06
*
*
      iau_FW2M
*
      iau_TR
*
      iau_RXR
                    product of two r-matrices
*
*
  References:
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*
      Capitaine, N. & Wallace, P.T., 2006, Astron.Astrophys. 450, 855
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*
      Wallace, P.T. & Capitaine, N., 2006, Astron.Astrophys. 459, 981
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SUBROUTINE iau_PN06A (DATE1, DATE2, : DPSI, DEPS, EPSA, RB, RP, RBP, RN, RBPN) *+ * * іаи_ Р № 0 6 А * Precession-nutation, IAU 2006/2000A models: a multi-purpose routine, * * supporting classical (equinox-based) use directly and CIO-based use indirectly. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * * Given: DATE1,DATE2 d * TT as a 2-part Julian Date (Note 1) * * Returned: DPSI,DEPS d nutation (Note 2) * EPSA d mean obliquity (Note 3) RB d(3,3) frame bias matrix (Note 4) precession matrix (Note 5) RP d(3,3) bias-precession matrix (Note 6) RBP d(3,3) RN d(3,3) d(3,3) nutation matrix (Note 7) RBPN GCRS-to-true matrix (Notes 8,9) Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any * convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) -1421.3D0 2451545D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way % f(x) = 0* the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * 2) The nutation components (luni-solar + planetary, IAU 2000A) in longitude and obliquity are in radians and with respect to the * equinox and ecliptic of date. Free core nutation is omitted; for the utmost accuracy, use the iau_PN06 routine, where the nutation components are caller-specified. 3) The mean obliquity is consistent with the IAU 2006 precession. * * 4) The matrix RB transforms vectors from GCRS to mean J2000.0 by * applying frame bias. 5) The matrix RP transforms vectors from mean J2000.0 to mean of date * by applying precession. 6) The matrix RBP transforms vectors from GCRS to mean of date by applying frame bias then precession. It is the product RP x RB. 7) The matrix RN transforms vectors from mean of date to true of date by applying the nutation (luni-solar + planetary). * * * 8) The matrix RBPN transforms vectors from GCRS to true of date * (CIP/equinox). It is the product RN x RBP, applying frame bias, precession and nutation in that order.

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*
   9) The X,Y,Z coordinates of the IAU 2006/2000A Celestial Intermediate Pole are elements (3,1-3) of the matrix RBPN.
*
*
*
   Called:
*
       iau_NUT06A nutation, IAU 2006/2000A
iau_PN06 bias/precession/nutation results, IAU 2006
*
*
*
*
   Reference:
*
*
        Capitaine, N. & Wallace, P.T., 2006, Astron.Astrophys. 450, 855
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SUBROUTINE iau_PNM00A (DATE1, DATE2, RBPN) *+ * i a u _ P N M O O A * * * Form the matrix of precession-nutation for a given date (including * frame bias), equinox-based, IAU 2000A model. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * RBPN d(3,3) classical NPB matrix (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) The matrix operates in the sense V(date) = RBPN * V(GCRS), where * the p-vector V(date) is with respect to the true equatorial triad of date DATE1+DATE2 and the p-vector V(GCRS) is with respect to the Geocentric Celestial Reference System (IAU, 2000). * 3) A faster, but slightly less accurate result (about 1 mas), can be obtained by using instead the iau_PNM00B routine. * Called: iau_PN00A bias/precession/nutation, IAU 2000A * Reference: IAU: Trans. International Astronomical Union, Vol. XXIVB; Proc. * 24th General Assembly, Manchester, UK. Resolutions B1.3, B1.6. * (2000)* *

SUBROUTINE iau_PNM00B (DATE1, DATE2, RBPN) *+ * i a u _ P N M O O B * * * Form the matrix of precession-nutation for a given date (including * frame bias), equinox-based, IAU 2000B model. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * RBPN d(3,3) bias-precession-nutation matrix (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) The matrix operates in the sense V(date) = RBPN * V(GCRS), where * the p-vector V(date) is with respect to the true equatorial triad of date DATE1+DATE2 and the p-vector V(GCRS) is with respect to the Geocentric Celestial Reference System (IAU, 2000). * 3) The present routine is faster, but slightly less accurate (about 1 mas), than the iau_PNM00A routine. * Called: iau_PN00B bias/precession/nutation, IAU 2000B * Reference: IAU: Trans. International Astronomical Union, Vol. XXIVB; Proc. * 24th General Assembly, Manchester, UK. Resolutions B1.3, B1.6. * (2000)* *

SUBROUTINE iau_PNM06A (DATE1, DATE2, RNPB) *+ * i a u _ P N M O 6 A * * * Form the matrix of precession-nutation for a given date (including * frame bias), IAU 2006 precession and IAU 2000A nutation models. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * RNPB d(3,3) bias-precession-nutation matrix (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: * DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution * is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) The matrix operates in the sense V(date) = RNPB * V(GCRS), where * the p-vector V(date) is with respect to the true equatorial triad * of date DATE1+DATE2 and the p-vector $\mathtt{V}(\mathtt{GCRS})$ is with respect to the Geocentric Celestial Reference System (IAU, 2000). * * Called: iau_PFW06 bias-precession F-W angles, IAU 2006 * iau_NUT06A nutation, IAU 2006/2000A * iau_FW2M F-W angles to r-matrix * Reference: Capitaine, N. & Wallace, P.T., 2006, Astron.Astrophys. 450, 855. * *_

SUBROUTINE iau_PNM80 (DATE1, DATE2, RMATPN) *+ * iau_PNM80 * * * Form the matrix of precession/nutation for a given date, IAU 1976 * precession model, IAU 1980 nutation model. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * DATE1,DATE2 d TDB date (Note 1) * Returned: * RMATPN d(3,3) combined precession/nutation matrix Notes: * 1) The date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TDB)=2450123.7 could be expressed in any of these ways, among others: * DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) The matrix operates in the sense V(date) = RMATPN * V(J2000), where the p-vector V(date) is with respect to the true equatorial triad of date DATE1+DATE2 and the p-vector * V(J2000) is with respect to the mean equatorial triad of * epoch J2000.0. Called: * precession matrix, IAU 1976 iau_PMAT76 nutation matrix, IAU 1980 iau_NUTM80 * product of two r-matrices iau_RXR * Reference: * Explanatory Supplement to the Astronomical Almanac, * P. Kenneth Seidelmann (ed), University Science Books (1992), Section 3.3 (p145). * *

SUBROUTINE iau_POM00 (XP, YP, SP, RPOM) *+ * iаи_РОМОО * * _ _ _ _ _ _ * Form the matrix of polar motion for a given date, IAU 2000. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * Given: * XP,YP d coordinates of the pole (radians, Note 1) * the TIO locator s' (radians, Note 2) SP d * Returned: * RPOM d(3,3) polar-motion matrix (Note 3) * * Notes: * 1) XP and YP are the coordinates (in radians) of the Celestial Intermediate Pole with respect to the International Terrestrial Reference System (see IERS Conventions 2003), measured along the meridians to 0 and 90 deg west respectively. 2) SP is the TIO locator $s^{\,\prime}\,,$ in radians, which positions the Terrestrial Intermediate Origin on the equator. It is obtained from polar motion observations by numerical integration, and so is in essence unpredictable. However, it is dominated by a secular drift of about 47 microarcseconds per century, and so can be taken into account by using s' = -47*t, where t is centuries since J2000.0. The routine iau SP00 implements this approximation. 3) The matrix operates in the sense V(TRS) = RPOM * V(CIP), meaning * that it is the final rotation when computing the pointing * direction to a celestial source. * Called: * iau_IR initialize r-matrix to identity rotate around Z-axis iau RZ * iau_RY rotate around Y-axis * iau_RX rotate around X-axis * Reference: McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), * IERS Technical Note No. 32, BKG (2004) * *_

SUBROUTINE iau_PPP (A, B, APB) *+ * - - - - -_ _ * iau_PPP * _ _ _ _ _ _ _ _ * * P-vector addition. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: d(3) first p-vector d(3) second p-vector * A * В * * Returned: * d(3) A + B APB * *_

SUBROUTINE iau_PPSP (A, S, B, APSB) *+ * - iau_PPSP * * _ _ _ _ _ _ _ _ _ _ * * P-vector plus scaled p-vector. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: d(3) first p-vector d scalar (multiplier for B) d(3) second p-vector * А * S * В * * Returned: * d(3) A + S*B APSB * *_

SUBROUTINE iau_PR00 (DATE1, DATE2, DPSIPR, DEPSPR) *+ * iau_PR00 * * Precession-rate part of the IAU 2000 precession-nutation models * (part of MHB2000). * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) Returned: DPSIPR, DEPSPR d precession corrections (Notes 2,3) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. 2) The precession adjustments are expressed as "nutation components" corrections in longitude and obliquity with respect to the J2000.0 equinox and ecliptic. 3) Although the precession adjustments are stated to be with respect to Lieske et al. (1977), the MHB2000 model does not specify which set of Euler angles are to be used and how the adjustments are to be applied. The most literal and straightforward procedure is to adopt the 4-rotation epsilon_0, psi_A, omega_A, xi_A option, and to add DPSIPR to psi_A and DEPSPR to both omega_A and eps_A. 4) This is an implementation of one aspect of the IAU 2000A nutation model, formally adopted by the IAU General Assembly in 2000, namely MHB2000 (Mathews et al. 2002). References: Lieske, J.H., Lederle, T., Fricke, W. & Morando, B., "Expressions for the precession quantities based upon the IAU (1976) System of Astronomical Constants", Astron.Astrophys., 58, 1-16 (1977) Mathews, P.M., Herring, T.A., Buffet, B.A., "Modeling of nutation and precession New nutation series for nonrigid Earth and insights into the Earth's interior", J.Geophys.Res., 107, B4, 2002. The MHB2000 code itself was obtained on 9th September 2002 from ftp://maia.usno.navy.mil/conv2000/chapter5/IAU2000A. * Wallace, P.T., "Software for Implementing the IAU 2000 * Resolutions", in IERS Workshop 5.1 (2002).

*_

SUBROUTINE iau_PREC76 (EP01, EP02, EP11, EP12, ZETA, Z, THETA) *+ * iau_PREC76 * _ _ _ _ _ _ * IAU 1976 precession model. * * This routine forms the three Euler angles which implement general precession between two epochs, using the IAU 1976 model (as for * * the FK5 catalog). This routine is part of the International Astronomical Union's SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * Given: * EP01,EP02 d TDB starting epoch (Note 1) TDB ending epoch (Note 1) EP11,EP12 d * * Returned: 1st rotation: radians clockwise around \boldsymbol{z} ZETA d 3rd rotation: radians clockwise around z 7. d 2nd rotation: radians counterclockwise around y * THETA d Notes: 1) The epochs EP01+EP02 and EP11+EP12 are Julian Dates, apportioned in any convenient way between the arguments EPn1 and EPn2. For example, JD(TDB)=2450123.7 could be expressed in any of these ways, among others: EPn2 EPn1 2450123.7D0 0D0(JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. The two epochs may be expressed using different methods, but at the risk of losing some resolution. * 2) The accumulated precession angles zeta, z, theta are expressed through canonical polynomials which are valid only for a limited * time span. In addition, the IAU 1976 precession rate is known to be imperfect. The absolute accuracy of the present formulation is better than 0.1 arcsec from 1960AD to 2040AD, better than 1 arcsec from 1640AD to 2360AD, and remains below 3 arcsec for the whole of the period 500BC to 3000AD. The errors exceed 10 arcsec outside the range 1200BC to 3900AD, exceed 100 arcsec outside 4200BC to 5600AD and exceed 1000 arcsec outside 6800BC to 8200AD. 3) The three angles are returned in the conventional order, which is not the same as the order of the corresponding Euler rotations. The precession matrix is $R_3(-z) \propto R_2(+theta) \propto R_3(-zeta)$. Reference: * Lieske, J.H., 1979, Astron.Astrophys. 73, 282. * equations (6) & (7), p283. * *_

SUBROUTINE iau_PV2P (PV, P) *+ * _ _ _ * iau_PV2P * _ _ _ _ _ _ _ _ _ _ * Discard velocity component of a pv-vector. * * * This routine is part of the International Astronomical Union's
 * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * d(3,2) pv-vector PV * * Returned: d(3) p-vector * Ρ * * Called: * iau_CP copy p-vector * *_

SUBROUTINE iau_PV2S (PV, THETA, PHI, R, TD, PD, RD) *+ * iau_PV2S * * _ _ _ _ _ * Convert position/velocity from Cartesian to spherical coordinates. * * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * Given: * ΡV d(3,2) pv-vector * * Returned: * THETA d longitude angle (radians) latitude angle (radians) * PHI d * d radial distance rate of change of THETA R * TD d * PD d rate of change of PHI rate of change of R * RD d * * Notes: * 1) If the position part of PV is null, THETA, PHI, TD and PD * * are indeterminate. This is handled by extrapolating the * position through unit time by using the velocity part of * PV. This moves the origin without changing the direction * of the velocity component. If the position and velocity * components of PV are both null, zeroes are returned for all * six results. * 2) If the position is a pole, THETA, TD and PD are indeterminate. In such cases zeroes are returned for all three. * * * *_

```
SUBROUTINE iau_PVDPV ( A, B, ADB )
*+
*
   iau_PVDPV
*
*
   _ _ _ _ _ _ _ _ _ _ _
*
*
   Inner (=scalar=dot) product of two pv-vectors.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: vector/matrix support routine.
*
*
   Given:
*
                 d(3,2)
      А
                               first pv-vector
*
                 d(3,2)
                               second pv-vector
      В
*
*
   Returned:
*
                d(2)
                              A . B (see note)
      ADB
*
*
   Note:
*
*
       If the position and velocity components of the two pv-vectors are
      ( Ap, Av ) and ( Bp, Bv ), the result, A . B, is the pair of numbers ( Ap . Bp , Ap . Bv + Av . Bp ). The two numbers are the dot-product of the two p-vectors and its derivative.
*
*
*
*
*
  Called:
*
      iau_PDP
                    scalar product of two p-vectors
*
*_
```

SUBROUTINE iau_PVM (PV, R, S) *+ * _ _ _ iau_PVM * * _ _ _ _ _ _ _ _ * * Modulus of pv-vector. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * d(3,2) pv-vector PV * * Returned: * modulus of position component modulus of velocity component d R * S d * * Called: * modulus of p-vector iau_PM * *_

```
SUBROUTINE iau_PVMPV ( A, B, AMB )
*+
*
  iau_PVMPV
*
*
  _ _ _ _ _ _ _ _ _ _ _
*
*
  Subtract one pv-vector from another.
*
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
  Status: vector/matrix support routine.
*
*
  Given:
*
              d(3,2)
                          first pv-vector
   A
             d(3,2) first pv-vector
d(3,2) second pv-vector
*
     В
*
*
  Returned:
*
             d(3,2) A - B
     AMB
*
* Called:
*
    iau_PMP
                 p-vector minus p-vector
*
*_
```

SUBROUTINE iau_PVPPV (A, B, APB) *+ * _ _ iau_PVPPV * * _ _ _ _ _ _ _ _ _ _ _ * * Add one pv-vector to another. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * A d(3,2) first pv-vector d(3,2) first pv-vector d(3,2) second pv-vector * В * * Returned: * d(3,2) A + B APB * * Called: * iau_PPP p-vector plus p-vector * *_

SUBROUTINE iau_PVSTAR (PV, RA, DEC, PMR, PMD, PX, RV, J) *+ * i a u _ P V S T A R * * _ _ _ _ _ * Convert star position+velocity vector to catalog coordinates. * * This routine is part of the International Astronomical Union's SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * Given (Note 1): * ΡV d(3,2) pv-vector (AU, AU/day) Returned (Note 2): d right ascension (radians) RA * DEC d declination (radians) PMR d RA proper motion (radians/year) PMD d Dec proper motion (radians/year) * РΧ d parallax (arcsec) RV d radial velocity (km/s, positive = receding) J status: i 0 = OK -1 = superluminal speed (Note 5) -2 = null position vector * Notes: 1) The specified pv-vector is the coordinate direction (and its rate * of change) for the epoch at which the light leaving the star reached the solar-system barycenter. 2) The star data returned by this routine are "observables" for an imaginary observer at the solar-system barycenter. Proper motion and radial velocity are, strictly, in terms of barycentric coordinate time, TCB. For most practical applications, it is permissible to neglect the distinction between TCB and ordinary "proper" time on Earth (TT/TAI). The result will, as a rule, be limited by the intrinsic accuracy of the proper-motion and radial-velocity data; moreover, the supplied pv-vector is likely to be merely an intermediate result (for example generated by the routine iau_STARPV), so that a change of time unit will cancel out overall. In accordance with normal star-catalog conventions, the object's * right ascension and declination are freed from the effects of secular aberration. The frame, which is aligned to the catalog equator and equinox, is Lorentzian and centered on the SSB. * Summarizing, the specified pv-vector is for most stars almost * identical to the result of applying the standard geometrical "space motion" transformation to the catalog data. The differences, which are the subject of the Stumpff paper cited below, are: (i) In stars with significant radial velocity and proper motion, the constantly changing light-time distorts the apparent proper motion. Note that this is a classical, not a relativistic, effect. (ii) The transformation complies with special relativity. 3) Care is needed with units. The star coordinates are in radians and the proper motions in radians per Julian year, but the parallax is in arcseconds; the radial velocity is in $\ensuremath{\text{km/s}}\xspace$, but the pv-vector result is in AU and AU/day. * * 4) The proper motions are the rate of change of the right ascension * and declination at the catalog epoch and are in radians per Julian year. The RA proper motion is in terms of coordinate angle, not

* * *	true angle, a declinations.	and will thus be numerically larger at high
	assumed. If	e motion at constant speed in the inertial frame is the speed is greater than or equal to the speed of outine aborts with an error status.
<pre>* * 6) The inverse transformation is performed by the routine iau_ST *</pre>		
* (Called:	
*	iau_PN	decompose p-vector into modulus and direction
*	iau_PDP	scalar product of two p-vectors
*	iau_SXP	multiply p-vector by scalar
*	iau_PMP	p-vector minus p-vector
*	iau_PM	modulus of p-vector
*	iau_PPP	p-vector plus p-vector
*	iau_PV2S	pv-vector to spherical
*	iau_ANP	normalize angle into range 0 to 2pi
*		
* F	Reference:	
*		
*	Stumpff, P.,	Astron.Astrophys. 144, 232-240 (1985).
*		
*_		

```
SUBROUTINE iau_PVU ( DT, PV, UPV )
*+
*
  iau_PVU
*
*
  - - - - - - - -
*
*
  Update a pv-vector.
*
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
  Status: vector/matrix support routine.
*
*
  Given:
*
     DT
              d
                          time interval
*
     ΡV
              d(3,2)
                          pv-vector
*
*
  Returned:
*
     UPV
              d(3,2) p updated, v unchanged
*
*
  Notes:
*
  1) "Update" means "refer the position component of the vector
*
*
      to a new epoch DT time units from the existing epoch".
*
*
  2) The time units of DT must match those of the velocity.
*
*
  Called:
                 p-vector plus scaled p-vector
*
      iau_PPSP
*
      iau_CP
                  copy p-vector
*
*_
```

SUBROUTINE iau_PVUP (DT, PV, P) *+ * iau_PVUP * * _ _ _ _ _ _ _ _ _ _ * * Update a pv-vector, discarding the velocity component. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * DTd time interval pv-vector * ΡV d(3,2) * * Returned: * d(3) p-vector Ρ * * Notes: * * 1) "Update" means "refer the position component of the vector to a * new date DT time units from the existing date". * * 2) The time units of DT must match those of the velocity. * *_

SUBROUTINE iau_PVXPV (A, B, AXB) *+ * * iau_PVXPV * _ _ _ _ _ _ _ _ _ _ _ * * Outer (=vector=cross) product of two pv-vectors. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * А d(3,2) first pv-vector * d(3,2) second pv-vector В * * Returned: * d(3,2) A x B AXB * * Note: * * If the position and velocity components of the two pv-vectors are (Ap, Av) and (Bp, Bv), the result, A x B, is the pair of vectors (Ap x Bp, Ap x Bv + Av x Bp). The two vectors are the cross-product of the two p-vectors and its derivative. * * * * * Called: iau_CPV copy pv-vector iau_PXP vector product of two p-vectors iau_PPP p-vector plus p-vector * * * * *_

SUBROUTINE iau_PXP (A, B, AXB) *+ * _ _ _ * iau_PXP * _ _ _ _ _ _ _ _ _ * * p-vector outer (=vector=cross) product. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: d(3) first p-vector d(3) second p-vector * A * В * * Returned: * AXB d(3) AxB * *_

SUBROUTINE iau_RM2V (R, W) *+ * iau_RM2V * * _ _ _ _ _ * * Express an r-matrix as an r-vector. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * R d(3,3) rotation matrix * * Returned: * W d(3) rotation vector (Note 1) * * Notes: * * 1) A rotation matrix describes a rotation through some angle about some arbitrary axis called the Euler axis. The "rotation vector" * returned by this routine has the same direction as the Euler axis, and its magnitude is the angle in radians. (The magnitude and * direction can be separated by means of the routine iau_PN.) * 2) If R is null, so is the result. If R is not a rotation matrix the result is undefined. R must be proper (i.e. have a positive determinant) and real orthogonal (inverse = transpose). * * 3) The reference frame rotates clockwise as seen looking along the rotation vector from the origin. * * * *_

SUBROUTINE iau_RV2M (W, R) *+ * * iau_RV2M * _ _ _ _ _ _ _ _ _ * * Form the r-matrix corresponding to a given r-vector. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * W d(3) rotation vector (Note 1) * * Returned: * R d(3,3) rotation matrix * * Notes: * * 1) A rotation matrix describes a rotation through some angle about some arbitrary axis called the Euler axis. The "rotation vector" * * supplied to this routine has the same direction as the Euler axis, * and its magnitude is the angle in radians. * * 2) If W is null, the unit matrix is returned. * * 3) The reference frame rotates clockwise as seen looking along the * rotation vector from the origin. * *_

```
SUBROUTINE iau_RX ( PHI, R )
*+
*
*
   iau_RX
*
   _ _ _ _ _ _ _
*
*
  Rotate an r-matrix about the x-axis.
*
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: vector/matrix support routine.
*
*
   Given:
*
                         angle (radians)
     PHI
               d
*
*
   Given and returned:
*
     R
              d(3,3)
                        r-matrix, rotated
*
*
  Notes:
*
*
   1) Calling this routine with positive PHI incorporates in the
*
      supplied r-matrix R an additional rotation, about the x-axis,
*
      anticlockwise as seen looking towards the origin from positive x.
*
*
   2) The additional rotation can be represented by this matrix:
*
*
          (
            1
                      0
                                    0
                                           )
*
          (
                                           )
                 + cos(PHI) + sin(PHI)
*
             0
          (
                                           )
*
          (
*
             0
                - sin(PHI) + cos(PHI)
          (
                                           )
*
*_
```

SUBROUTINE iau_RXP (R, P, RP) *+ * _ _ _ iau_RXP * * _ _ _ _ _ _ _ _ _ * * Multiply a p-vector by an r-matrix. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: d(3,3) r-matrix d(3) p-vector * R * Ρ d(3) * * Returned: d(3) R * P * RP * * Called: * iau_CP copy p-vector * *_

SUBROUTINE iau_RXPV (R, PV, RPV) *+ * _ _ iau_RXPV * * - - - - - - - - -* * Multiply a pv-vector by an r-matrix. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: d(3,3) r-matrix d(3,2) pv-vector * R * PV * * Returned: * RPV d(3,2) R * PV * * Called: * iau_RXP product of r-matrix and p-vector * *_

SUBROUTINE iau_RXR (A, B, ATB) *+ * _ _ _ * iau_RXR * _ _ _ _ _ _ _ _ _ * * Multiply two r-matrices. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: d(3,3) first r-matrix d(3,3) second r-matrix * A * В * * Returned: * ATB d(3,3) A * B * * Called: * iau_CR copy r-matrix * *_

```
SUBROUTINE iau_RY ( THETA, R )
*+
*
*
   iau_RY
*
   _ _ _ _ _ _ _
*
*
  Rotate an r-matrix about the y-axis.
*
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: vector/matrix support routine.
*
*
  Given:
*
     THETA
             d
                        angle (radians)
*
*
   Given and returned:
*
     R
          d(3,3)
                       r-matrix, rotated
*
*
  Notes:
*
*
   1) Calling this routine with positive THETA incorporates in the
*
      supplied r-matrix R an additional rotation, about the y-axis,
*
      anticlockwise as seen looking towards the origin from positive y.
*
*
  2) The additional rotation can be represented by this matrix:
*
*
          ( + cos(THETA)
                            0
                                    - sin(THETA)
                                                   )
*
          (
*
                              1
                  0
                                          0
          (
                                                    )
*
          (
*
            + \sin(\text{THETA}) = 0 + \cos(\text{THETA})
          (
*
*_
```

```
SUBROUTINE iau_RZ ( PSI, R )
*+
*
*
   iau_RZ
*
   _ _ _ _ _ _ _
*
*
  Rotate an r-matrix about the z-axis.
*
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
  Status: vector/matrix support routine.
*
*
  Given:
*
                        angle (radians)
     PSI
              d
*
*
  Given and returned:
*
     R
          d(3,3) r-matrix, rotated
*
*
  Notes:
*
*
  1) Calling this routine with positive PSI incorporates in the
*
     supplied r-matrix R an additional rotation, about the z-axis,
*
      anticlockwise as seen looking towards the origin from positive z.
*
*
  2) The additional rotation can be represented by this matrix:
*
*
          ( + \cos(PSI) + \sin(PSI)
                                        0
                                           )
*
          (
                                            )
            - sin(PSI) + cos(PSI)
*
                                           )
                                       0
          (
*
          (
                                            )
*
                 0
                              0
                                         1
          (
                                           )
*
*_
```

DOUBLE PRECISION FUNCTION iau_S00 (DATE1, DATE2, X, Y) *+ * iau_S00 * * The CIO locator s, positioning the Celestial Intermediate Origin on * the equator of the Celestial Intermediate Pole, given the CIP's X,Y * coordinates. Compatible with IAU 2000A precession-nutation. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * * Given: DATE1,DATE2 TT as a 2-part Julian Date (Note 1) d * X,Y d CIP coordinates (Note 3) * * Returned: the CIO locator s in radians (Note 2) * iau_S00 d * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 2450123.7D0 000 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in * cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) The CIO locator s is the difference between the right ascensions of the same point in two systems: the two systems are the GCRS and the CIP,CIO, and the point is the ascending node of the CIP equator. The quantity s remains below 0.1 arcsecond * throughout 1900-2100. 3) The series used to compute s is in fact for s+XY/2, where X and Y are the x and y components of the CIP unit vector; this series is more compact than a direct series for s would be. This routine requires X,Y to be supplied by the caller, who is responsible for providing values that are consistent with the supplied date. 4) The model is consistent with the IAU 2000A precession-nutation. * Called: mean anomaly of the Moon iau_FAL03 * mean anomaly of the Sun iau_FALP03 iau_FAF03 mean argument of the latitude of the Moon iau_FAD03 mean elongation of the Moon from the Sun * mean longitude of the Moon's ascending node iau FAOM03 mean longitude of Venus mean longitude of Earth iau_FAVE03 iau FAE03 iau FAPA03 general accumulated precession in longitude References: * * Capitaine, N., Chapront, J., Lambert, S. and Wallace, P., * "Expressions for the Celestial Intermediate Pole and Celestial Ephemeris Origin consistent with the IAU 2000A precession-nutation

```
* model", Astron.Astrophys. 400, 1145-1154 (2003)
*
* n.b. The celestial ephemeris origin (CEO) was renamed "celestial
* intermediate origin" (CIO) by IAU 2006 Resolution 2.
*
* McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003),
* IERS Technical Note No. 32, BKG (2004)
*-
```

DOUBLE PRECISION FUNCTION iau_S00A (DATE1, DATE2) *+ * i a u _ S 0 0 A * * The CIO locator s, positioning the Celestial Intermediate Origin on * the equator of the Celestial Intermediate Pole, using the IAU 2000A * precession-nutation model. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: DATE1,DATE2 TT as a 2-part Julian Date (Note 1) d * Returned: * iau_S00A d the CIO locator s in radians (Note 2) * * Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) -1421.3D0 (J2000 method) 2451545D0 50123.2D0 2400000.5D0 (MJD method) 0.2D0 (date & time method) * 2450123.5D0 The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * 2) The CIO locator s is the difference between the right ascensions * of the same point in two systems. The two systems are the GCRS and the CIP,CIO, and the point is the ascending node of the * CIP equator. The CIO locator s remains a small fraction of 1 arcsecond throughout 1900-2100. * 3) The series used to compute s is in fact for s+XY/2, where X and Y are the x and y components of the CIP unit vector; this series is more compact than a direct series for s would be. The present * * routine uses the full IAU 2000A nutation model when predicting the CIP position. Faster results, with no significant loss of accuracy, can be obtained via the routine iau_S00B, which uses * instead the IAU 2000B truncated model. * Called: iau_PNM00A classical NPB matrix, IAU 2000A extract CIP X,Y from the BPN matrix iau_BNP2XY iau S00 the CIO locator s, given X,Y, IAU 2000A References: Capitaine, N., Chapront, J., Lambert, S. and Wallace, P., "Expressions for the Celestial Intermediate Pole and Celestial Ephemeris Origin consistent with the IAU 2000A precession-nutation model", Astron.Astrophys. 400, 1145-1154 (2003) * n.b. The celestial ephemeris origin (CEO) was renamed "celestial intermediate origin" (CIO) by IAU 2006 Resolution 2. McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003),

IERS Technical Note No. 32, BKG (2004)

* * *_

DOUBLE PRECISION FUNCTION iau_S00B (DATE1, DATE2) *+ * i a u _ S 0 0 B * * The CIO locator s, positioning the Celestial Intermediate Origin on * the equator of the Celestial Intermediate Pole, using the IAU 2000B * precession-nutation model. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: DATE1,DATE2 TT as a 2-part Julian Date (Note 1) d * Returned: * iau_S00B d the CIO locator s in radians (Note 2) * * Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) -1421.3D0 (J2000 method) 2451545D0 50123.2D0 2400000.5D0 (MJD method) 0.2D0 (date & time method) * 2450123.5D0 The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * 2) The CIO locator s is the difference between the right ascensions * of the same point in two systems. The two systems are the GCRS and the CIP,CIO, and the point is the ascending node of the CIP equator. The CIO locator s remains a small fraction of 1 arcsecond throughout 1900-2100. * 3) The series used to compute s is in fact for s+XY/2, where X and Y are the x and y components of the CIP unit vector; this series is more compact than a direct series for s would be. The present * * routine uses the IAU 2000B truncated nutation model when predicting the CIP position. The routine iau_S00A uses instead the full IAU 2000A model, but with no significant increase in * accuracy and at some cost in speed. * Called: iau_PNM00B classical NPB matrix, IAU 2000B extract CIP X,Y from the BPN matrix iau_BNP2XY iau S00 the CIO locator s, given X,Y, IAU 2000A References: Capitaine, N., Chapront, J., Lambert, S. and Wallace, P., "Expressions for the Celestial Intermediate Pole and Celestial Ephemeris Origin consistent with the IAU 2000A precession-nutation model", Astron.Astrophys. 400, 1145-1154 (2003) * n.b. The celestial ephemeris origin (CEO) was renamed "celestial intermediate origin" (CIO) by IAU 2006 Resolution 2. McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003),

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* * *_

DOUBLE PRECISION FUNCTION iau_S06 (DATE1, DATE2, X, Y) *+ * iau_S06 * * The CIO locator s, positioning the Celestial Intermediate Origin on * the equator of the Celestial Intermediate Pole, given the CIP's X,Y * coordinates. Compatible with IAU 2006/2000A precession-nutation. This routine is part of the International Astronomical Union's SOFA (Standards of Fundamental Astronomy) software collection. * Status: canonical model. * * Given: DATE1,DATE2 TT as a 2-part Julian Date (Note 1) d * X,Y d CIP coordinates (Note 3) * * Returned: the CIO locator s in radians (Note 2) * iau_S06 d * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) The CIO locator s is the difference between the right ascensions of the same point in two systems: the two systems are the GCRS and the CIP, CIO, and the point is the ascending node of the CIP equator. The quantity s remains below 0.1 arcsecond * throughout 1900-2100. 3) The series used to compute s is in fact for s+XY/2, where X and Y are the x and y components of the CIP unit vector; this series is more compact than a direct series for s would be. This routine requires X,Y to be supplied by the caller, who is responsible for providing values that are consistent with the supplied date. 4) The model is consistent with the "P03" precession (Capitaine et al. 2003), adopted by IAU 2006 Resolution 1, 2006, and the IAU 2000A nutation (with PO3 adjustments). Called: iau_FAL03 mean anomaly of the Moon iau_FALP03 mean anomaly of the Sun mean argument of the latitude of the Moon iau_FAF03 iau_FAD03 mean elongation of the Moon from the Sun iau FAOM03 mean longitude of the Moon's ascending node mean longitude of Venus iau FAVE03 iau_FAE03 mean longitude of Earth iau_FAPA03 general accumulated precession in longitude * References: Capitaine, N., Wallace, P.T. & Chapront, J., 2003, Astron.

* Astrophys. 432, 355
*
* McCarthy, D.D., Petit, G. (eds.) 2004, IERS Conventions (2003),
* IERS Technical Note No. 32, BKG
*
*_

DOUBLE PRECISION FUNCTION iau_S06A (DATE1, DATE2) *+ * iau_ S 0 6 A * * The CIO locator s, positioning the Celestial Intermediate Origin on * the equator of the Celestial Intermediate Pole, using the IAU 2006 * precession and IAU 2000A nutation models. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: DATE1,DATE2 TT as a 2-part Julian Date (Note 1) d * Returned: * iau_S06A d the CIO locator s in radians (Note 2) * * Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 50123.2D0 2400000.5D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the * optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * 2) The CIO locator s is the difference between the right ascensions * of the same point in two systems. The two systems are the GCRS and the CIP,CIO, and the point is the ascending node of the CIP equator. The CIO locator s remains a small fraction of 1 arcsecond throughout 1900-2100. * 3) The series used to compute s is in fact for s+XY/2, where X and Y are the x and y components of the CIP unit vector; this series is more compact than a direct series for s would be. The present * * routine uses the full IAU 2000A nutation model when predicting the CIP position. * Called: * iau_PNM06A classical NPB matrix, IAU 2006/2000A extract CIP X,Y coordinates from NPB matrix iau BPN2XY iau_S06 the CIO locator s, given X,Y, IAU 2006 References: Capitaine, N., Chapront, J., Lambert, S. and Wallace, P., "Expressions for the Celestial Intermediate Pole and Celestial Ephemeris Origin consistent with the IAU 2000A precession-nutation model", Astron.Astrophys. 400, 1145-1154 (2003) n.b. The celestial ephemeris origin (CEO) was renamed "celestial intermediate origin" (CIO) by IAU 2006 Resolution 2. Capitaine, N. & Wallace, P.T., 2006, Astron.Astrophys. 450, 855 McCarthy, D. D., Petit, G. (eds.), 2004, IERS Conventions (2003),

* IERS Technical Note No. 32, BKG
*
* Wallace, P.T. & Capitaine, N., 2006, Astron.Astrophys. 459, 981
*
*-

SUBROUTINE iau_S2C (THETA, PHI, C) *+ * _ _ -* iau_S2C * _ _ _ _ _ _ _ _ _ * * Convert spherical coordinates to Cartesian. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * THETA d longitude angle (radians) d * PHI latitude angle (radians) * * Returned: * d(3) direction cosines С * *_

```
SUBROUTINE iau_S2P ( THETA, PHI, R, P )
*+
*
*
  iau_S2P
*
  _ _ _ _ _ _ _ _
*
*
  Convert spherical polar coordinates to p-vector.
*
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
  Status: vector/matrix support routine.
*
*
  Given:
*
     THETA
              d
                        longitude angle (radians)
                      latitude angle (radians)
*
              d
     PHI
*
                       radial distance
     R
              d
*
*
  Returned:
*
             d(3) Cartesian coordinates
     Ρ
*
*
  Called:
                spherical coordinates to unit vector
multiply p-vector by scalar
*
     iau_S2C
*
     iau_SXP
*
*_
```

SUBROUTINE iau_S2PV (THETA, PHI, R, TD, PD, RD, PV) *+ * * iau_S2PV * _ _ _ _ _ _ _ _ _ * * Convert position/velocity from spherical to Cartesian coordinates. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: * THETA d longitude angle (radians) latitude angle (radians) * PHI d radial distance rate of change of THETA rate of change of PHI rate of change of R * d d R * TDd * PD * RD d * * Returned: * PV d(3,2) pv-vector * *_

SUBROUTINE iau_S2XPV (S1, S2, PV, SPV) *+ * * iau_S2XPV * _ _ _ _ _ _ _ _ _ _ _ * * Multiply a pv-vector by two scalars. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: scalar to multiply position component by scalar to multiply velocity component by * d S1 d d * S2 * d(3,2) pv-vector ΡV * * Returned: * d(3,2) pv-vector: p scaled by S1, v scaled by S2 SPV * * Called: * iau_SXP multiply p-vector by scalar * *_

```
SUBROUTINE iau_SEPP ( A, B, S )
*+
*
*
   iau_SEPP
*
            _ _ _ _ _
   _ _ _ _
*
*
  Angular separation between two p-vectors.
*
*
   This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: vector/matrix support routine.
*
*
   Given:
*
     А
                d(3)
                           first p-vector (not necessarily unit length)
*
                d(3)
                           second p-vector (not necessarily unit length)
      В
*
*
   Returned:
*
                           angular separation (radians, always positive)
     S
                d
*
*
  Notes:
*
*
   1) If either vector is null, a zero result is returned.
*
   2) The angular separation is most simply formulated in terms of
*
      scalar product. However, this gives poor accuracy for angles
near zero and pi. The present algorithm uses both cross product
*
*
      and dot product, to deliver full accuracy whatever the size of
*
      the angle.
*
  Called:
*
      iau_PXP
                    vector product of two p-vectors
*
                    modulus of p-vector
      iau_PM
*
      iau PDP
                   scalar product of two p-vectors
*
*_
```

```
SUBROUTINE iau_SEPS ( AL, AP, BL, BP, S )
*+
*
*
   iau_SEPS
*
  _ _ _ _ _ _ _ _ _ _
*
*
  Angular separation between two sets of spherical coordinates.
*
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: vector/matrix support routine.
*
*
  Given:
*
                           first longitude (radians)
      AL
                d
                          first latitude (radians)
*
      AP
                d
*
                          second longitude (radians)
second latitude (radians)
                d
      BL
*
      ΒP
                d
*
*
  Returned:
*
               d
                          angular separation (radians)
      S
*
*
  Called:
      iau_S2C spherical coordinates to unit vector
iau_SEPP angular separation between two p-vectors
*
*
*
*_
```

DOUBLE PRECISION FUNCTION iau_SP00 (DATE1, DATE2) *+ * iau_SP00 * * * The TIO locator s', positioning the Terrestrial Intermediate Origin * on the equator of the Celestial Intermediate Pole. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * iau SP00 d the TIO locator s' in radians (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others: * DATE1 * DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. 2) The TIO locator $\mathbf{s}^{\,\prime}$ is obtained from polar motion observations by * numerical integration, and so is in essence unpredictable. * However, it is dominated by a secular drift of about 47 microarcseconds per century, which is the approximation * evaluated by the present routine. Reference: * McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * * *_

SUBROUTINE iau_STARPM (RA1, DEC1, PMR1, PMD1, PX1, RV1, : EP1A, EP1B, EP2A, EP2B, RA2, DEC2, PMR2, PMD2, PX2, RV2, J) *+ * _ _ _ _ _ _ _ * iau_STARPM * _ _ _ _ _ . * * Star proper motion: update star catalog data for space motion. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * * Given: * right ascension (radians), before RA1 d * DEC1 declination (radians), before d * PMR1 d RA proper motion (radians/year), before * d PMD1 Dec proper motion (radians/year), before * PX1 d parallax (arcseconds), before * RV1 d radial velocity (km/s, +ve = receding), before "before" epoch, part A (Note 1) "before" epoch, part B (Note 1) "after" epoch, part A (Note 1) EP1A d * EP1B d * EP2A d * "after" epoch, part B (Note 1) EP2B d * * Returned: * d right ascension (radians), after RA2 DEC2 d declination (radians), after * RA proper motion (radians/year), after PMR2 d * PMD2 Dec proper motion (radians/year), after d * PX2 d parallax (arcseconds), after * radial velocity (km/s, +ve = receding), after RV2 d * i J status: -1 = system error (should not occur) * 0 = no warnings or errors * 1 = distance overridden (Note 6) 2 = excessive velocity (Note 7) * 4 = solution didn't converge (Note 8) else = binary logical OR of the above warnings * * Notes: * 1) The starting and ending TDB epochs EP1A+EP1B and EP2A+EP2B are * Julian Dates, apportioned in any convenient way between the two parts (A and B). For example, JD(TDB)=2450123.7 could be expressed in any of these ways, among others: * * * EPnA EPnB * 2450123.7D0 0D0 (JD method) -1421.3D0 * 2451545D0 (J2000 method) 50123.2D0 2400000.5D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) * The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. * 2) In accordance with normal star-catalog conventions, the object's right ascension and declination are freed from the effects of * secular aberration. The frame, which is aligned to the catalog * equator and equinox, is Lorentzian and centered on the SSB. * * The proper motions are the rate of change of the right ascension and declination at the catalog epoch and are in radians per TDB Julian year. *

*		The parallax	and radial velocity are in the same frame.		
* * *	3)	and the prope	ed with units. The star coordinates are in radians er motions in radians per Julian year, but the in arcseconds.		
* * * *	4)	angle. If th	r motion is in terms of coordinate angle, not true ne catalog uses arcseconds for both RA and Dec proper RA proper motion will need to be divided by cos(Dec)		
* * *	5)	Straight-line is assumed.	e motion at constant speed, in the inertial frame,		
* * * *	6)	An extremely small (or zero or negative) parallax is interpreted to mean that the object is on the "celestial sphere", the radius of which is an arbitrary (large) value (see the iau_STARPV routine for the value used). When the distance is overridden in this way, the status, initially zero, has 1 added to it.			
* * * *	7)	If the space velocity is a significant fraction of c (see the constant VMAX in the routine iau_STARPV), it is arbitrarily set to zero. When this action occurs, 2 is added to the status.			
* * * *	8)	The relativistic adjustment carried out in the iau_STARPV routine involves an iterative calculation. If the process fails to converge within a set number of iterations, 4 is added to the status.			
* * * * * *	Ca	lled: iau_STARPV iau_PVU iau_PDP iau_PVSTAR	star catalog data to space motion pv-vector update a pv-vector scalar product of two p-vectors space motion pv-vector to star catalog data		

SUBROUTINE iau_STARPV (RA, DEC, PMR, PMD, PX, RV, PV, J) *+ * i a u _ S T A R P V * * Convert star catalog coordinates to position+velocity vector. * * This routine is part of the International Astronomical Union's SOFA (Standards of Fundamental Astronomy) software collection. Status: support routine. * Given (Note 1): * RA d right ascension (radians) DEC d declination (radians) PMR RA proper motion (radians/year) d PMD d Dec proper motion (radians/year) parallax (arcseconds) РΧ d radial velocity (km/s, positive = receding) RV d * Returned (Note 2): ΡV d(3,2) pv-vector (AU, AU/day) J i status: 0 = no warnings 1 = distance overridden (Note 6) 2 = excessive velocity (Note 7) 4 = solution didn't converge (Note 8) else = binary logical OR of the above * Notes: 1) The star data accepted by this routine are "observables" for an imaginary observer at the solar-system barycenter. Proper motion and radial velocity are, strictly, in terms of barycentric coordinate time, TCB. For most practical applications, it is permissible to neglect the distinction between TCB and ordinary "proper" time on Earth (TT/TAI). The result will, as a rule, be limited by the intrinsic accuracy of the proper-motion and radial-* velocity data; moreover, the pv-vector is likely to be merely an * intermediate result, so that a change of time unit would cancel out overall. * * In accordance with normal star-catalog conventions, the object's right ascension and declination are freed from the effects of secular aberration. The frame, which is aligned to the catalog equator and equinox, is Lorentzian and centered on the SSB. * 2) The resulting position and velocity pv-vector is with respect to * the same frame and, like the catalog coordinates, is freed from the effects of secular aberration. Should the "coordinate * direction", where the object was located at the catalog epoch, be required, it may be obtained by calculating the magnitude of the position vector PV(1-3,1) dividing by the speed of light in AU/day to give the light-time, and then multiplying the space velocity PV(1-3,2) by this light-time and adding the result to PV(1-3,1). * Summarizing, the pv-vector returned is for most stars almost identical to the result of applying the standard geometrical "space motion" transformation. The differences, which are the * subject of the Stumpff paper referenced below, are: (i) In stars with significant radial velocity and proper motion, the constantly changing light-time distorts the apparent proper motion. Note that this is a classical, not a relativistic, effect. (ii) The transformation complies with special relativity. * * 3) Care is needed with units. The star coordinates are in radians * and the proper motions in radians per Julian year, but the parallax is in arcseconds; the radial velocity is in km/s, but

```
*
      the pv-vector result is in AU and AU/day.
*
   4) The RA proper motion is in terms of coordinate angle, not true
      angle. If the catalog uses arcseconds for both RA and Dec proper
*
      motions, the RA proper motion will need to be divided by cos(Dec)
*
      before use.
   5) Straight-line motion at constant speed, in the inertial frame,
*
      is assumed.
*
   6) An extremely small (or zero or negative) parallax is interpreted
      to mean that the object is on the "celestial sphere", the radius
      of which is an arbitrary (large) value (see the constant PXMIN).
      When the distance is overridden in this way, the status, initially
      zero, has 1 added to it.
   7) If the space velocity is a significant fraction of c (see the
      constant VMAX), it is arbitrarily set to zero. When this action
      occurs, 2 is added to the status.
   8) The relativistic adjustment involves an iterative calculation.
      If the process fails to converge within a set number (IMAX) of
      iterations, 4 is added to the status.
   9) The inverse transformation is performed by the routine iau_PVSTAR.
   Called:
      iau_S2PV
*
                   spherical coordinates to pv-vector
      iau_PM
                   modulus of p-vector
*
      iau_ZP
                   zero p-vector
                   decompose p-vector into modulus and direction
      iau_PN
                   scalar product of two p-vectors
*
      iau_PDP
      iau_SXP
                   multiply p-vector by scalar
      iau_PMP
                   p-vector minus p-vector
      iau PPP
                   p-vector plus p-vector
*
  Reference:
*
*
      Stumpff, P., Astron.Astrophys. 144, 232-240 (1985).
*
*_
```

SUBROUTINE iau_SXP (S, P, SP) *+ * - - -- -* iau_SXP * _ _ _ _ _ _ _ _ _ * * Multiply a p-vector by a scalar. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: d scalar d(3) p-vector * S * Ρ * * Returned: * d(3) S * P SP * *_

SUBROUTINE iau_SXPV (S, PV, SPV) *+ * _ _ iau_SXPV * * - - - - - - - - -* * Multiply a pv-vector by a scalar. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: d scalar d(3,2) pv-vector * S * ΡV * * Returned: * SPV d(3,2) S * PV * * Called: * iau_S2XPV multiply pv-vector by two scalars * *_

```
SUBROUTINE iau_TAITT ( TAI1, TAI2, TT1, TT2, J )
*+
*
*
   iau_TAITT
*
*
   Time scale transformation: International Atomic Time, TAI, to
*
*
   Terrestrial Time, TT.
*
   This routine is part of the International Astronomical Union's
*
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical.
*
*
   Given:
*
     TAI1,TAI2
                     d
                             TAI as a 2-part Julian Date
*
*
   Returned:
                      d TT as a 2-part Julian Date
*
      TT1,TT2
                              status: 0 = 0K
*
      J
                      i
*
*
   Note:
      TAI1+TAI2 is Julian Date, apportioned in any convenient way
between the two arguments, for example where TAI1 is the Julian
Day Number and TAI2 is the fraction of a day. The returned
*
*
*
       TT1,TT2 follow suit.
*
*
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
*
*
       Explanatory Supplement to the Astronomical Almanac,
*
      P. Kenneth Seidelmann (ed), University Science Books (1992)
*_
```

```
SUBROUTINE iau_TAIUT1 ( TAI1, TAI2, DTA, UT11, UT12, J )
*+
*
*
   iau_TAIUT1
*
            _ _ _ _ _
*
  Time scale transformation: International Atomic Time, TAI, to Universal Time, UT1.
*
*
*
   This routine is part of the International Astronomical Union's
*
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical.
*
*
   Given:
                          TAI as a 2-part Julian Date
UT1-TAI in seconds
*
     TAI1,TAI2 d
*
                    d
      DTA
*
*
   Returned:
*
                  d UT1 as a 2-part Julian Date
i status: 0 = OK
      UT11,UT12
*
      J
*
*
   Notes:
*
*
   1) TAI1+TAI2 is Julian Date, apportioned in any convenient way
      between the two arguments, for example where TAI1 is the Julian
Day Number and TAI2 is the fraction of a day. The returned
*
*
*
      UT11,UT12 follow suit.
   2) The argument DTA, i.e. UT1-TAI, is an observed quantity, and is
*
*
      available from IERS tabulations.
*
*
  Reference:
*
*
      Explanatory Supplement to the Astronomical Almanac,
*
      P. Kenneth Seidelmann (ed), University Science Books (1992)
*
*_
```

```
SUBROUTINE iau_TAIUTC ( TAI1, TAI2, UTC1, UTC2, J )
*+
*
   iau_TAIUTC
*
*
*
  Time scale transformation: International Atomic Time, TAI, to
*
  Coordinated Universal Time, UTC.
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
  Status: canonical.
*
*
  Given:
*
     TAI1,TAI2
                  d
                         TAI as a 2-part Julian Date (Note 1)
*
  Returned:
     UTC1,UTC2
                   d
                          UTC as a 2-part quasi Julian Date (Notes 1-3)
                          status: +1 = dubious year (Note 4)
      ιT
                   i
                                   0 = OK
                                   -1 = unacceptable date
*
  Notes:
  1) TAI1+TAI2 is Julian Date, apportioned in any convenient way
      between the two arguments, for example where TAI1 is the Julian
      Day Number and TAI2 is the fraction of a day. The returned UTC1
      and UTC2 form an analogous pair, except that a special convention
      is used, to deal with the problem of leap seconds - see the next
     note.
   2) JD cannot unambiguously represent UTC during a leap second unless
      special measures are taken. The convention in the present routine
      is that the JD day represents UTC days whether the length is
      86399, 86400 or 86401 SI seconds.
  3) The routine iau_D2DTF can be used to transform the UTC quasi-JD \,
      into calendar date and clock time, including UTC leap second
*
      handling.
   4) The warning status "dubious year" flags UTCs that predate the
      introduction of the time scale and that are too far in the future
      to be trusted. See iau_DAT for further details.
*
  Called:
                  JD to Gregorian calendar
*
      iau JD2CAL
                   delta(AT) = TAI-UTC
      iau DAT
      iau_CAL2JD Gregorian calendar to JD
*
  References:
     McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
      Explanatory Supplement to the Astronomical Almanac,
*
      P. Kenneth Seidelmann (ed), University Science Books (1992)
*
*
```

SUBROUTINE iau_TCBTDB (TCB1, TCB2, TDB1, TDB2, J) *+ * iau_TCBTDB * * Time scale transformation: Barycentric Coordinate Time, TCB, to * Barycentric Dynamical Time, TDB. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical. * * Given: * TCB1,TCB2 d TCB as a 2-part Julian Date * Returned: TDB1,TDB2 d TDB as a 2-part Julian Date status: 0 = OKιT i * * Notes: 1) TCB1+TCB2 is Julian Date, apportioned in any convenient way between the two arguments, for example where TCB1 is the Julian Day Number and TCB2 is the fraction of a day. The returned TDB1, TDB2 follow suit. 2) The 2006 IAU General Assembly introduced a conventional linear transformation between TDB and TCB. This transformation compensates for the drift between TCB and terrestrial time TT, and keeps TDB approximately centered on TT. Because the relationship between TT and TCB depends on the adopted solar system ephemeris, the degree of alignment between TDB and TT over long intervals will vary according to which ephemeris is used. Former definitions of TDB attempted to avoid this problem by stipulating that TDB and TT should differ only by periodic effects. This is a good description of the nature of the relationship but eluded precise mathematical formulation. The conventional linear relationship adopted in 2006 sidestepped these difficulties whilst delivering a TDB that in practice was * consistent with values before that date. 3) TDB is essentially the same as Teph, the time argument for the JPL solar system ephemerides. * Reference: * IAU 2006 Resolution B3 * *_

```
SUBROUTINE iau_TCGTT ( TCG1, TCG2, TT1, TT2, J )
*+
*
*
   iau_TCGTT
*
             _ _ _ _ _
   _ _ _ _
*
*
   Time scale transformation: Geocentric Coordinate Time, TCG, to
*
   Terrestrial Time, TT.
*
   This routine is part of the International Astronomical Union's
*
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical.
*
*
   Given:
                             TCG as a 2-part Julian Date
*
     TCG1,TCG2
                     d
*
*
   Returned:
                      d TT as a 2-part Julian Date
*
      TT1,TT2
*
                              status: 0 = OK
       J
                       i
*
*
   Note:
      TCG1+TCG2 is Julian Date, apportioned in any convenient way
between the two arguments, for example where TCG1 is the Julian
Day Number and TCG2 is the fraction of a day. The returned
*
*
*
       TT1,TT2 follow suit.
*
*
*
   References:
       McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003),. IERS Technical Note No. 32, BKG (2004)
*
*
*
*
       IAU 2000 Resolution B1.9
*
*_
```

SUBROUTINE iau_TDBTCB (TDB1, TDB2, TCB1, TCB2, J) *+ * * іаи_ТДВТСВ * * Time scale transformation: Barycentric Dynamical Time, TDB, to * Barycentric Coordinate Time, TCB. This routine is part of the International Astronomical Union's * * SOFA (Standards of Fundamental Astronomy) software collection. * Status: canonical. * * Given: * TDB1,TDB2 d TDB as a 2-part Julian Date * Returned: * TCB1,TCB2 d TCB as a 2-part Julian Date * J i status: 0 = OK* * Notes: 1) TDB1+TDB2 is Julian Date, apportioned in any convenient way between the two arguments, for example where TDB1 is the Julian Day Number and TDB2 is the fraction of a day. The returned * TCB1,TCB2 follow suit. 2) The 2006 IAU General Assembly introduced a conventional linear transformation between TDB and TCB. This transformation compensates for the drift between TCB and terrestrial time TT, * and keeps TDB approximately centered on TT. Because the relationship between TT and TCB depends on the adopted solar * system ephemeris, the degree of alignment between TDB and TT over long intervals will vary according to which ephemeris is used. Former definitions of TDB attempted to avoid this problem by stipulating that TDB and TT should differ only by periodic effects. This is a good description of the nature of the * relationship but eluded precise mathematical formulation. The conventional linear relationship adopted in 2006 sidestepped these difficulties whilst delivering a TDB that in practice was consistent with values before that date. 3) TDB is essentially the same as Teph, the time argument for the JPL solar system ephemerides. Reference: * * IAU 2006 Resolution B3 *_

SUBROUTINE iau_TDBTT (TDB1, TDB2, DTR, TT1, TT2, J) *+ * iau_TDBTT * * * Time scale transformation: Barycentric Dynamical Time, TDB, to * Terrestrial Time, TT. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical. * * Given: * TDB1,TDB2 d TDB as a 2-part Julian Date * DTR TDB-TT in seconds d * Returned: * d TT as a 2-part Julian Date i status: 0 = OK TT1,TT2 * Л * * Notes: 1) TDB1+TDB2 is Julian Date, apportioned in any convenient way between the two arguments, for example where TDB1 is the Julian Day Number and TDB2 is the fraction of a day. The returned * TT1,TT2 follow suit. 2) The argument DTR represents the quasi-periodic component of the GR transformation between TT and TCB. It is dependent upon the adopted solar-system ephemeris, and can be obtained by numerical * integration, by interrogating a precomputed time ephemeris or by * evaluating a model such as that implemented in the SOFA routine iau_DTDB. The quantity is dominated by an annual term of 1.7 ms amplitude. 3) TDB is essentially the same as Teph, the time argument for the JPL solar system ephemerides. * References: McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * * IAU 2006 Resolution 3 * *_

```
SUBROUTINE iau_TF2A ( S, IHOUR, IMIN, SEC, RAD, J )
*+
*
*
   iau_TF2A
*
            _ _ _ _
*
*
  Convert hours, minutes, seconds to radians.
*
*
   This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: support routine.
*
   Given:
                         sign: '-' = negative, otherwise positive
*
      S
                  С
*
      IHOUR
                 i
                        hours
*
      IMIN
                  i
                         minutes
*
      SEC
                  d
                         seconds
*
*
   Returned:
*
                         angle in radians
      RAD
                  d
*
      J
                         status: 0 = OK
                  i
                                   1 = IHOUR outside range 0-23
                                   2 = IMIN outside range 0-59
                                   3 = SEC outside range 0-59.999...
*
  Notes:
*
*
   1) If the s argument is a string, only the leftmost character is
*
       used and no warning status is provided.
*
*
   2) The result is computed even if any of the range checks fail.
*

    Negative IHOUR, IMIN and/or SEC produce a warning status, but the
absolute value is used in the conversion.

*
*
*
*
   4) If there are multiple errors, the status value reflects only the
*
       first, the smallest taking precedence.
*
*_
```

```
SUBROUTINE iau_TF2D ( S, IHOUR, IMIN, SEC, DAYS, J )
*+
*
*
   iau_TF2D
*
           _ _ _ _ _
*
*
  Convert hours, minutes, seconds to days.
*
*
   This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: support routine.
*
   Given:
                         sign: '-' = negative, otherwise positive
*
      S
                  С
*
      IHOUR
                 i
                        hours
*
      IMIN
                  i
                         minutes
*
      SEC
                  d
                         seconds
*
*
   Returned:
*
                         interval in days
      DAYS
                  d
*
      J
                         status: 0 = OK
                  i
                                   1 = IHOUR outside range 0-23
                                   2 = IMIN outside range 0-59
                                   3 = SEC outside range 0-59.999...
*
  Notes:
*
*
   1) If the s argument is a string, only the leftmost character is
*
       used and no warning status is provided.
*
*
   2) The result is computed even if any of the range checks fail.
*

    Negative IHOUR, IMIN and/or SEC produce a warning status, but the
absolute value is used in the conversion.

*
*
*
*
   4) If there are multiple errors, the status value reflects only the
*
       first, the smallest taking precedence.
*
*_
```

```
SUBROUTINE iau_TR ( R, RT )
*+
*
    - - - - -
  iau_TR
-----
*
*
*
*
  Transpose an r-matrix.
*
*
  This routine is part of the International Astronomical Union's
* SOFA (Standards of Fundamental Astronomy) software collection.
*
*
  Status: vector/matrix support routine.
*
*
  Given:
*
             d(3,3) r-matrix
    R
*
*
  Returned:
*
           d(3,3) transpose
    RT
*
* Called:
*
    iau_CR copy r-matrix
*
*_
```

SUBROUTINE iau_TRXP (R, P, TRP) *+ * _ _ _ iau_TRXP * * _ _ _ _ _ _ _ _ _ _ * * Multiply a p-vector by the transpose of an r-matrix. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: d(3,3) r-matrix d(3) p-vector * R * Ρ d(3) * * Returned: * d(3) R*P TRP * * Called: * iau_TR iau_TR transpose r-matrix iau_RXP product of r-matrix and p-vector * * *_

SUBROUTINE iau_TRXPV (R, PV, TRPV) *+ * iau_TRXPV * * _ _ _ _ _ _ _ _ _ _ _ * * Multiply a pv-vector by the transpose of an r-matrix. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Given: d(3,3) r-matrix d(3,2) pv-vector * R * PV * * Returned: * TRPV d(3,2) R * PV * * Called: * iau_TR transpose r-matrix iau_RXPV product of r-matrix and pv-vector iau_TR * * *_

```
SUBROUTINE iau_TTTAI ( TT1, TT2, TAI1, TAI2, J )
*+
*
*
   iau_TTTAI
*
*
*
  Time scale transformation: Terrestrial Time, TT, to International
*
  Atomic Time, TAI.
*
   This routine is part of the International Astronomical Union's
*
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
  Status: canonical.
*
*
   Given:
*
     TT1,TT2
                            TT as a 2-part Julian Date
                    d
*
*
   Returned:
                  d TAI as a 2-part Julian Date
i status: 0 = OK
*
      TAI1,TAI2
*
      J
*
*
  Note:
      TT1+TT2 is Julian Date, apportioned in any convenient way between the two arguments, for example where TT1 is the Julian Day Number
*
*
*
      and TT2 is the fraction of a day. The returned TAI1, TAI2 follow
*
      suit.
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
*
*
      Explanatory Supplement to the Astronomical Almanac,
*
      P. Kenneth Seidelmann (ed), University Science Books (1992)
*_
```

```
SUBROUTINE iau_TTTCG ( TT1, TT2, TCG1, TCG2, J )
*+
*
*
   iau_TTTCG
*
             _ _ _ _
   _ _ _ _
*
  Time scale transformation: Terrestrial Time, TT, to Geocentric
*
*
  Coordinate Time, TCG.
*
   This routine is part of the International Astronomical Union's
*
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical.
*
*
   Given:
                            TT as a 2-part Julian Date
*
     TT1,TT2
                    d
*
*
   Returned:
                  d TCG as a 2-part Julian Date
i status: 0 = OK
*
      TCG1,TCG2
*
      J
*
*
   Note:
      TT1+TT2 is Julian Date, apportioned in any convenient way between the two arguments, for example where TT1 is the Julian Day Number
*
*
*
       and TT2 is the fraction of a day. The returned TCG1, TCG2 follow
*
       suit.
*
*
   References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
*
*
      IAU 2000 Resolution B1.9
*
*_
```

SUBROUTINE iau_TTTDB (TT1, TT2, DTR, TDB1, TDB2, J) *+ * i a u _ T T T D B * * * Time scale transformation: Terrestrial Time, TT, to Barycentric * Dynamical Time, TDB. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical. * * Given: * TT1,TT2 d TT as a 2-part Julian Date * DTR TDB-TT in seconds d * Returned: * d TDB as a 2-part Julian Date i status: 0 = OK TDB1,TDB2 * J * * Notes: 1) TT1+TT2 is Julian Date, apportioned in any convenient way between the two arguments, for example where TT1 is the Julian Day Number and TT2 is the fraction of a day. The returned TDB1, TDB2 follow * suit. 2) The argument DTR represents the quasi-periodic component of the GR transformation between TT and TCB. It is dependent upon the adopted solar-system ephemeris, and can be obtained by numerical * integration, by interrogating a precomputed time ephemeris or by * evaluating a model such as that implemented in the SOFA routine iau_DTDB. The quantity is dominated by an annual term of 1.7 ms amplitude. 3) TDB is essentially the same as Teph, the time argument for the JPL solar system ephemerides. * References: McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * * IAU 2006 Resolution 3 * *_

```
SUBROUTINE iau_TTUT1 ( TT1, TT2, DT, UT11, UT12, J )
*+
*
*
   iau_TTUT1
*
   _ _ _ _
             _ _ _ _ _
*
*
   Time scale transformation: Terrestrial Time, TT, to Universal Time,
*
   UT1.
*
*
   This routine is part of the International Astronomical Union's
*
   SOFA (Standards of Fundamental Astronomy) software collection.
*
*
   Status: canonical.
*
*
   Given:
                            TT as a 2-part Julian Date
TT-UT1 in seconds
*
    TT1,TT2
                     d
*
      DT
                      d
*
*
   Returned:
*
                   d UT1 as a 2-part Julian Date
i status: 0 = OK
      UT11,UT12
*
       J
*
*
   Notes:
*
   1) TT1+TT2 is Julian Date, apportioned in any convenient way between
the two arguments, for example where TT1 is the Julian Day Number
and TT2 is the fraction of a day. The returned UT11,UT12 follow
*
*
*
*
       suit.
*
   2) The argument DT is classical Delta T.
*
*
*
  Reference:
*
*
       Explanatory Supplement to the Astronomical Almanac,
*
       P. Kenneth Seidelmann (ed), University Science Books (1992)
*
*_
```

```
SUBROUTINE iau_UT1TAI ( UT11, UT12, DTA, TAI1, TAI2, J )
*+
*
*
   iau_UT1TAI
*
            _ _ _ _ _
*
*
  Time scale transformation: Universal Time, UT1, to International
*
  Atomic Time, TAI.
*
  This routine is part of the International Astronomical Union's
*
*
  SOFA (Standards of Fundamental Astronomy) software collection.
*
*
  Status: canonical.
*
*
   Given:
*
    UT11,UT12 d UT1 as a 2-part Julian Date
*
     DTA
                   d
                          UT1-TAI in seconds
*
*
  Returned:
*
     TAI1,TAI2 d TAI as a 2-part Julian Date
J i status: 0 = OK
*
*
*
  Notes:
*
*
   1) UT11+UT12 is Julian Date, apportioned in any convenient way
      between the two arguments, for example where UT11 is the Julian
Day Number and UT12 is the fraction of a day. The returned
*
*
*
      TAI1, TAI2 follow suit.
   2) The argument DTA, i.e. UT1-TAI, is an observed quantity, and is
*
*
      available from IERS tabulations.
*
*
  Reference:
*
*
      Explanatory Supplement to the Astronomical Almanac,
*
      P. Kenneth Seidelmann (ed), University Science Books (1992)
*
*_
```

SUBROUTINE iau_UT1TT (UT11, UT12, DT, TT1, TT2, J) *+ * * iau_UT1TT * _ _ _ _ _ _ _ _ _ * * Time scale transformation: Universal Time, UT1, to Terrestrial Time, * TT. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: canonical. * * Given: UT1 as a 2-part Julian Date TT-UT1 in seconds * UT11,UT12 d * d DT* * Returned: d TT as a 2-part Julian Date i status: 0 = OK * TT1,TT2 * J * * Notes: * 1) UT11+UT12 is Julian Date, apportioned in any convenient way between the two arguments, for example where UT11 is the Julian Day Number and UT12 is the fraction of a day. The returned * * * * TT1,TT2 follow suit. * * 2) The argument DT is classical Delta T. * * Reference: * * Explanatory Supplement to the Astronomical Almanac, * P. Kenneth Seidelmann (ed), University Science Books (1992) * *_

SUBROUTINE iau_UT1UTC (UT11, UT12, DUT1, UTC1, UTC2, J) *+ * iau_UT1UTC * * Time scale transformation: Universal Time, UT1, to Coordinated * Universal Time, UTC. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical. * * Given: * UT11,UT12 d UT1 as a 2-part Julian Date (Note 1) * DUT1 Delta UT1: UT1-UTC in seconds (Note 2) d * Returned: d UTC1,UTC2 UTC as a 2-part quasi Julian Date (Notes 3,4) status: +1 = dubious year (Note 5) J i * 0 = OK -1 = unacceptable date * Notes: 1) UT11+UT12 is Julian Date, apportioned in any convenient way between the two arguments, for example where UT11 is the Julian Day Number and UT12 is the fraction of a day. The returned UTC1 and UTC2 form an analogous pair, except that a special convention is used, to deal with the problem of leap seconds - see Note 3. 2) Delta UT1 can be obtained from tabulations provided by the International Earth Rotation and Reference Systems Service. The value changes abruptly by 1s at a leap second; however, close to a leap second the algorithm used here is tolerant of the "wrong" choice of value being made. 3) JD cannot unambiguously represent UTC during a leap second unless special measures are taken. The convention in the present routine is that the returned quasi JD day UTC1+UTC2 represents UTC days whether the length is 86399, 86400 or 86401 SI seconds. 4) The routine iau_D2DTF can be used to transform the UTC quasi-JD into calendar date and clock time, including UTC leap second handling. 5) The warning status "dubious year" flags UTCs that predate the introduction of the time scale and that are too far in the future * to be trusted. See iau_DAT for further details. * Called: iau JD2CAL * JD to Gregorian calendar delta(AT) = TAI-UTC iau DAT * iau_CAL2JD Gregorian calendar to JD * References: McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * * Explanatory Supplement to the Astronomical Almanac, * P. Kenneth Seidelmann (ed), University Science Books (1992) *

```
SUBROUTINE iau_UTCTAI ( UTC1, UTC2, TAI1, TAI2, J )
*+
*
   i a u _ U T C T A I
*
*
*
  Time scale transformation: Coordinated Universal Time, UTC, to
*
  International Atomic Time, TAI.
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
  Status: canonical.
*
*
  Given:
*
     UTC1,UTC2
                   d
                          UTC as a 2-part quasi Julian Date (Notes 1-4)
*
  Returned:
*
     TAI1,TAI2
                   d
                           TAI as a 2-part Julian Date (Note 5)
                           status: +1 = dubious year (Note 3)
      ιT
                   i
                                   0 = OK
                                   -1 = unacceptable date
*
  Notes:
  1) UTC1+UTC2 is quasi Julian Date (see Note 2), apportioned in any
*
      convenient way between the two arguments, for example where UTC1
      is the Julian Day Number and UTC2 is the fraction of a day.
*
   2) JD cannot unambiguously represent UTC during a leap second unless
      special measures are taken. The convention in the present routine
      is that the JD day represents UTC days whether the length is
      86399, 86400 or 86401 SI seconds.
   3) The warning status "dubious year" flags UTCs that predate the
      introduction of the time scale and that are too far in the future
      to be trusted. See iau_DAT for further details.
   4) The routine iau_DTF2D converts from calendar date and time of day
      into 2-part Julian Date, and in the case of UTC implements the
*
      leap-second-ambiguity convention described above.
*
  5) The returned TAI1, TAI2 are such that their sum is the TAI Julian
*
     Date.
*
  Called:
*
      iau JD2CAL
                   JD to Gregorian calendar
                   delta(AT) = TAI-UTC
      iau DAT
      iau_CAL2JD Gregorian calendar to JD
*
  References:
     McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
      Explanatory Supplement to the Astronomical Almanac,
*
      P. Kenneth Seidelmann (ed), University Science Books (1992)
*
*
```

```
SUBROUTINE iau_UTCUT1 ( UTC1, UTC2, DUT1, UT11, UT12, J )
*+
*
   iau_UTCUT1
*
*
  Time scale transformation: Coordinated Universal Time, UTC, to
*
  Universal Time, UT1.
*
  This routine is part of the International Astronomical Union's
*
  SOFA (Standards of Fundamental Astronomy) software collection.
  Status: canonical.
*
*
   Given:
*
     UTC1,UTC2
                   d
                          UTC as a 2-part quasi Julian Date (Notes 1-4)
*
     DUT1
                           Delta UT1 = UT1-UTC in seconds (Note 5)
                   d
*
  Returned:
                   d
                           UT1 as a 2-part Julian Date (Note 6)
     UT11,UT12
      J
                   i
                           status: +1 = dubious year (Note 7)
*
                                    0 = OK
                                    -1 = unacceptable date
*
  Notes:
   1) UTC1+UTC2 is quasi Julian Date (see Note 2), apportioned in any
      convenient way between the two arguments, for example where UTC1
      is the Julian Day Number and UTC2 is the fraction of a day.
   2) JD cannot unambiguously represent UTC during a leap second unless
      special measures are taken. The convention in the present routine
is that the JD day represents UTC days whether the length is
      86399, 86400 or 86401 SI seconds.
   3) The warning status "dubious year" flags UTCs that predate the
      introduction of the time scale and that are too far in the future
      to be trusted. See iau_DAT for further details.
   4) The routine iau_DTF2D converts from calendar date and time of day
      into 2-part Julian Date, and in the case of UTC implements the
      leap-second-ambiguity convention described above.
   5) Delta UT1 can be obtained from tabulations provided by the
      International Earth Rotation and Reference Systems Service.
                                                                       It
      It is the caller's responsibility to supply a DUT argument
*
      containing the UT1-UTC value that matches the given UTC.
   6) The returned UT11,UT12 are such that their sum is the UT1 Julian
*
      Date.
   7) The warning status "dubious year" flags UTCs that predate the
      introduction of the time scale and that are too far in the future
      to be trusted. See iau_DAT for further details.
*
  References:
      McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004)
*
*
      Explanatory Supplement to the Astronomical Almanac,
*
      P. Kenneth Seidelmann (ed), University Science Books (1992)
  Called:
*
      iau_JD2CAL
                   JD to Gregorian calendar
*
      iau DAT
                   delta(AT) = TAI-UTC
*
      iau_UTCTAI
                   UTC to TAI
*
                   TAI to UT1
      iau_TAIUT1
*
*_
```

SUBROUTINE iau_XY06 (DATE1, DATE2, X, Y) *+ * i a u _ X Y O 6 * * X,Y coordinates of celestial intermediate pole from series based * on IAU 2006 precession and IAU 2000A nutation. This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. Status: canonical model. * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) Returned: * Х.Ү Р CIP X, Y coordinates (Note 2) Notes: * 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, among others: DATE1 DATE2 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods are both good compromises between resolution and convenience. 2) The $\ensuremath{\mathtt{X}},\ensuremath{\mathtt{Y}}$ coordinates are those of the unit vector towards the celestial intermediate pole. They represent the combined effects of frame bias, precession and nutation. 3) The fundamental arguments used are as adopted in IERS Conventions (2003) and are from Simon et al. (1994) and Souchay et al. (1999). 4) This is an alternative to the angles-based method, via the SOFA * routine iau_FW2XY and as used in iau_XYS06A for example. The two methods agree at the 1 microarcsecond level (at present), a negligible amount compared with the intrinsic accuracy of the models. However, it would be unwise to mix the two methods (angles-based and series-based) in a single application. Called: * iau_FAL03 mean anomaly of the Moon iau_FALP03 mean anomaly of the Sun mean argument of the latitude of the Moon iau_FAF03 mean elongation of the Moon from the Sun iau_FAD03 iau_FAOM03 mean longitude of the Moon's ascending node iau_FAME03 mean longitude of Mercury * mean longitude of Venus iau FAVE03 mean longitude of Earth iau_FAE03 iau_FAMA03 * mean longitude of Mars iau FAJU03 mean longitude of Jupiter iau_FASA03 mean longitude of Saturn iau_FAUR03 mean longitude of Uranus * iau_FANE03 mean longitude of Neptune iau_FAPA03 general accumulated precession in longitude

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        Wallace, P.T. & Capitaine, N., 2006, Astron.Astrophys. 459, 981
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SUBROUTINE iau_XYS00A (DATE1, DATE2, X, Y, S) *+ * i a u _ X Y S O O A * * * For a given TT date, compute the X,Y coordinates of the Celestial * Intermediate Pole and the CIO locator s, using the IAU 2000A * precession-nutation model. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * d Celestial Intermediate Pole (Note 2) Х,Ү * S d the CIO locator s (Note 2) * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 * 2450123.7D0 0D0 (JD method) * 2451545D0 -1421.3D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in * * cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * * 2) The Celestial Intermediate Pole coordinates are the x,y components of the unit vector in the Geocentric Celestial Reference System. * 3) The CIO locator s (in radians) positions the Celestial Intermediate Origin on the equator of the CIP. * 4) A faster, but slightly less accurate result (about 1 mas for X,Y), can be obtained by using instead the iau_XYS00B routine. * Called: iau_PNM00A classical NPB matrix, IAU 2000A iau_BPN2XY extract CIP X,Y coordinates from NPB matrix iau_S00 the CIO locator s, given X,Y, IAU 2000A * * Reference: McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * *_

SUBROUTINE iau_XYS00B (DATE1, DATE2, X, Y, S) *+ * i a u _ X Y S O O B * * * For a given TT date, compute the X,Y coordinates of the Celestial * Intermediate Pole and the CIO locator s, using the IAU 2000B * precession-nutation model. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * d Celestial Intermediate Pole (Note 2) Х,Ү * S d the CIO locator s (Note 2) * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 * 2450123.7D0 0D0 (JD method) 2451545D0 -1421.3D0 (J2000 method) * 2400000.5D0 50123.2D0 (MJD method) 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in * cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * * 2) The Celestial Intermediate Pole coordinates are the x,y components of the unit vector in the Geocentric Celestial Reference System. * 3) The CIO locator s (in radians) positions the Celestial Intermediate Origin on the equator of the CIP. * 4) The present routine is faster, but slightly less accurate (about 1 mas in X,Y), than the iau_XYSOOA routine. * Called: iau_PNM00B classical NPB matrix, IAU 2000B iau_BPN2XY extract CIP X,Y coordinates from NPB matrix iau_S00 the CIO locator s, given X,Y, IAU 2000A * * Reference: McCarthy, D. D., Petit, G. (eds.), IERS Conventions (2003), IERS Technical Note No. 32, BKG (2004) * * *_

SUBROUTINE iau_XYS06A (DATE1, DATE2, X, Y, S) *+ * i a u _ X Y S O 6 A * * * For a given TT date, compute the X,Y coordinates of the Celestial * Intermediate Pole and the CIO locator s, using the IAU 2006 * precession and IAU 2000A nutation models. * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * Status: support routine. * * Given: * DATE1,DATE2 d TT as a 2-part Julian Date (Note 1) * Returned: * d Celestial Intermediate Pole (Note 2) Х,Ү * S d the CIO locator s (Note 2) * Notes: 1) The TT date DATE1+DATE2 is a Julian Date, apportioned in any convenient way between the two arguments. For example, JD(TT)=2450123.7 could be expressed in any of these ways, * among others: DATE1 DATE2 2450123.7D0 000 (JD method) 2451545D0 -1421.3D0 (J2000 method) 2400000.5D0 50123.2D0 (MJD method) * 2450123.5D0 0.2D0 (date & time method) The JD method is the most natural and convenient to use in * cases where the loss of several decimal digits of resolution is acceptable. The J2000 method is best matched to the way * the argument is handled internally and will deliver the optimum resolution. The MJD method and the date & time methods * are both good compromises between resolution and convenience. * * 2) The Celestial Intermediate Pole coordinates are the x,y components of the unit vector in the Geocentric Celestial Reference System. * 3) The CIO locator s (in radians) positions the Celestial Intermediate Origin on the equator of the CIP. * 4) Series-based solutions for generating X and Y are also available: see Capitaine & Wallace (2006) and iau_XY06. * Called: classical NPB matrix, IAU 2006/2000A iau PNM06A iau_BPN2XY extract CIP X,Y coordinates from NPB matrix iau_S06 the CIO locator s, given X,Y, IAU 2006 * * References: Capitaine, N. & Wallace, P.T., 2006, Astron.Astrophys. 450, 855 * * Wallace, P.T. & Capitaine, N., 2006, Astron.Astrophys. 459, 981 * *_

SUBROUTINE iau_ZP (P) *+ * - - - - - iau_ZP _____ * * * * Zero a p-vector. * * This routine is part of the International Astronomical Union's
 * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Returned: * d(3) p-vector Ρ * *_

SUBROUTINE iau_ZPV (PV) *+ * _ _ _ _ _ _ _ i a u _ Z P V * * * * Zero a pv-vector. * * This routine is part of the International Astronomical Union's * SOFA (Standards of Fundamental Astronomy) software collection. * * Status: vector/matrix support routine. * * Returned: * d(3,2) pv-vector PV * * Called: * iau_ZP zero p-vector * *_

```
SUBROUTINE iau_ZR ( R )
*+
*
          - - -
*
   iau_ZR
*
   _ _ _ _ _ _ _ _
*
*
  Initialize an r-matrix to the null matrix.
*

    * This routine is part of the International Astronomical Union's
    * SOFA (Standards of Fundamental Astronomy) software collection.

*
*
   Status: vector/matrix support routine.
*
* Returned:
*
                d(3,3) r-matrix
      R
*
*_
```

```
copyr.lis
```

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SOFA Fortran constants _____ These must be used exactly as presented below. * Pi DOUBLE PRECISION DPI PARAMETER (DPI = 3.141592653589793238462643D0) * 2Pi DOUBLE PRECISION D2PI PARAMETER (D2PI = 6.283185307179586476925287D0) * Radians to hours DOUBLE PRECISION DR2H PARAMETER (DR2H = 3.819718634205488058453210D0) * Radians to seconds DOUBLE PRECISION DR2S PARAMETER (DR2S = 13750.98708313975701043156D0) * Radians to degrees DOUBLE PRECISION DR2D PARAMETER (DR2D = 57.29577951308232087679815D0) * Radians to arc seconds DOUBLE PRECISION DR2AS PARAMETER (DR2AS = 206264.8062470963551564734D0) Hours to radians DOUBLE PRECISION DH2R PARAMETER (DH2R = 0.2617993877991494365385536D0) * Seconds to radians DOUBLE PRECISION DS2R PARAMETER (DS2R = 7.272205216643039903848712D-5) * Degrees to radians DOUBLE PRECISION DD2R PARAMETER (DD2R = 1.745329251994329576923691D-2) Arc seconds to radians DOUBLE PRECISION DAS2R PARAMETER (DAS2R = 4.848136811095359935899141D-6)

SOFA C constants

The constants used by the C version of SOFA are defined in the header file sofam.h.

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